



# **CFD 101: OVERVIEW**

**NASA LANGLEY/AMES EDL SEMINAR SERIES FOR SUMMER  
INTERNS**

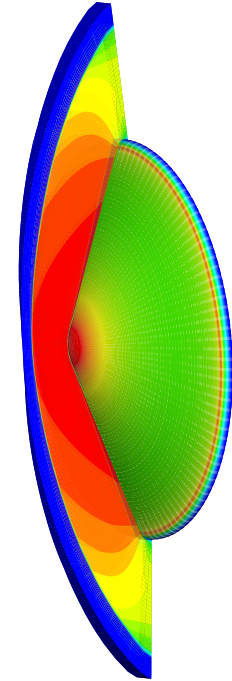
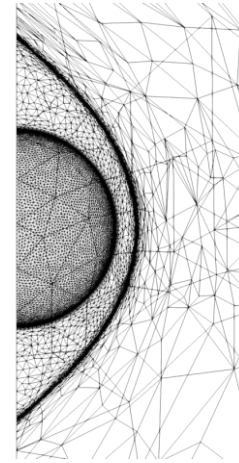
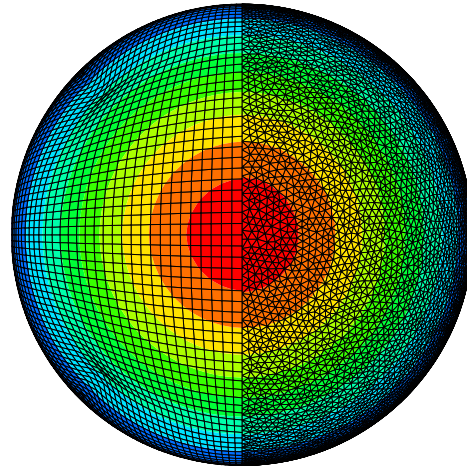
Kyle Thompson, [kyle.b.thompson@nasa.gov](mailto:kyle.b.thompson@nasa.gov)



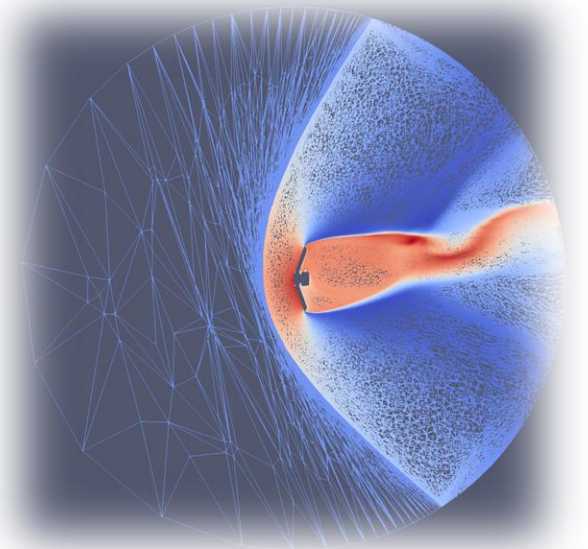
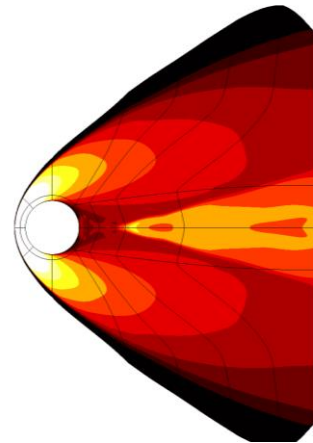
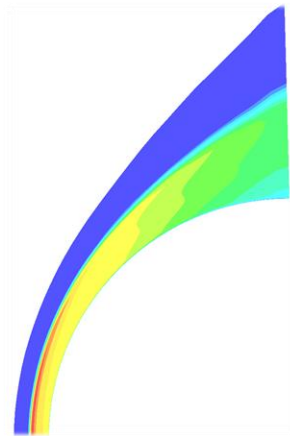
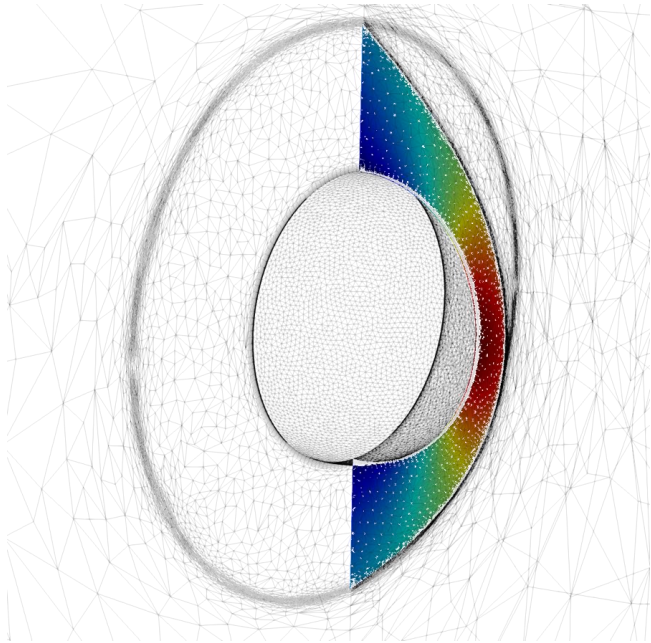
# Learning Objectives

