





# Outline

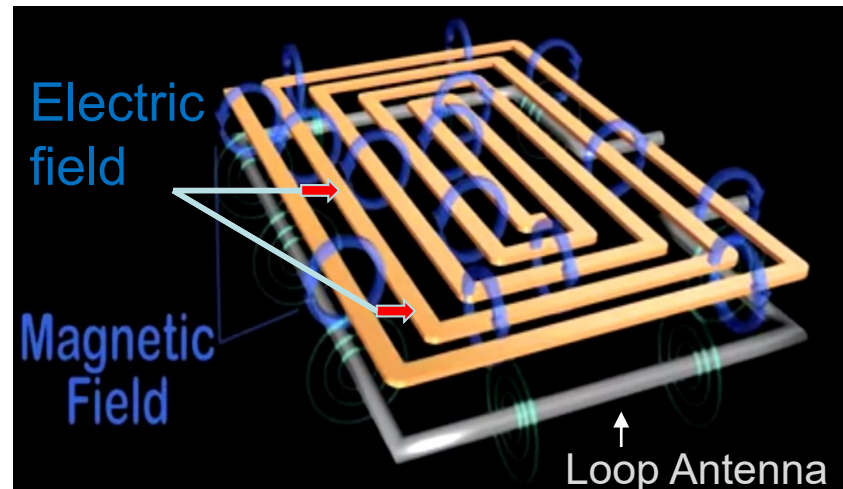
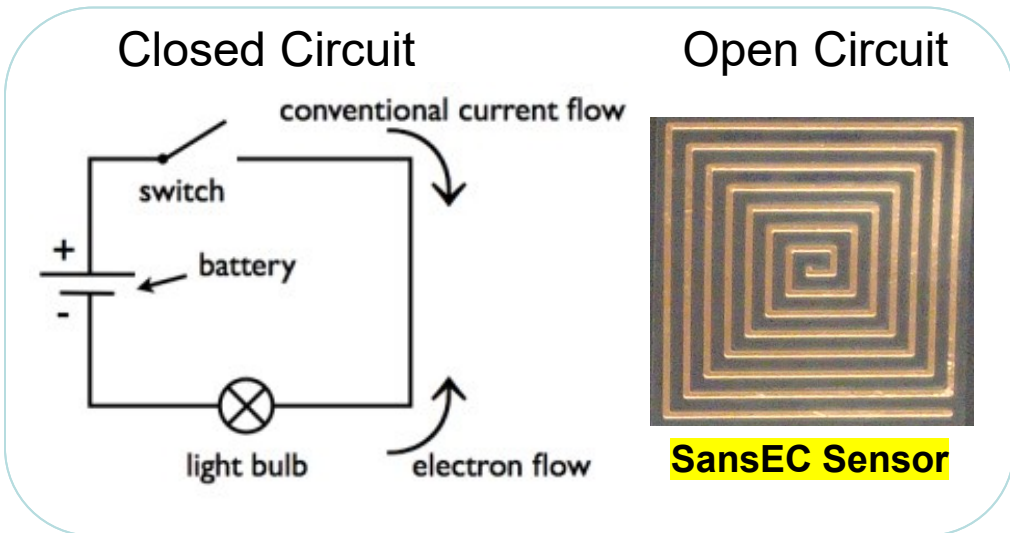
- **Introduction to SansEC Sensors**
  - Description & Theory
  - Computational Electromagnetic Modeling
  - Sensor Fabrication
  - Detectable Phenomena
- **Applications**
  - Composite Aircraft Smart Skin for Lightning Environments
  - Wearables
- **Detectable Phenomena**
- **Conclusions**
- **Reference Publications**



# Introduction to SansEC Sensors

SansEC ↔ Sans Electric Connection

- **Wireless** – No physical electrical connections
- **Passive** – No battery required;
- **Non-Contact** – EM Fields permeate out from sensor surface
- **Electromagnetic Resonator** – naturally oscillates with greater amplitude at some frequencies based on Inductance (L), Capacitance (C) & Resistance (R) of the total circuit

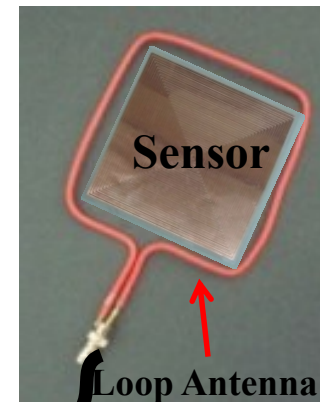
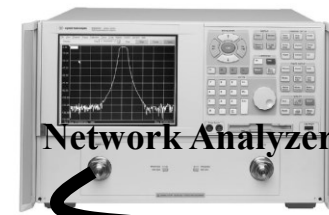




# Introduction to SansEC Sensors

## Components of the Measurement System

1. SansEC Sensor - Conductive coil designed to resonate at desired frequencies  
The trace length, trace width and gap width influence the resonate frequencies.
2. Interrogator RF Source -Typically a Network Analyzer is used, but excitation could come from any stable ambient RF source.
3. Interrogator Receiver - Typically a Network Analyzer is used to measure the reflection coefficient. This is the near field configuration. Data can be collected in the far field by using an electric field antenna connected to a loop antenna.
4. Loop Antenna – Induces magnetic excitation on sensor to generate a signal response  
Loop antenna is aligned parallel to sensor trace.
5. Signal Analysis – Deterministic Quantification using Machine Learning  
Looks for relative change in resonant response to indicate change in device under test.





# Introduction to SansEC Sensors

## Signal Processing Data Features

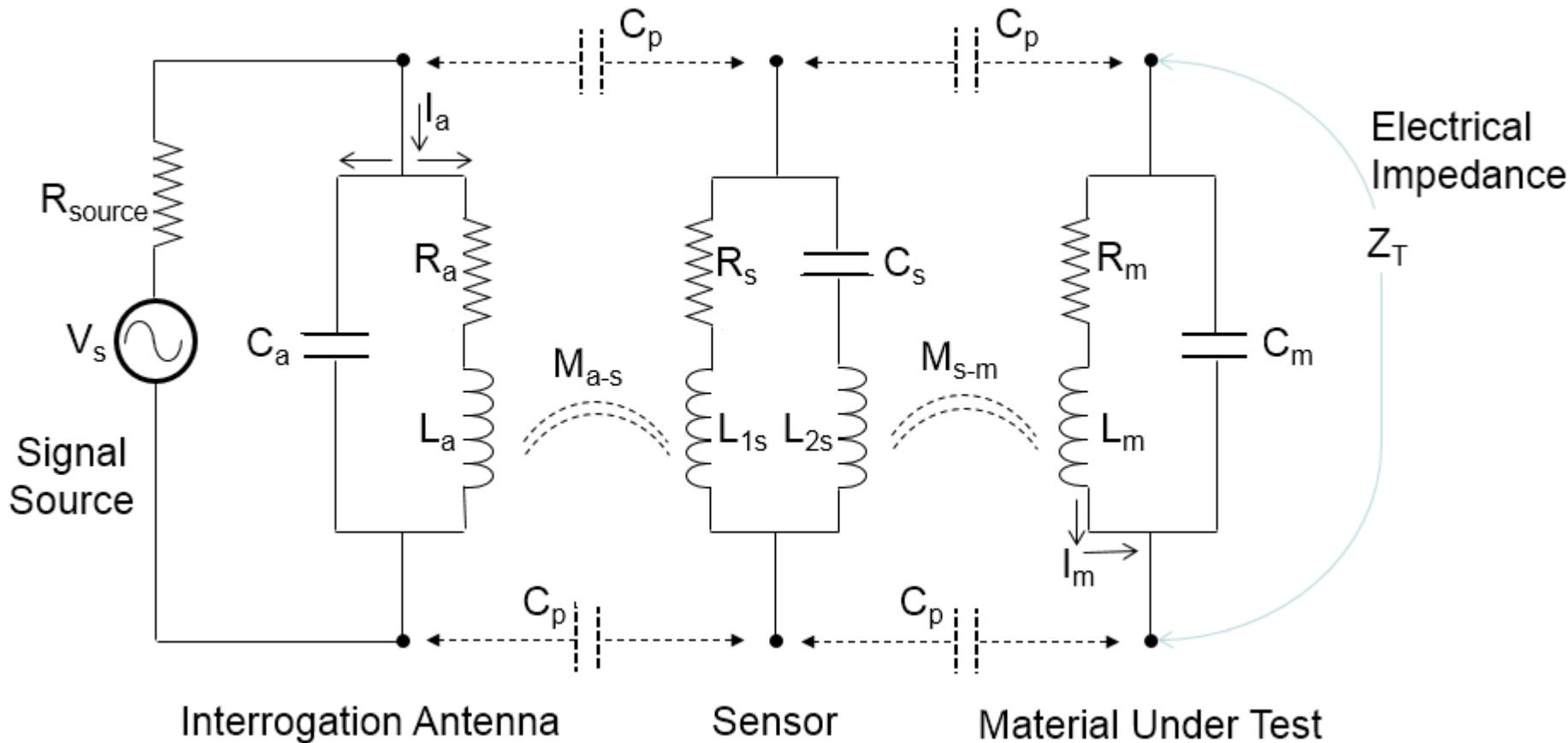
For each Resonant Frequency, the following data features can be used in machine learning quantification from the reflection coefficient measurement.

- Frequency
- Amplitude
- Phase
- Q factor (3dB bandwidth)
- Electrical Impedance
- Derived Parameters (Inductance, Capacitance and Resistance)



# SansEC Sensor Theory

## Equivalent Circuit Model



Lumped components in Antenna circuit will change resonance



# SansEC Sensor Theory

The SansEC sensor will have zero current at both ends of the resonant spiral. When excited by an oscillating electromagnetic field, the induced electromotive force (EMF) pushes the electrons carried by the conductor into the resonant state where the electrons move back and forth along the conductive trace.