- What is CFD?
- CFD History
- CFD Ingredients
  - Partial Differential Equation
  - Mesh
  - Discretization
  - Boundary Conditions
- CFD Resources

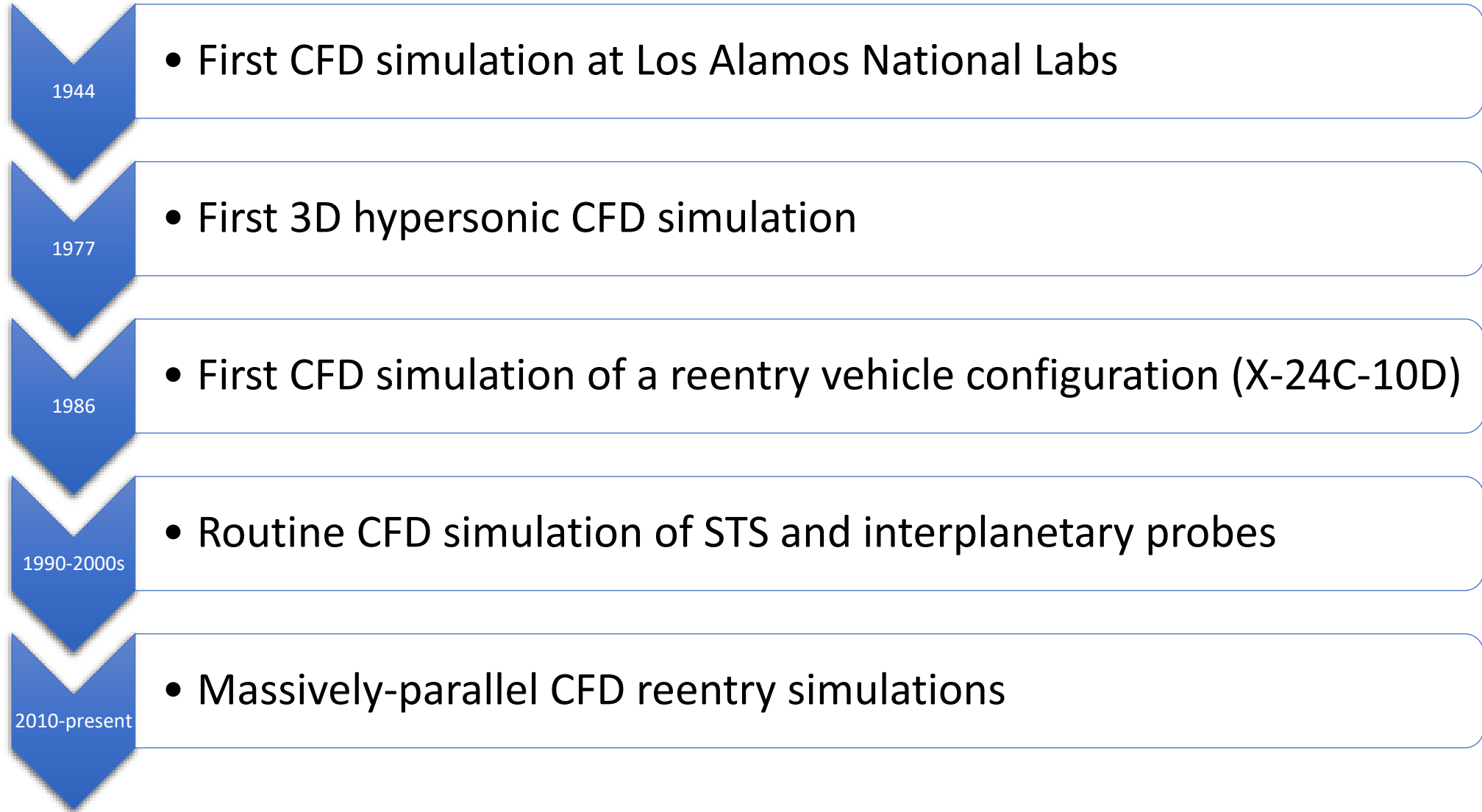
# What is CFD?



**Computational Fluid Dynamics:** The branch of fluid mechanics devoted to algorithms designed to simulate fluid flow.



# A Brief History of CFD



Ingredients  
of a CFD  
code

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**Partial Differential Equation**

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**Mesh**

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**Discretization**

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**Boundary Conditions**

**Mass Conservation :**

$$\frac{\partial \rho}{\partial t} + \frac{\partial (\rho u)}{\partial x} + \frac{\partial (\rho v)}{\partial y} + \frac{\partial (\rho w)}{\partial z} = 0$$

**Momentum Conservation :**

$$\frac{\partial (\rho u)}{\partial t} + \frac{\partial (\rho u^2)}{\partial x} + \frac{\partial (\rho uv)}{\partial y} + \frac{\partial (\rho uw)}{\partial z} = -\frac{\partial p}{\partial x} + \frac{\partial \tau_{xx}}{\partial x} + \frac{\partial \tau_{xy}}{\partial y} + \frac{\partial \tau_{xz}}{\partial z}$$

$$\frac{\partial (\rho v)}{\partial t} + \frac{\partial (\rho vu)}{\partial x} + \frac{\partial (\rho v^2)}{\partial y} + \frac{\partial (\rho vw)}{\partial z} = -\frac{\partial p}{\partial y} + \frac{\partial \tau_{yx}}{\partial x} + \frac{\partial \tau_{yy}}{\partial y} + \frac{\partial \tau_{yz}}{\partial z}$$

$$\frac{\partial (\rho w)}{\partial t} + \frac{\partial (\rho wu)}{\partial x} + \frac{\partial (\rho wv)}{\partial y} + \frac{\partial (\rho w^2)}{\partial z} = -\frac{\partial p}{\partial z} + \frac{\partial \tau_{zx}}{\partial x} + \frac{\partial \tau_{zy}}{\partial y} + \frac{\partial \tau_{zz}}{\partial z}$$

**Total Energy Conservation :**

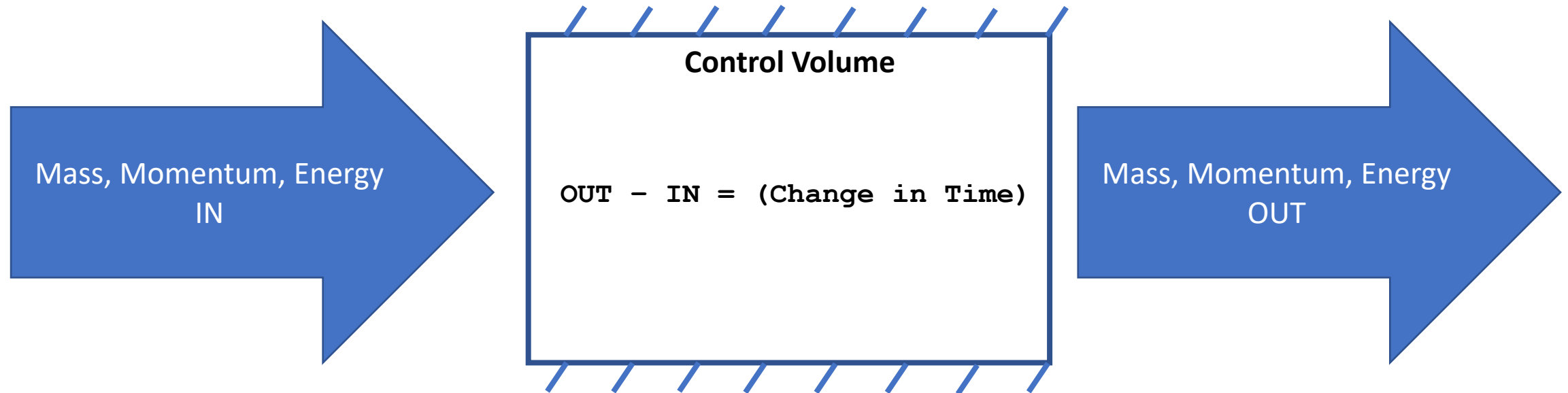
$$\frac{\partial \rho E}{\partial t} + \frac{\partial (\rho E + p) u}{\partial x} + \frac{\partial (\rho E + p) v}{\partial y} + \frac{\partial (\rho E + p) w}{\partial z} = \frac{\partial q_x}{\partial x} + \frac{\partial q_y}{\partial y} + \frac{\partial q_z}{\partial z}$$
$$\frac{\partial (u\tau_{xx} + v\tau_{xy} + w\tau_{xz})}{\partial x} + \frac{\partial (u\tau_{yx} + v\tau_{yy} + w\tau_{yz})}{\partial y} + \frac{\partial (u\tau_{zx} + v\tau_{zy} + w\tau_{zz})}{\partial z}$$