## Current Distribution

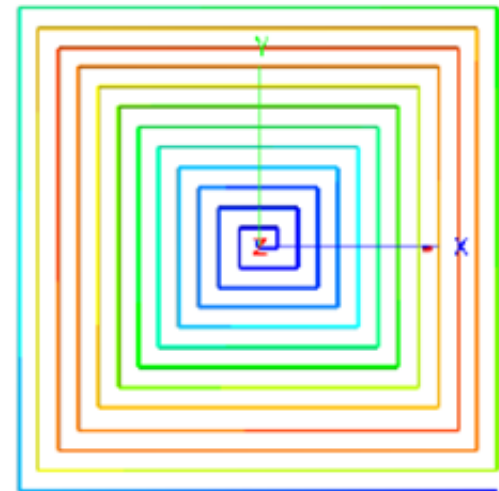
$$I = I_0 \cos\left(\frac{\pi x}{l}\right) e^{-i\omega t}$$

## Charge Distribution

$$q = q_0 \sin\left(\frac{\pi x}{l}\right) e^{-i(\omega t + \frac{\pi}{2})}$$

$$\omega = \frac{1}{\sqrt{LC}}$$

## Current Distribution (zero current at ends)

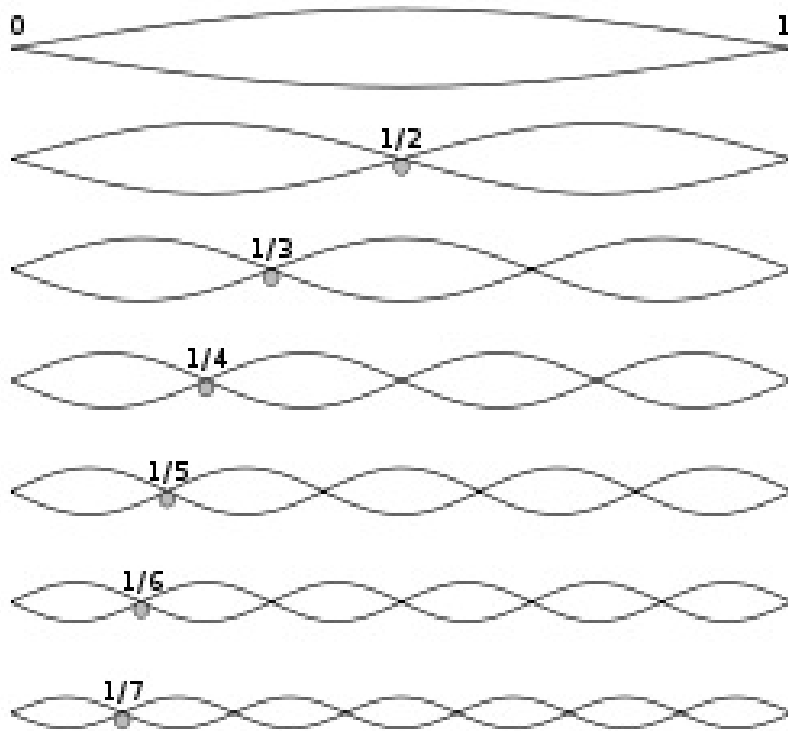


(blue: lowest currents to red: highest currents)

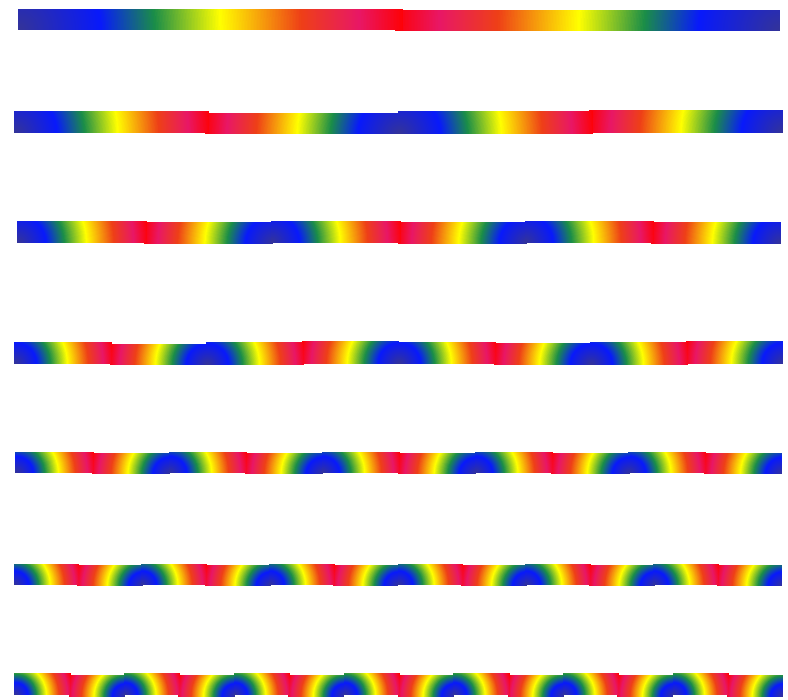


# SansEC Sensor Theory

## Resonant Modes



Current distribution along trace length

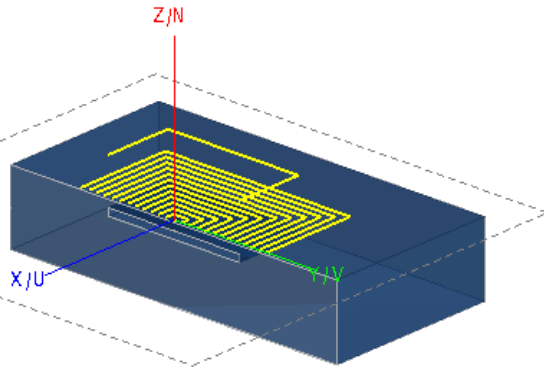


(blue: lowest currents to red: highest currents)

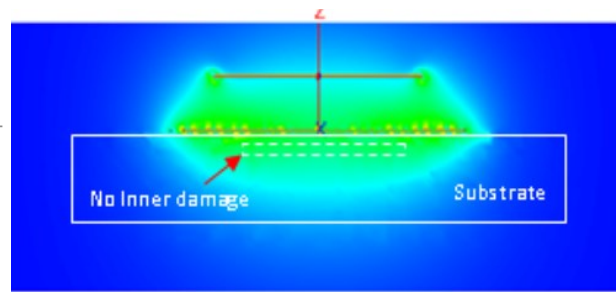


# Computational Electromagnetic Modeling

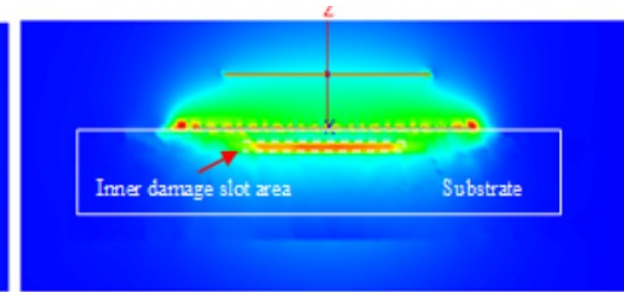
## Response Characteristics from Delamination in Composite



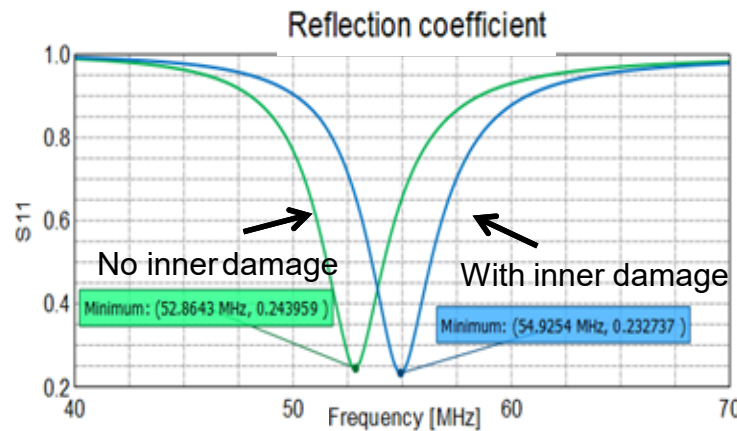
Fiberglass substrate model with inner damage



Electric field distribution No Damage ( $\epsilon_r=3$ )



Electric field distribution With Damage ( $\epsilon_r=1$ )



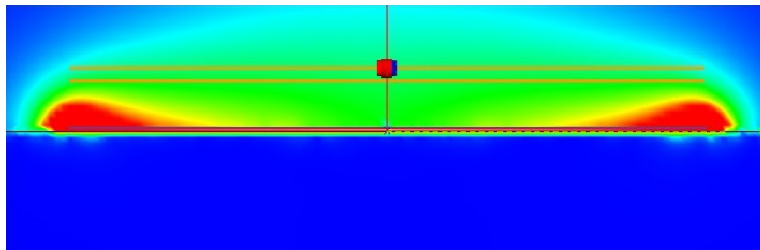
Computed SansEC resonant response



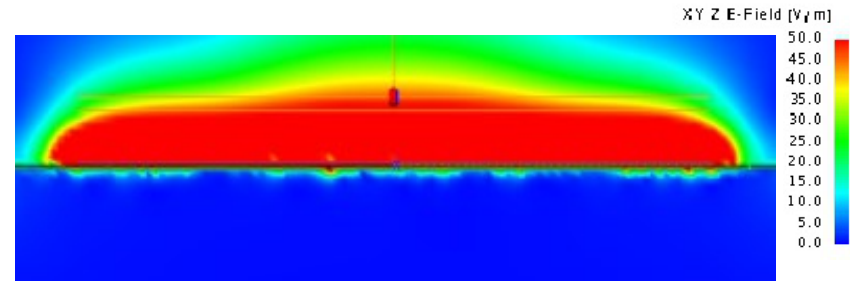
# Computational Electromagnetic Modeling