**The Navier-Stokes Equations**

# CFD Ingredient: Partial Differential Equation

At the core of a CFD code is a set of conservation equations

# Conservation Equations



- Based on fundamental laws of nature
  - Mass cannot be created or destroyed (Lomonosov)
  - Momentum must be conserved (Newton's Second Law)
  - Energy cannot be created or destroyed (First Law of Thermodynamics)

# Extension to Reacting Flow

Gas Mixture: Multiple “species” mass equations

Many other terms may be needed, due to ablation, shock-layer radiation, etc.

Thermodynamic nonequilibrium add an additional equation for vibrational energy carried by the gas mixture

**Species Conservation :**

$$\frac{\partial \rho_1}{\partial t} + \frac{\partial (\rho_1 u)}{\partial x} + \frac{\partial (\rho_1 v)}{\partial y} + \frac{\partial (\rho_1 w)}{\partial z} = \frac{\partial D_m}{\partial x} + \frac{\partial D_m}{\partial y} + \frac{\partial D_m}{\partial z} + \omega_1$$

⋮

$$\frac{\partial \rho_N}{\partial t} + \frac{\partial (\rho_N u)}{\partial x} + \frac{\partial (\rho_N v)}{\partial y} + \frac{\partial (\rho_N w)}{\partial z} = \frac{\partial D_m}{\partial x} + \frac{\partial D_m}{\partial y} + \frac{\partial D_m}{\partial z} + \omega_N$$

**Mixture Momentum Conservation :**

$$\frac{\partial (\rho u)}{\partial t} + \frac{\partial (\rho u^2)}{\partial x} + \frac{\partial (\rho uv)}{\partial y} + \frac{\partial (\rho uw)}{\partial z} = -\frac{\partial p}{\partial x} + \frac{\partial \tau_{xx}}{\partial x} + \frac{\partial \tau_{xy}}{\partial y} + \frac{\partial \tau_{xz}}{\partial z}$$

$$\frac{\partial (\rho v)}{\partial t} + \frac{\partial (\rho vu)}{\partial x} + \frac{\partial (\rho v^2)}{\partial y} + \frac{\partial (\rho vw)}{\partial z} = -\frac{\partial p}{\partial y} + \frac{\partial \tau_{yx}}{\partial x} + \frac{\partial \tau_{yy}}{\partial y} + \frac{\partial \tau_{yz}}{\partial z}$$

$$\frac{\partial (\rho w)}{\partial t} + \frac{\partial (\rho wu)}{\partial x} + \frac{\partial (\rho wv)}{\partial y} + \frac{\partial (\rho w^2)}{\partial z} = -\frac{\partial p}{\partial z} + \frac{\partial \tau_{zx}}{\partial x} + \frac{\partial \tau_{zy}}{\partial y} + \frac{\partial \tau_{zz}}{\partial z}$$

**Total Energy Conservation :**

$$\begin{aligned} \frac{\partial \rho E}{\partial t} + \frac{\partial (\rho E + p) u}{\partial x} + \frac{\partial (\rho E + p) v}{\partial y} + \frac{\partial (\rho E + p) w}{\partial z} &= \frac{\partial q_x}{\partial x} + \frac{\partial q_y}{\partial y} + \frac{\partial q_z}{\partial z} \\ + \frac{\partial (u\tau_{xx} + v\tau_{xy} + w\tau_{xz})}{\partial x} + \frac{\partial (u\tau_{yx} + v\tau_{yy} + w\tau_{yz})}{\partial y} + \frac{\partial (u\tau_{zx} + v\tau_{zy} + w\tau_{zz})}{\partial z} \\ &+ \frac{\partial D_e}{\partial x} + \frac{\partial D_e}{\partial y} + \frac{\partial D_e}{\partial z} - Q_{rad} \end{aligned}$$

**Vibrational Energy Conservation :**

$$\begin{aligned} \frac{\partial \rho E}{\partial t} + \frac{\partial (\rho e_v) u}{\partial x} + \frac{\partial (\rho e_v) v}{\partial y} + \frac{\partial (\rho e_v) w}{\partial z} &= \frac{\partial q_x^v}{\partial x} + \frac{\partial q_y^v}{\partial y} + \frac{\partial q_z^v}{\partial z} \\ + \frac{\partial (u\tau_{xx} + v\tau_{xy} + w\tau_{xz})}{\partial x} + \frac{\partial (u\tau_{yx} + v\tau_{yy} + w\tau_{yz})}{\partial y} + \frac{\partial (u\tau_{zx} + v\tau_{zy} + w\tau_{zz})}{\partial z} \\ &+ \frac{\partial D_v}{\partial x} + \frac{\partial D_v}{\partial y} + \frac{\partial D_v}{\partial z} + \omega_v \end{aligned}$$

# Turbulence

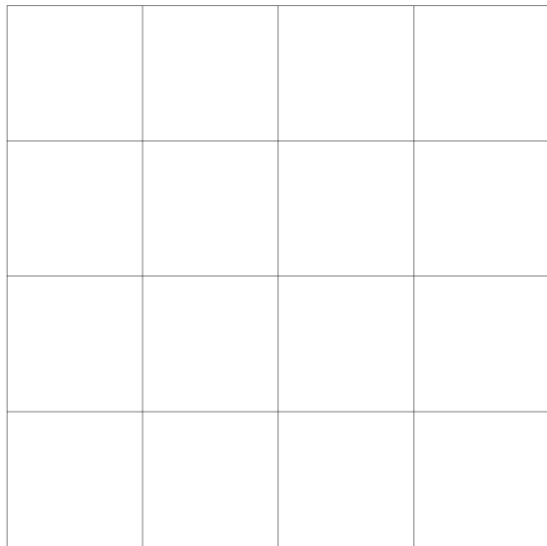
*"When I meet God, I am going to ask him two questions: Why relativity? And why turbulence? I really believe he will have an answer for the first."* - **Werner Heisenberg**

- Caused by chaotic motion of fluid
- Three types of methods to model turbulence
  - Reynold's Averaged Navier Stokes (RANS): Average all velocity fluctuations
  - Large Eddy-Simulation (LES): Resolve largest eddies in flow
  - Direct Numerical Simulation (DNS): Resolve all eddies
- RANS turbulence modeling is the most common and least expensive approach, but also the least accurate
- LES is an active area of research, requiring highly accurate and efficient solvers
- DNS is prohibitively expensive for almost all EDL applications

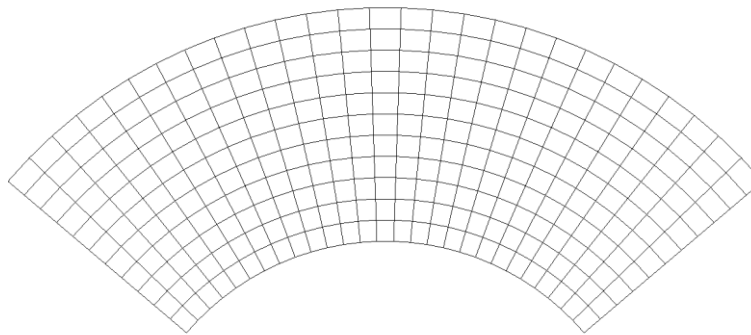
## CFD Ingredient: Mesh

- Need to represent the solution discretely
- Mesh is used to define the geometry and the domain where we need the solution
- Meshing techniques will be covered in a follow-on presentation