## Response Characteristics of CFRP Damage

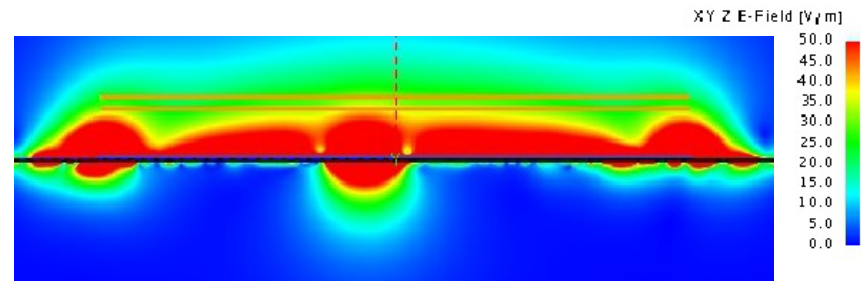
### Electric Field Visualizations



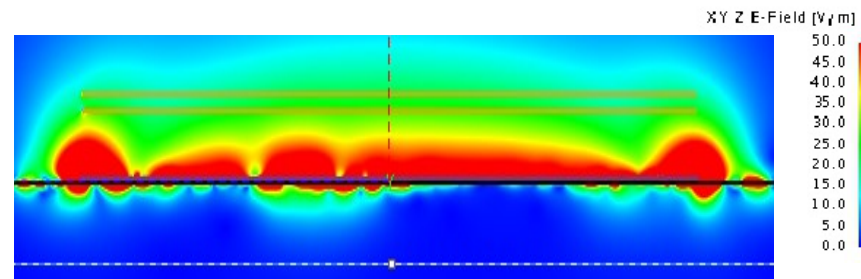
No Damage



Delamination



Puncture

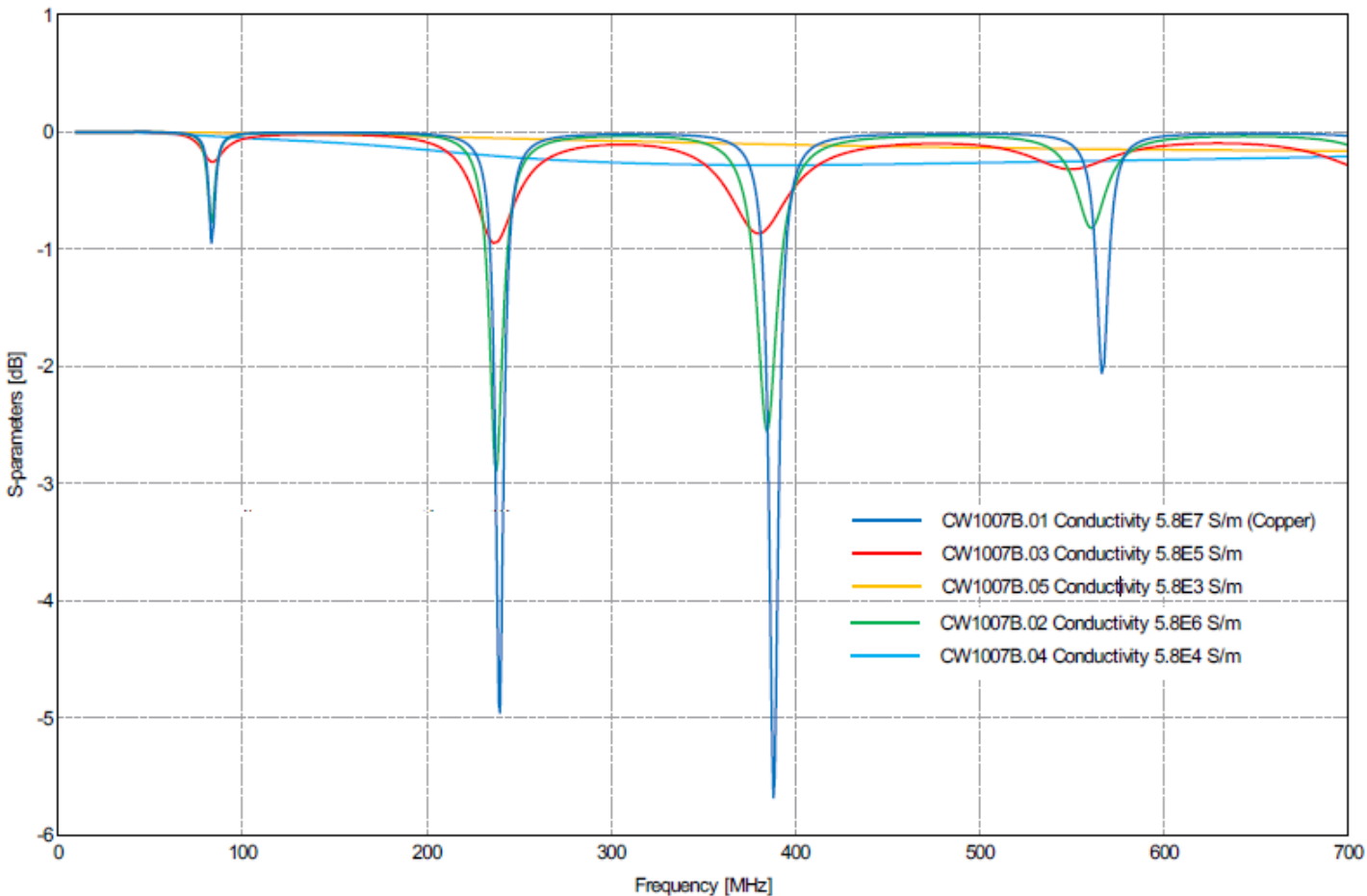


Void



# Computational Electromagnetic Modeling

## Response Characteristics with changing Trace conductivities

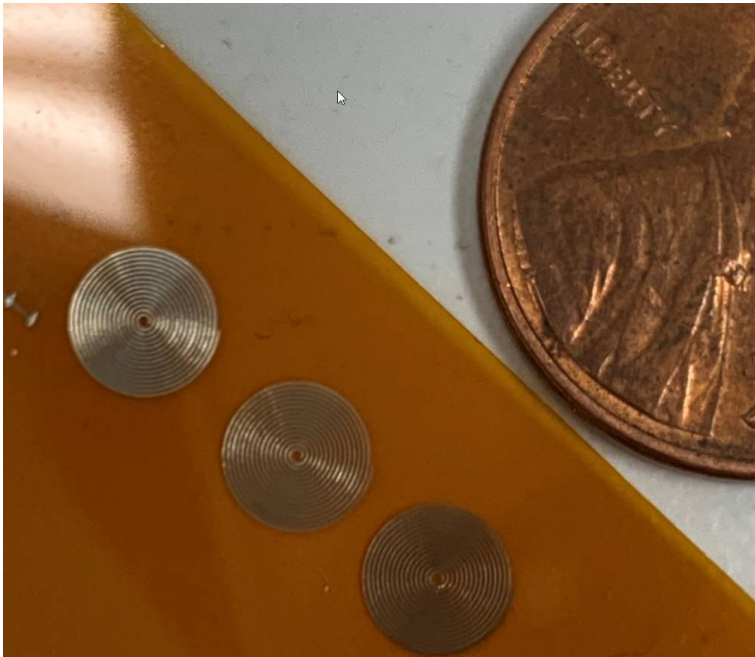




# SansEC Sensor Fabrication

## Sensors can be various shapes or sizes

The sensors free space resonate frequencies are determined by the overall length of the conductive trace along with the trace width and gap width.



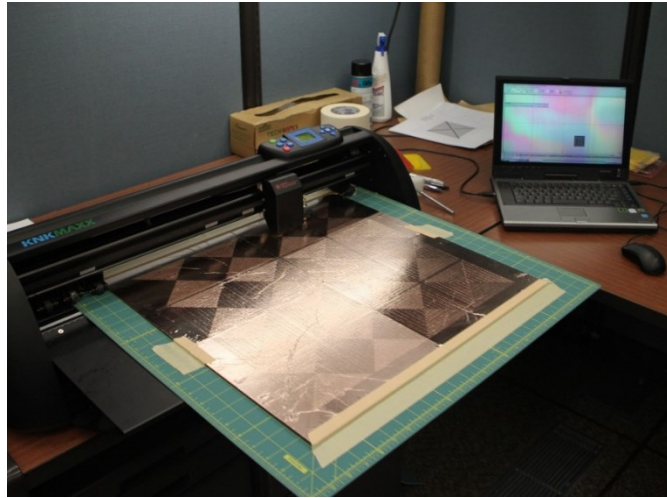
.04 mm trace width from NanoJet printer  
(second generation Aerosol 3D printing technology)



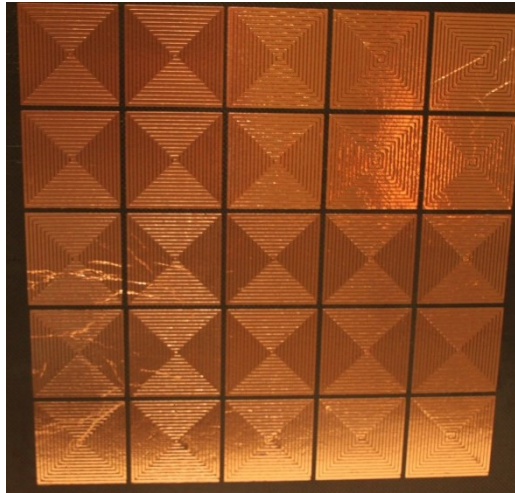
.15 mm trace width using conventional etching  
printed circuit board manufacturing techniques



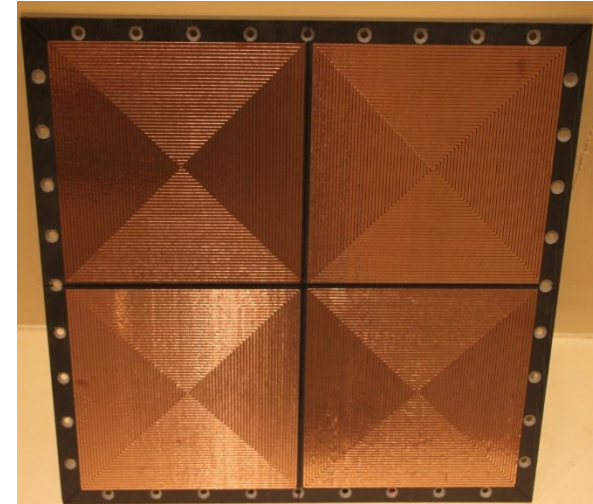
# SansEC Sensor Fabrication



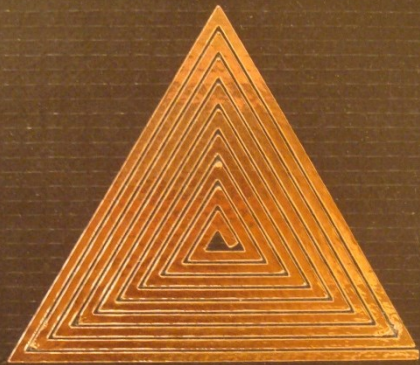
**Klick-n-Kut Cutter**



**5x5 Square Array**



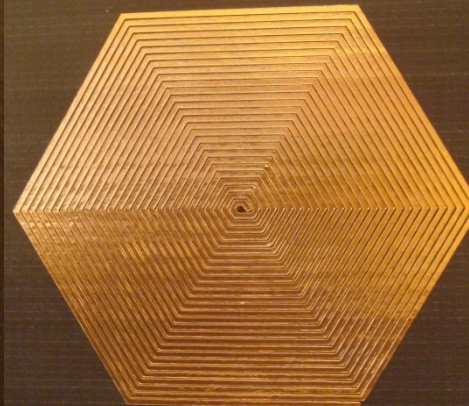
**2x2 Square Array**



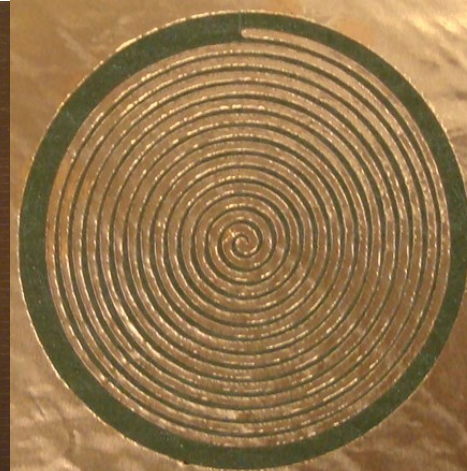
**Triangular**



**Square**



**Hexagonal**

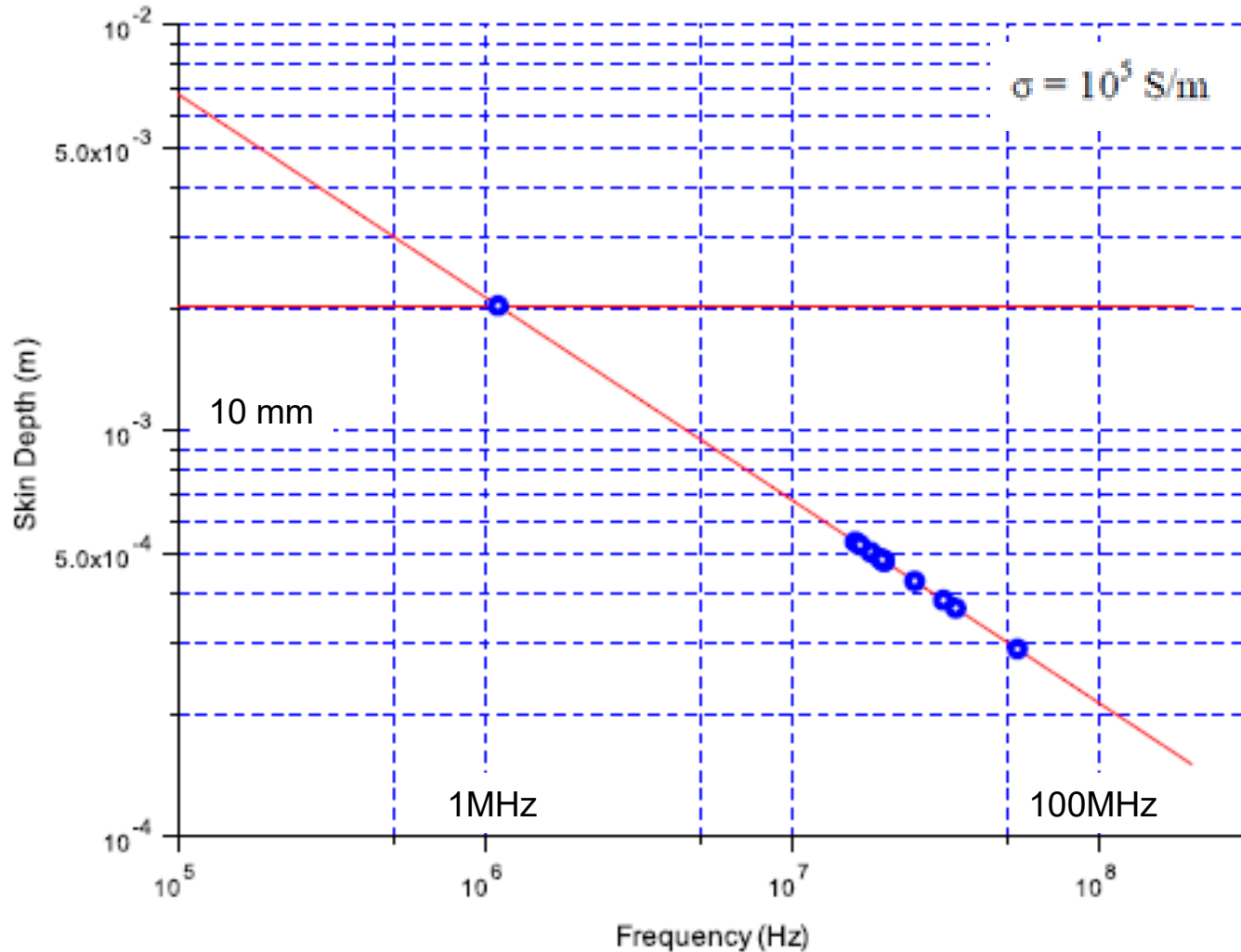


**Circular**

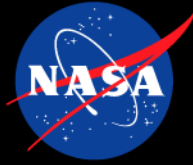


# SansEC Sensor Fabrication

## Skin Depth relative to Frequency



The sensor trace thickness must be greater than the skin depth



# Applications – Composite Smart Skin

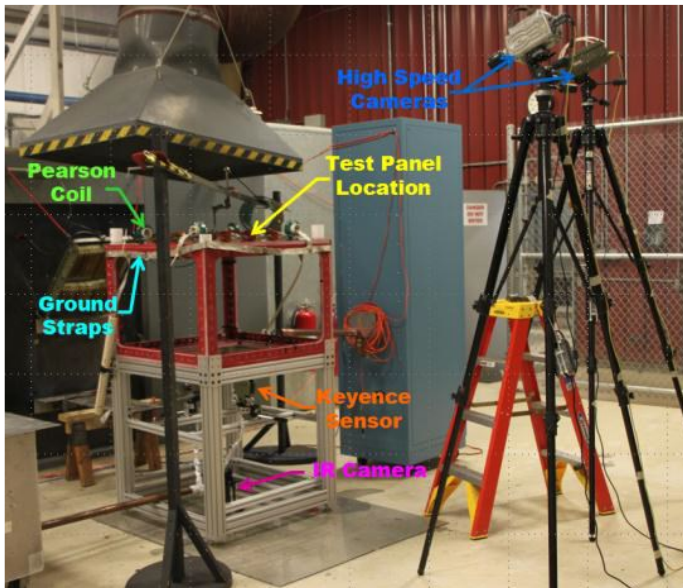
Program research focused on using SansEC Sensors on Carbon Composites

- Composite aircraft require a thin layer of metal in the outer skin to protect avionics and airframe from lightning damage.
- Post-flight ground inspection is conducted after a suspected lightning strike to identify continued air worthiness. (Visual inspections/Tap Test)

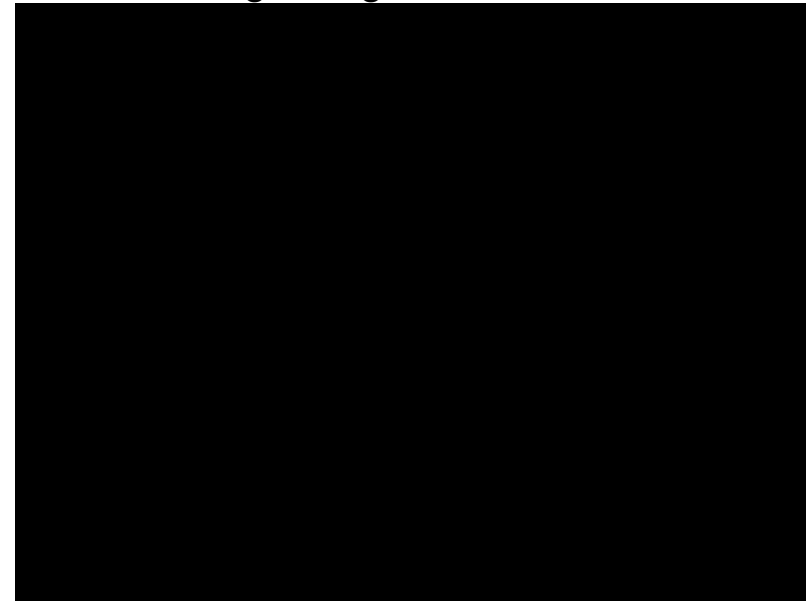


- Demonstrated Lightning Strike Protection conductive layer can be replaced with resonant sensors to provide comparable lightning mitigation characteristics and damage detection.

Lightning Testbed



Lightning Strike Video



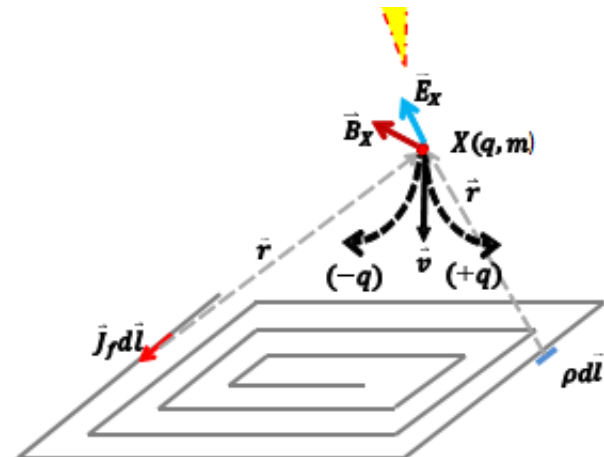
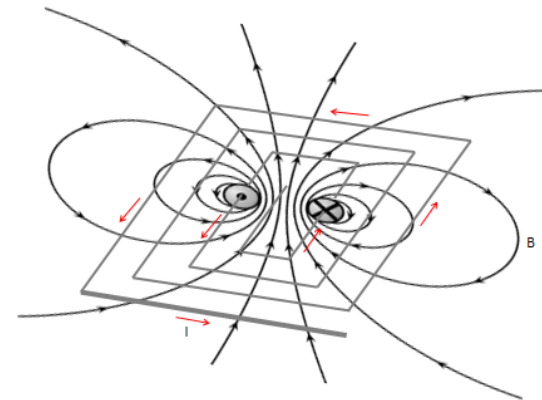
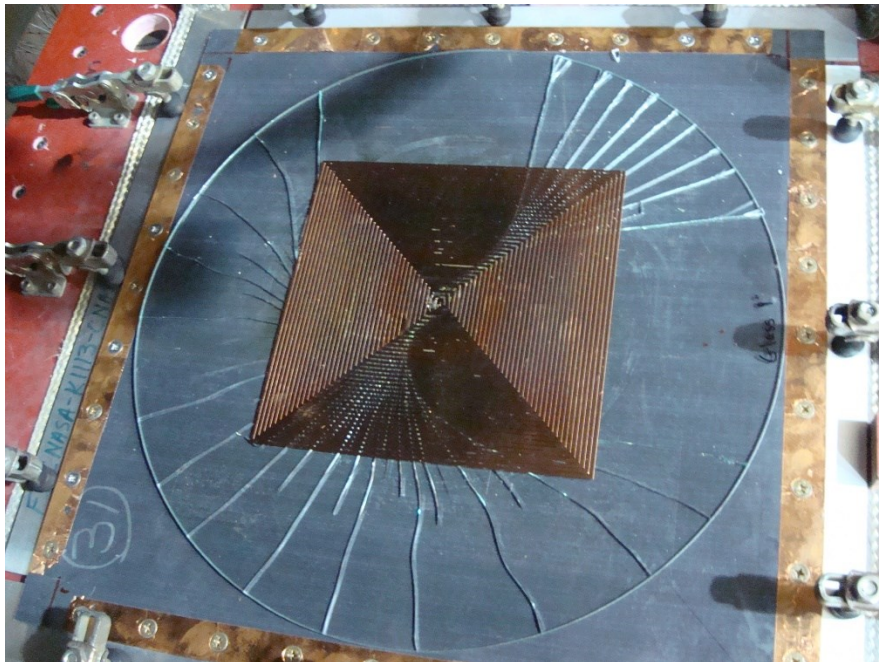