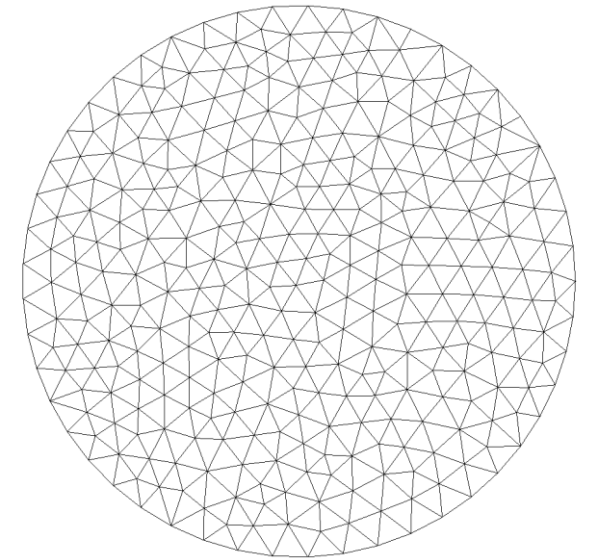
### Mesh Types



Cartesian Mesh



Structured Mesh



Unstructured Mesh

CFD

Ingredient:

Discretization

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## Differential Form

Directly approximate the derivatives

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## Integral Form

Integrate the PDE and then solve

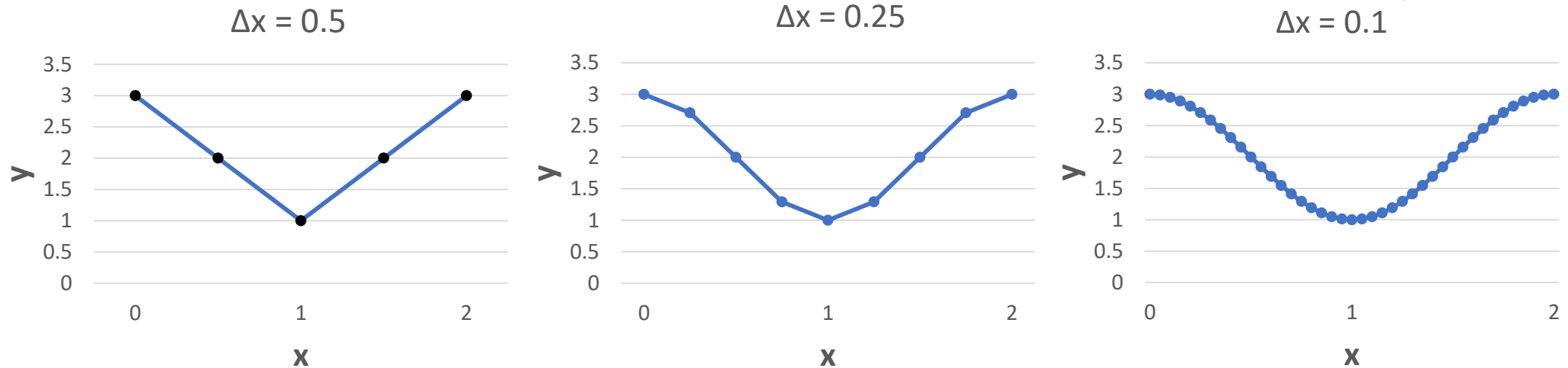
# Sidebar: What is Discretization?

- Discretization is the process of representing a continuous function as discrete data.