# Applications – Composite Smart Skin

## Lorentz Force

Sensor will generate a Lorentz Force on incoming charged particle

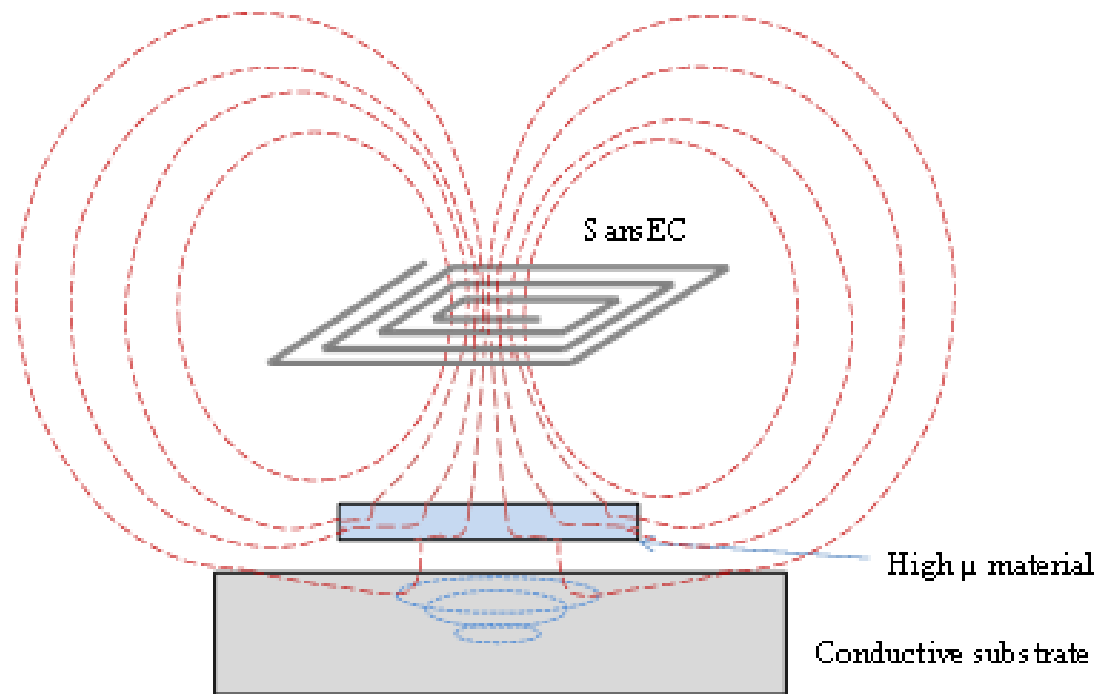
$$\vec{F} = q(\vec{v} \times \vec{B}_X + \vec{E}_X)$$





# Applications – Composite Smart Skin

Operation on Carbon Fiber Composites requires a thin sheet of high permeability material to sustain the resonance.





# Applications – Composite Smart Skin

Previous Program Research

## Universal Common Practice Guide for Lightning Damage Testing of Composites Panel Design

**Layup:** 16 ply tape with 2 ply fabric

**[(SURF)(ECF)(0/90F)/45/90/-45/0/45/90/-45/0/0/-45/90/45/0/-45/90/45/(0/90F)]**

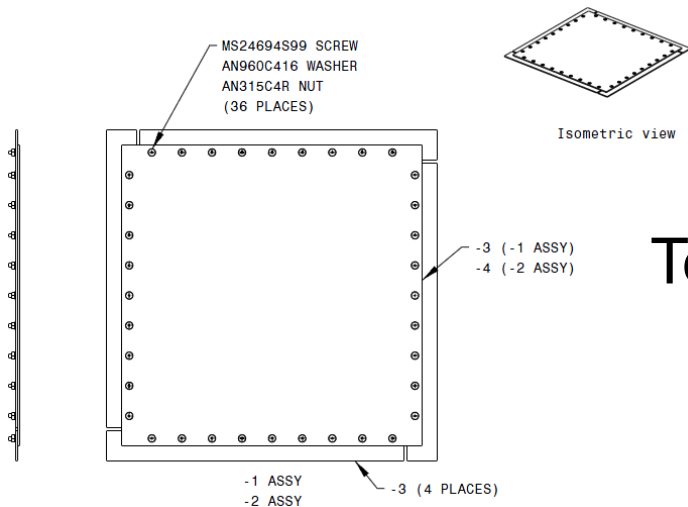
**Tap:** Hexcel 8552/AS4 (HexPly 8552/AS4),

**Fabric:** Hexcel 8552/AS4 (HexPly 8552 / A193-PW, 3K-70-PW)

**Lightning Protection:** Dexmet 3CU7-100FA (3mil copper mesh)/ SansEC Smart Skin

**Primer:** 0.5-1.5 mils thickness

**Top coat:** 4 to 6 mils per Mil-F-18264 or equivalent.



### Test Panel





# Applications – Composite Smart Skin

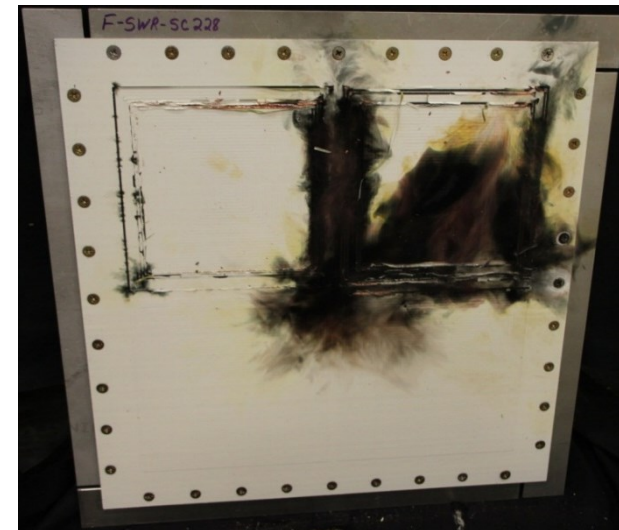
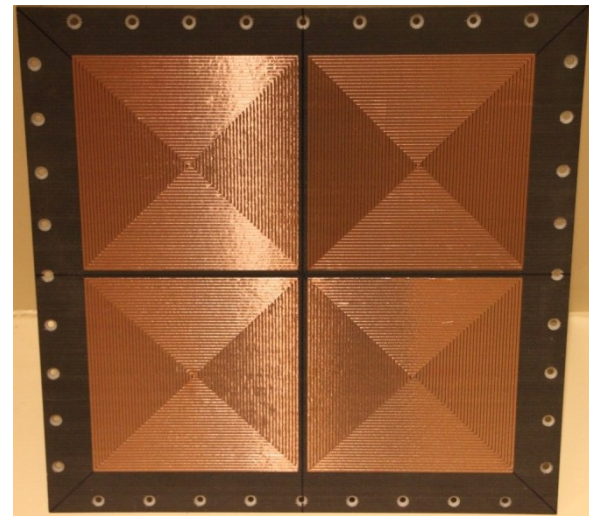
Previous Program Research

## SansEC Fiberglass Reinforced Plastic LSP Test Panels

**Pre-Paint Test Panel**

**Pre-Strike Test Panel**

**Post-Strike Test Panel**



**8 inch SansEC Array**



# Applications – Composite Smart Skin

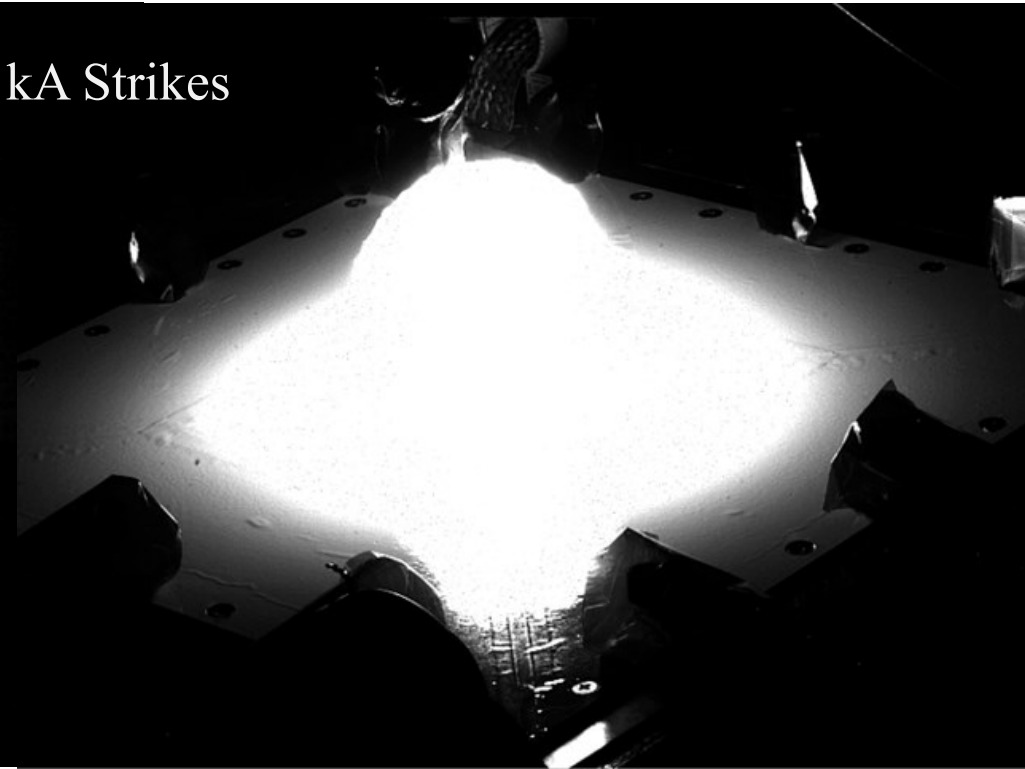
Previous Program Research

## Carbon Fiber Panel Lightning Direct Effects Testing

Two 100 kA Strikes



High Speed Video First Frame showing Lightning Strike Initial Attachment on Mesh Protected Carbon Panel