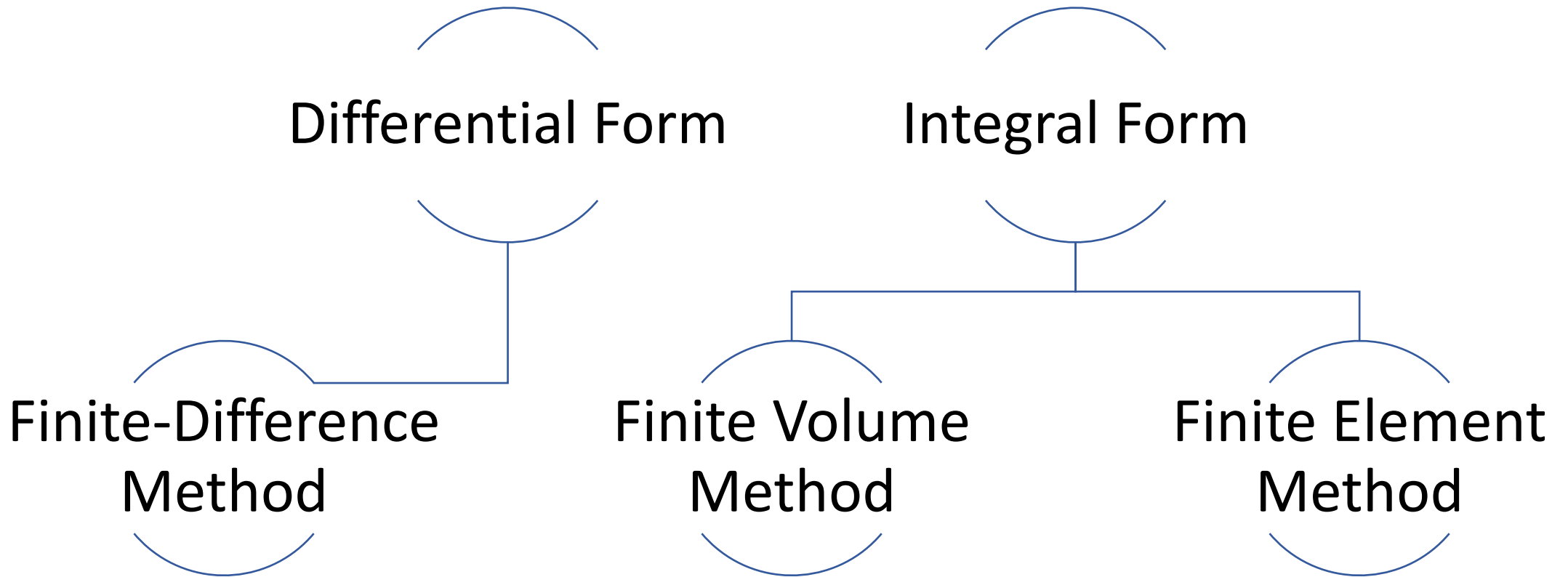
## Example Function

$$y = \cos(\pi x) + 2$$

Adding points better represents the continuous function



# Spatial Discretization Types



# Spatial Discretization: Finite Difference Method

## Finite-Difference Derivative Approximation

$$\frac{\partial y}{\partial x} = \frac{y_{i+1} - y_i}{x_{i+1} - x_i}$$

- Solve equations in differential form
- Derivatives in the equations approximated directly using finite-difference formulas
- Pros: Simple to implement and can be made highly accurate
- Cons: Care must be taken to ensure this method maintains conservation
- This approach is less common in current NASA EDL CFD codes, but still used
  - The OVERFLOW CFD code uses this method

## Divergence Formula

$$\iiint_V \frac{\partial \mathbf{F}}{\partial \mathbf{x}} dV = \iint_S \mathbf{F} \cdot \mathbf{n} dS$$

# Spatial Discretization: Finite Volume Method

- Solve equations in integral form
- Integrate the PDE and represent the solution as average of “control volumes”
- Convert control volume integrals to surface integrals via the divergence formula
- Pros: simple to implement and ensures conservation by design
- Cons: difficult to achieve high accuracy on arbitrary meshes
- The most common algorithm used by NASA EDL CFD codes
  - The DPLR, LAURA, US3D, and FUN3D codes use this algorithm