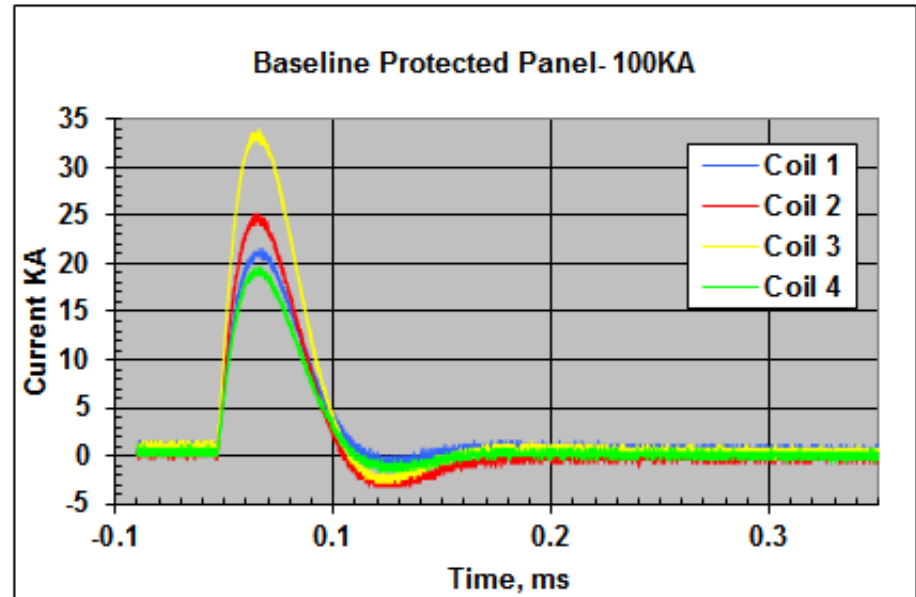
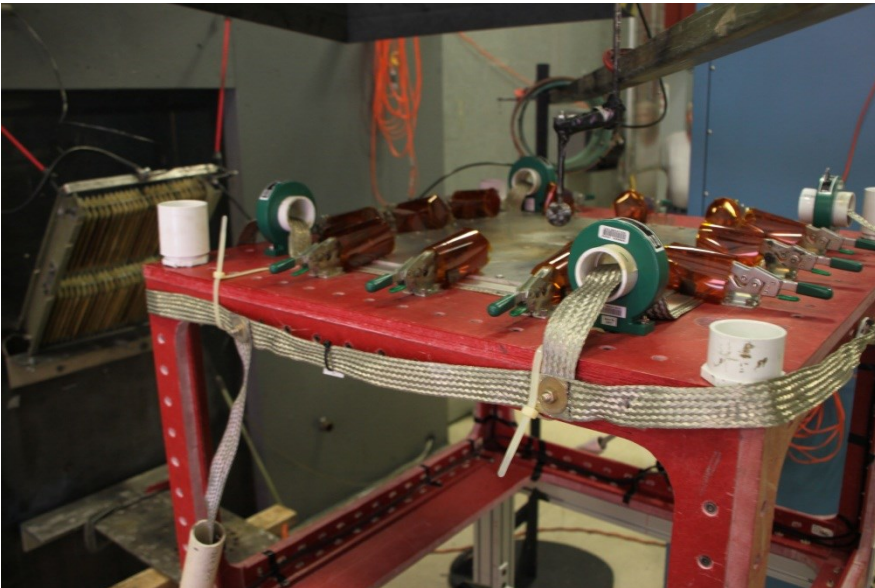
High Speed Video First Frame showing Lightning Strike Initial Attachment on SansEC Protected Carbon Panel



# Applications – Composite Smart Skin

Previous Program Research

Electrical Current on the 4 Edges of the Test Panel were Independently Monitored during Lightning Test



**Pearson Probes mounted on Test Bed**    **Current Data collected during 100KA Strike**

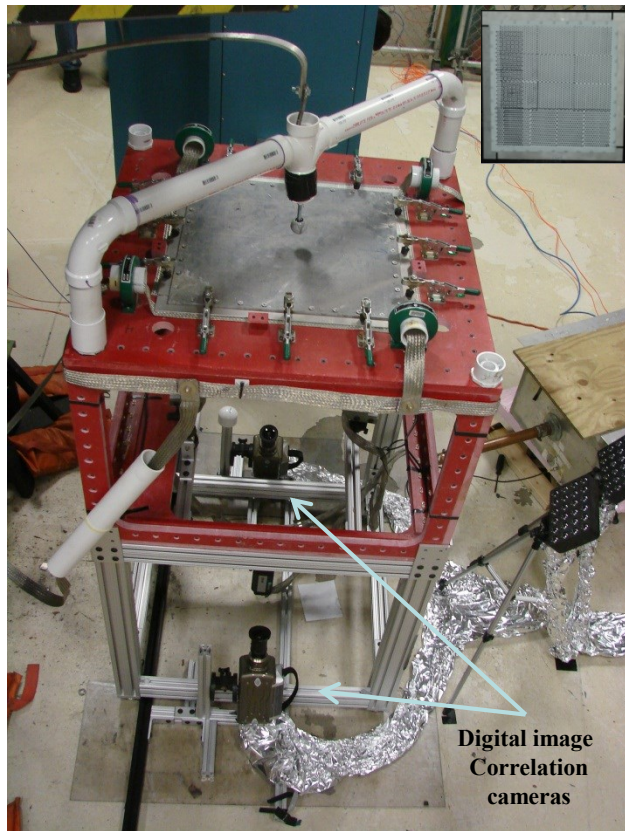
- 4 independent current monitors enable measurements on the 4 edges
- Inductive effects from ground strap installation should be minimized



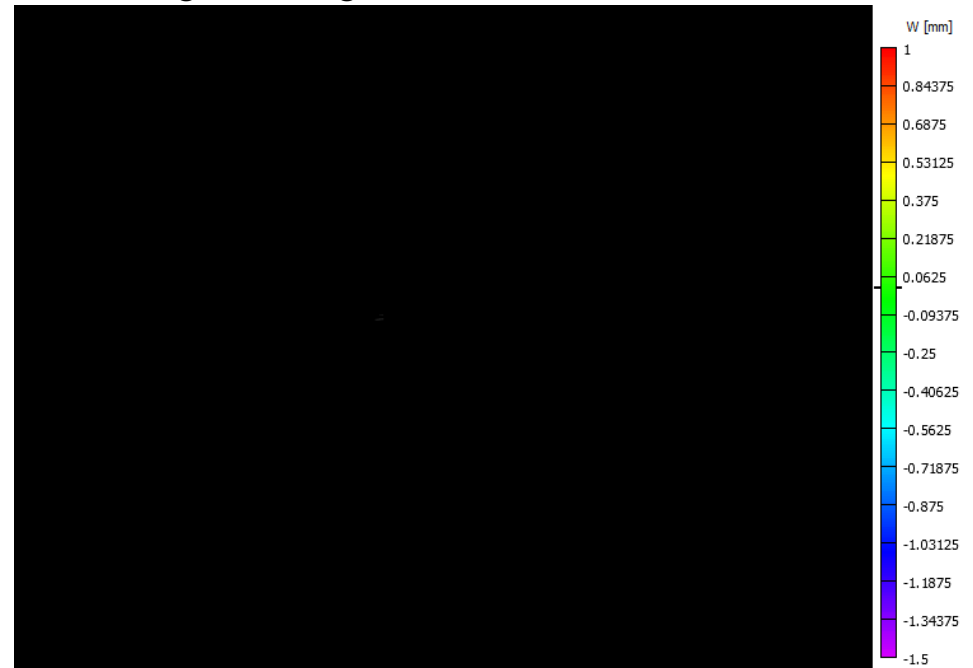
# Applications – Composite Smart Skin

## Previous Program Research

Mechanical deflection is measured using optical sensors or digital image correlation cameras



Digital Image Correlation Video

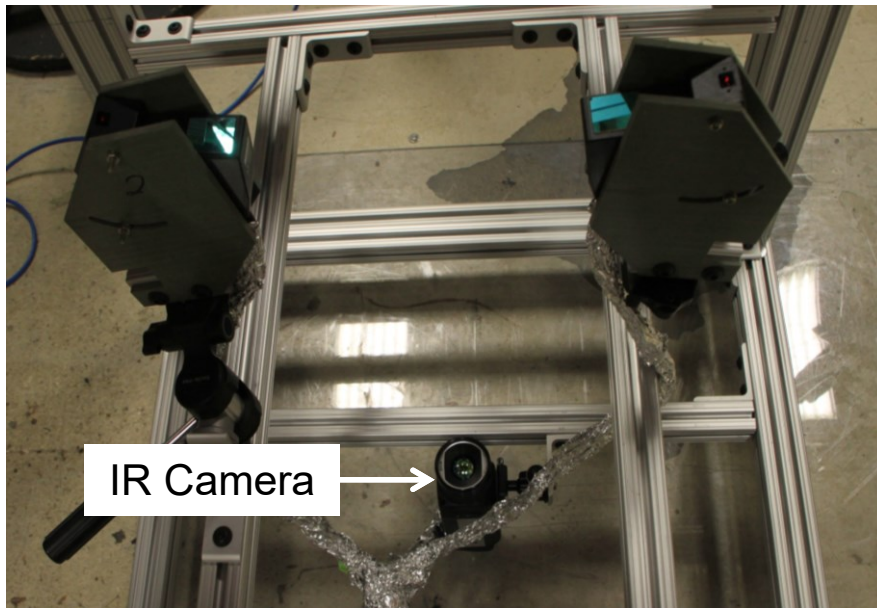




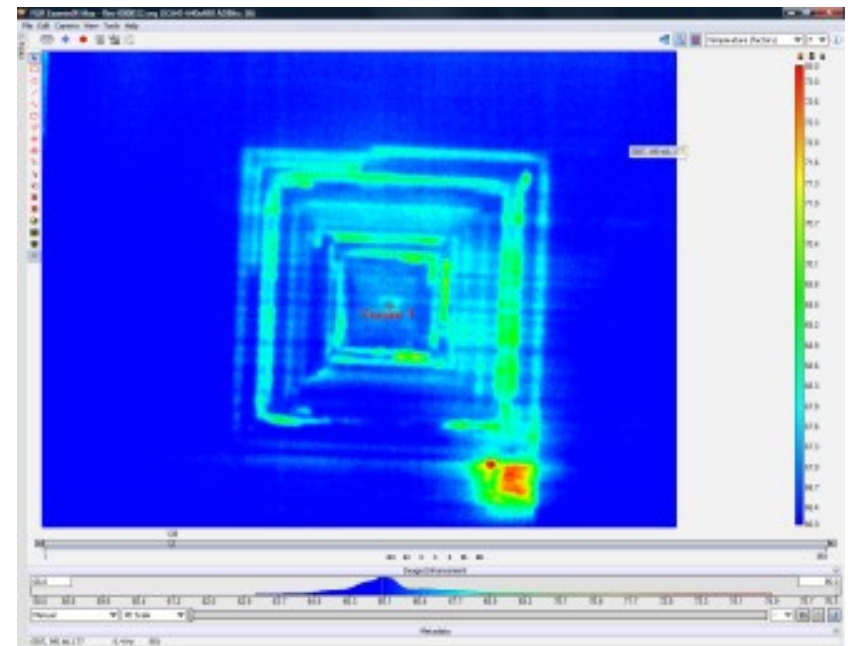
# Applications – Composite Smart Skin

Previous Program Research

Backside Thermal Data was Acquired using FLIR Systems SC645 IR Camera



**Infrared (IR) Camera Setup**



**Backside IR image of 8" SansEC LSP  
on Fiberglass Substrate**

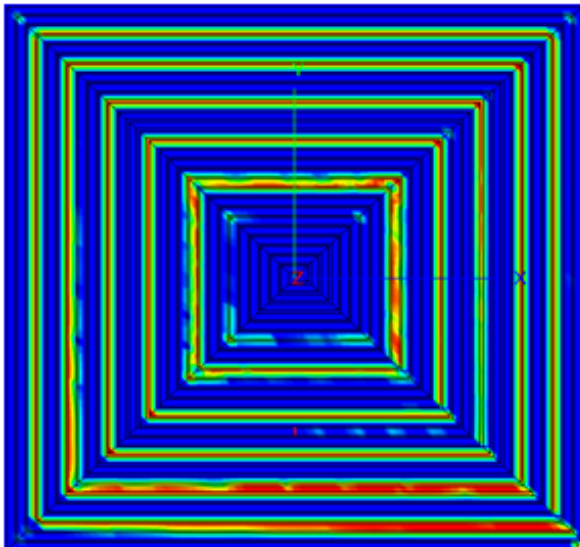
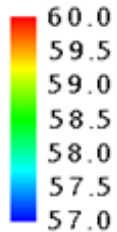


# Applications – Composite Smart Skin

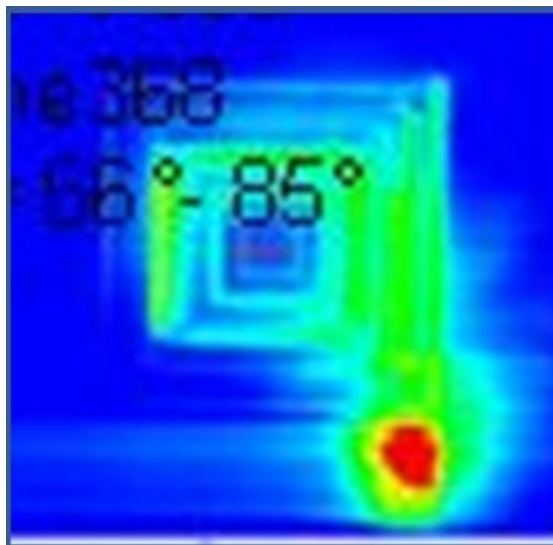
## Previous Program Research

- Hotter temperatures in the IR image represent the location of lightning attachment or current propagation.
- Lightning Attachment was observed to occur at high electric field locations.

XYZ E-Field [dBV/m]



Computed E Field



Backside IR Temperature



Post Strike Damage

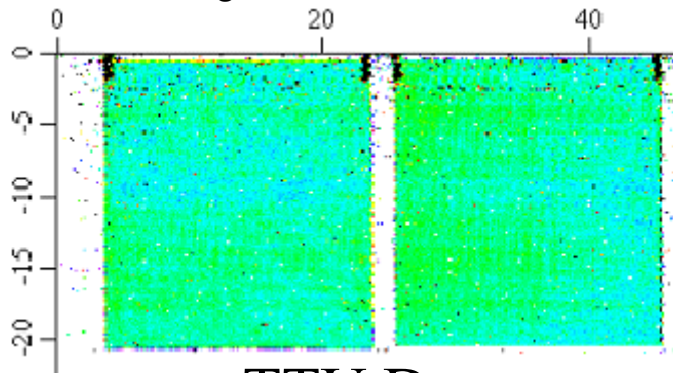


# Applications – Composite Smart Skin

## Previous Program Research

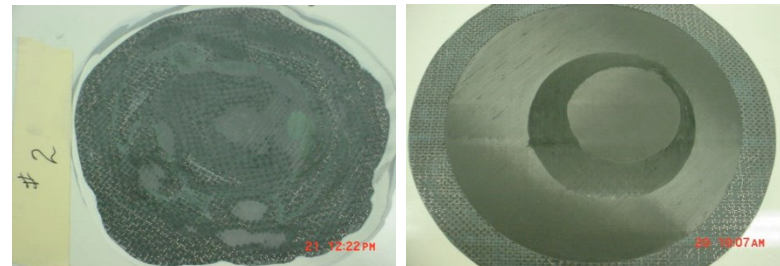
### Lightning Experiments – Damage Assessment

- **Pre-Test NDI:** Inspect the cured panel with a through transmission ultrasonic (TTU)'C' scan inspection at 5 MHz. A defect is defined as an area with ultrasonic attenuation that is at least 6db larger than the attenuation of the adjacent areas without flaws greater than 0.50 inch in diameter. Multiple defects are allowed, up to three (3) per panel as long as they are at least 1.00 inch apart.
- **Post-Test NDI:** Pulse-Echo Ultrasonic Assessment, X-ray Assessment (Optional)
  - Areas with ultrasonic attenuation >6dB larger than adjacent areas without flaws or defects.
- **Destructive Damage Assessments**
  - Taper sand to remove damaged material, use P-E Ultrasonics to verify. Then count plies remaining.



TTU Data

Photos from NASA Document Number: DOC-128697



Damage Assessment

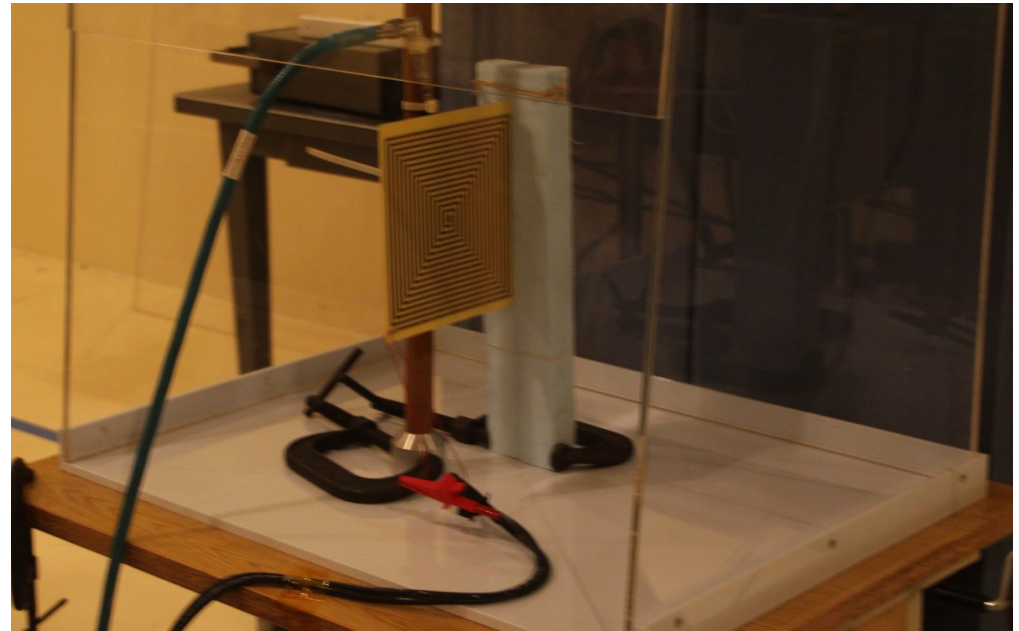
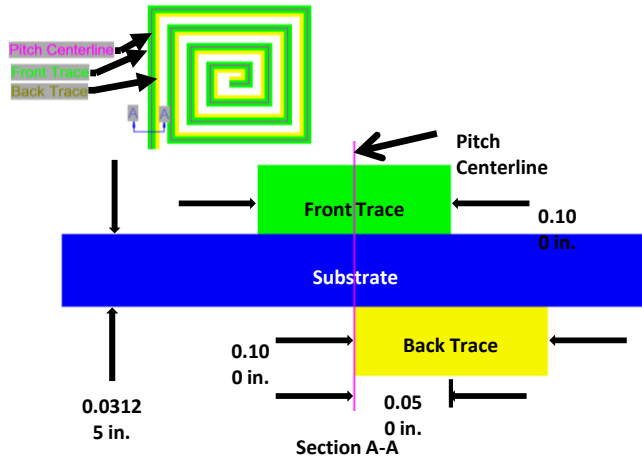


# Applications – High Efficiency Plasma Generator

Previous IRAD Research

## Dielectric Barrier Discharge (DBD) SansEC Sensor

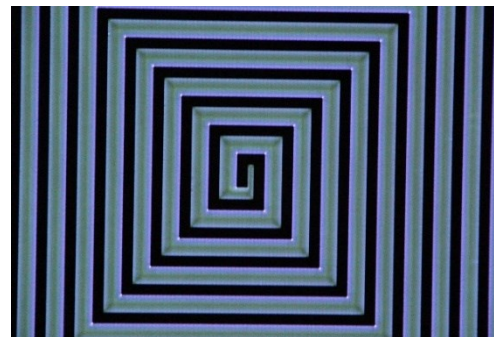
Double sided SanEC Design for DBD Test



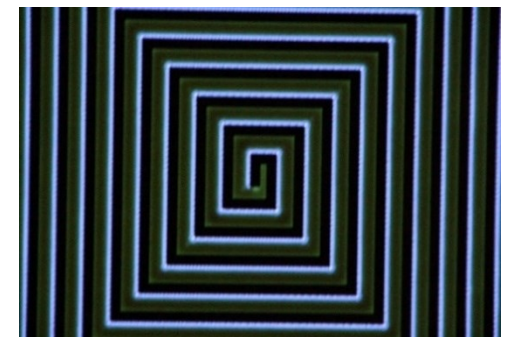
DBD Sensor Mounted inside Ventilation Hood



Experimental Setup



Response at 2KHz



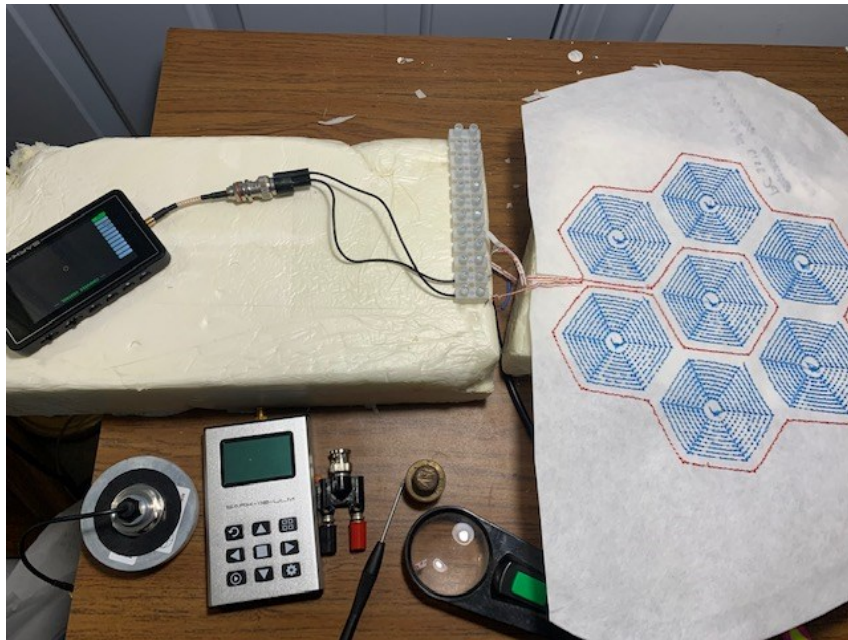
Response at 10KHz



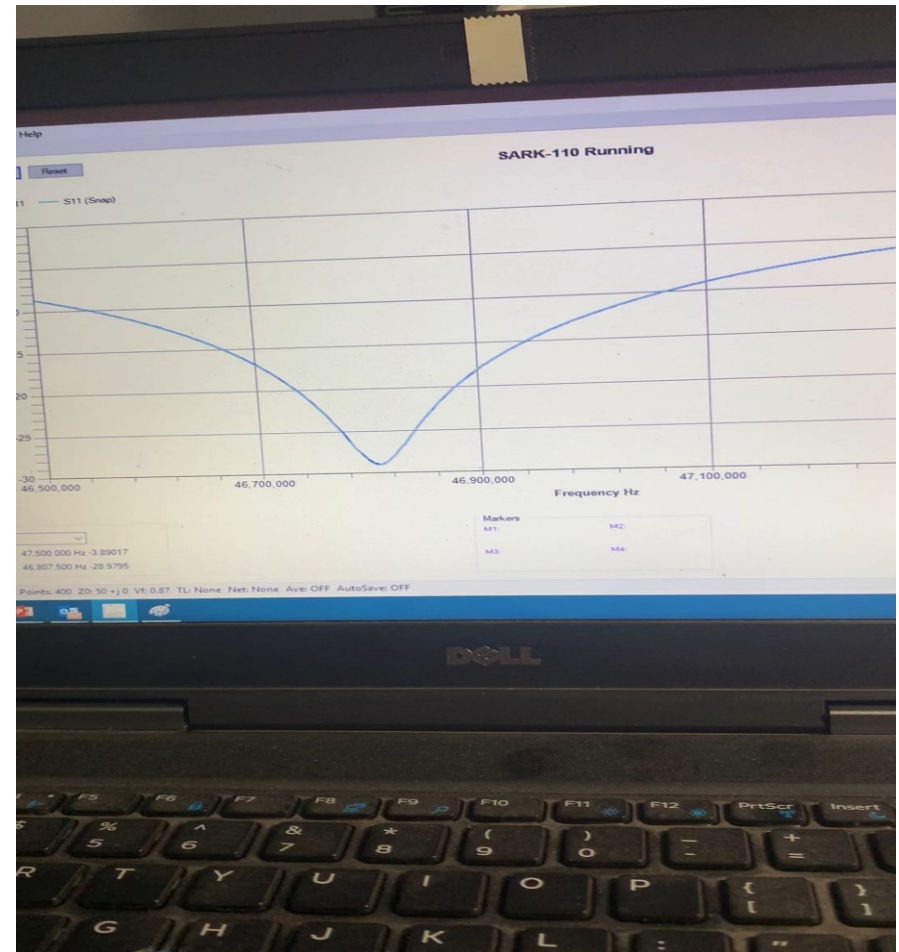
# Applications - Wearables

## Conductive Yarn Sensors Embedded in Textiles

Video showing instantaneous temperature change



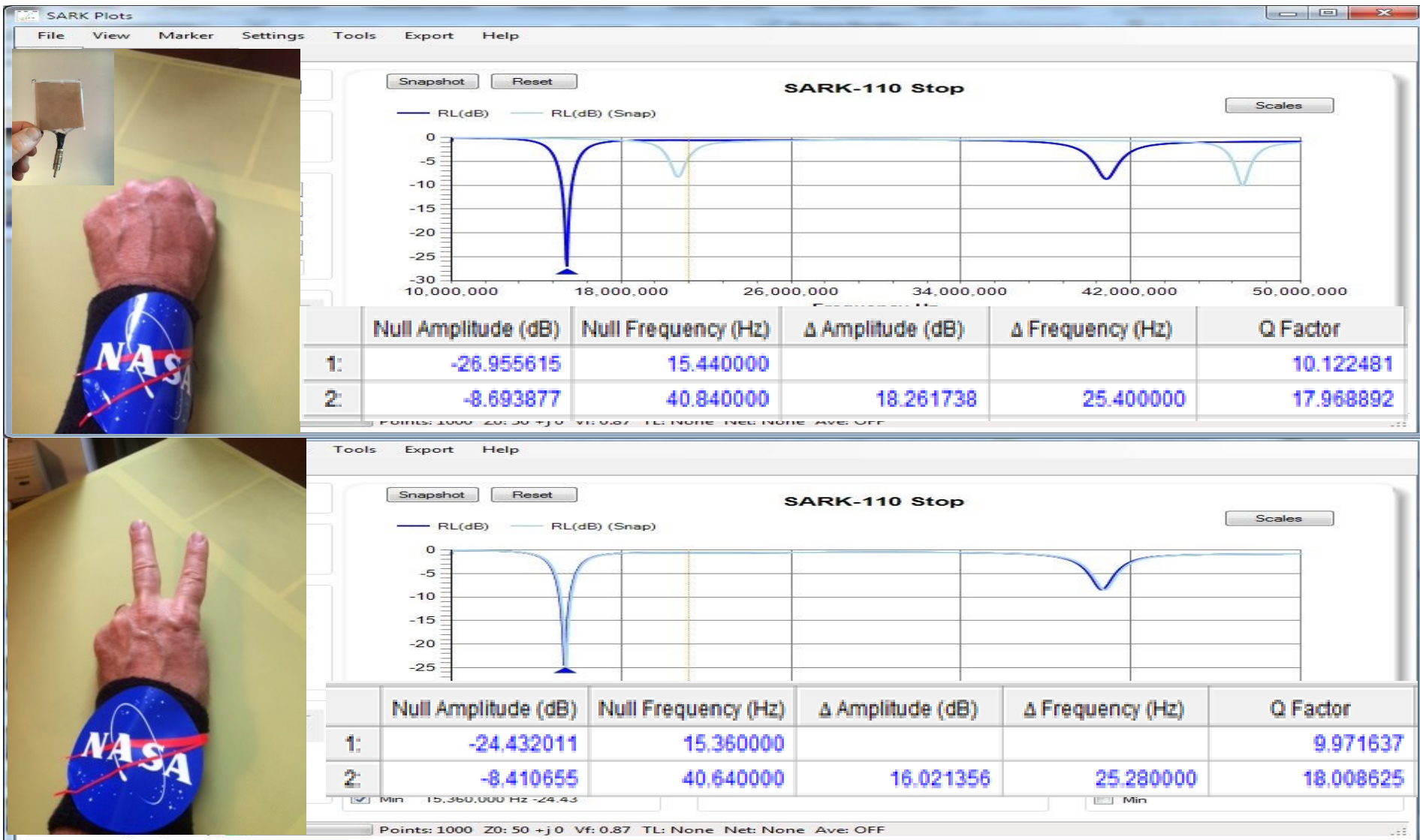
Experimental Setup





# Applications - Wearables

## Hand Gesture Measurements





# Detectable Phenomena

If the monitored phenomena undergoes a change in its electrical impedance, SansEC sensing can possibly detect it.

- Structures & Materials
  - Composite Damage
  - Stress & Strains
  - Flutter/Morphing
- Physiological Measurements
  - Vital Signs
  - Body Mechanics
  - Intercranial Fluids
- Chemistry
  - PH Level
  - Contaminations
  - Water/Ice Detection
  - Mineral Exploration
- Habitat Environmental Conditions
  - Temperature
  - Humidity
  - Pressure
- Rocket/Jet Engines
  - Fuel Cavitation
  - Performance Monitoring
  - Turbine Blade Monitoring
- Autonomous Assembly
  - Proximity
  - Angular Rotation
  - Fastener Torque
  - Integrity Monitoring



# Conclusions

## **SansEC Sensors**

- **Wireless**
- **Passive**
- **Non-Contact**
- **Electromagnetic Resonators**
  
- **Used to detect change in a physical phenomena by measuring/monitoring electrical impedance**
  
- **Measurement components include the sensor, interrogator, loop antenna and signal processing**
  
- **The physics-based data features work well in machine learning algorithms for quantification**



# Reference Publications

Szatkowski, G. N., et al, “Open Circuit Resonant (SansEC) Sensor Technology for Lightning Mitigation and Damage Detection and Diagnosis for Composite Aircraft Applications”, NASA TP 2014-218554

<https://www.worldcat.org/title/open-circuit-resonant-sansec-sensor-technology-for-lightning-mitigation-and-damage-detection-and-diagnosis-for-composite-aircraft-applications/oclc/916508919>

Woodard, S. E., et al, “SansEC sensing technology — A new tool for designing space systems and components”, IEEE 2011 Aerospace Conference <https://ieeexplore.ieee.org/document/5747495>

Wang, C., et al, “Open Circuit Resonant (SansEC) Sensor for Composite Damage Detection and Diagnosis in Aircraft Lightning Environments”, 4<sup>th</sup> AIAA Atmospheric and Space Environments Conference, June 25-28, 2012, New Orleans, LA <https://ntrs.nasa.gov/archive/nasa/casi.ntrs.nasa.gov/20120011948.pdf>

Delelegn, D. T. et al, “Data Driven Understanding of SansEC Sensor Using Random Forest”, NASA study [https://tpsas.larc.nasa.gov/lf99\\_view.cfm?lf99\\_id=34315&lf99\\_action=view](https://tpsas.larc.nasa.gov/lf99_view.cfm?lf99_id=34315&lf99_action=view)

Cluff , Kim, et al, “Passive Self Resonant Skin Patch Sensor to Monitor Intraventricular Volume using Electromagnetic Properties of Fluid Volume Changes” <https://ieeexplore.ieee.org/document/8472269>

Griffith, Jacob, et al, “Non-invasive Electromagnetic Skin Patch Sensor to Measure Intracranial Fluid-Volume Shifts” <https://ieeexplore.ieee.org/document/7516261>