## Integration by Parts

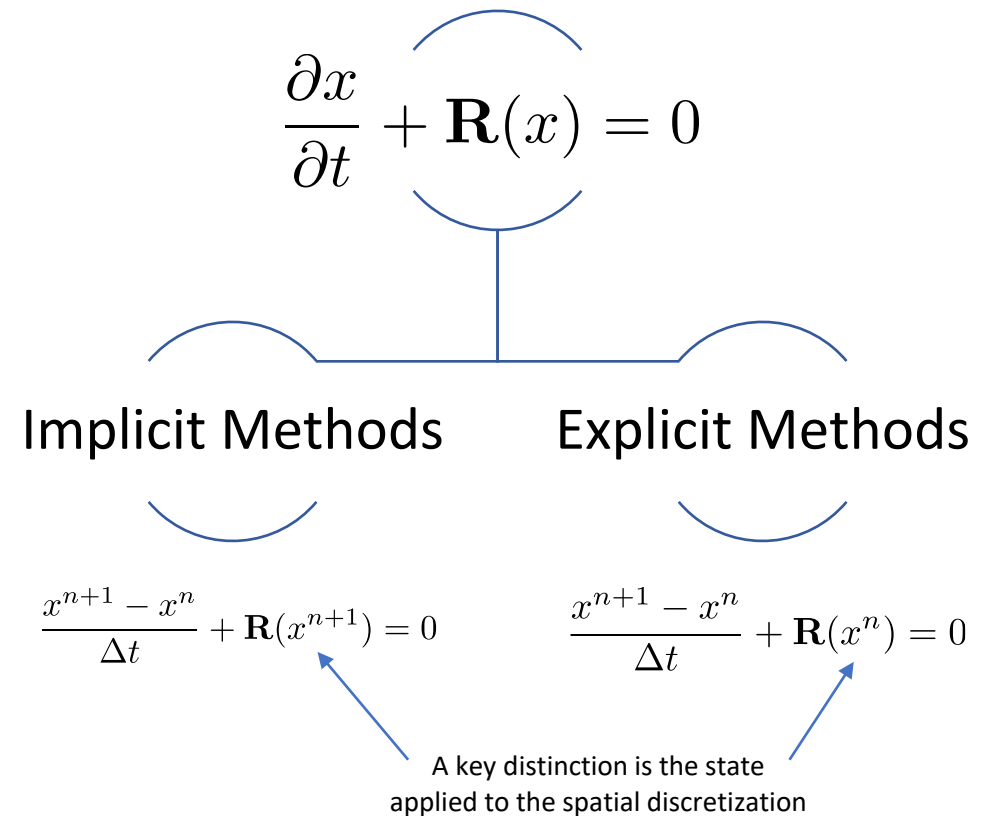
$$\int u dv = uv - \int v du$$

# Spatial Discretization: Finite Element Method

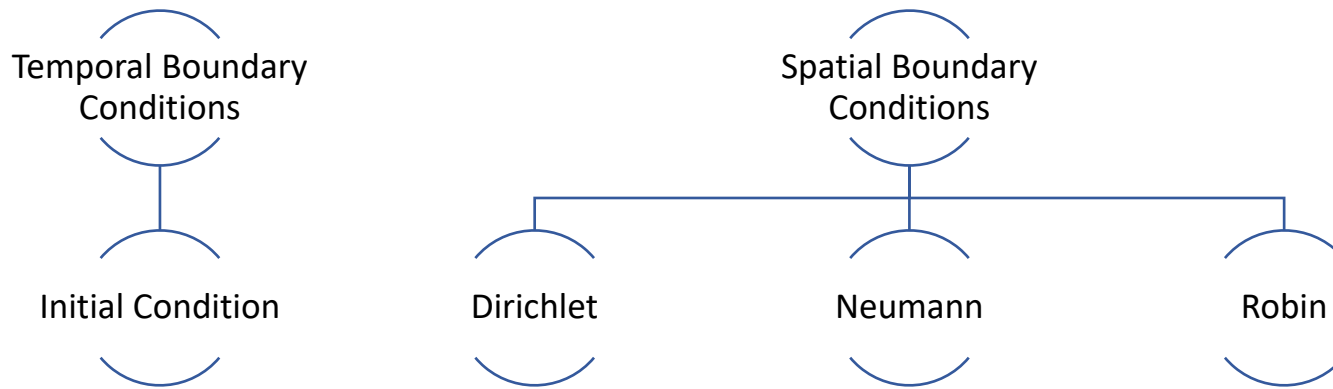
- Solve equations in integral form
- Integrate the PDE by parts and represent the solution via a “basis function”
- The solution is essentially a curve fit on each element in the mesh
- Pros: Highly accurate
- Cons: Relatively complex and can be computationally expensive
- This method is commonly used for turbulence modeling research requiring high accuracy (LES/DNS simulations)

# Temporal Discretization

- This is the part responsible for evolving the solution
- The CFD solver iterates until enough time history/statistics have accumulated
- Explicit vs Implicit
  - Explicit = simple, but restricted to small timesteps
  - Implicit = larger timesteps, but requires solving a linear system
- Steady State – no time dependence

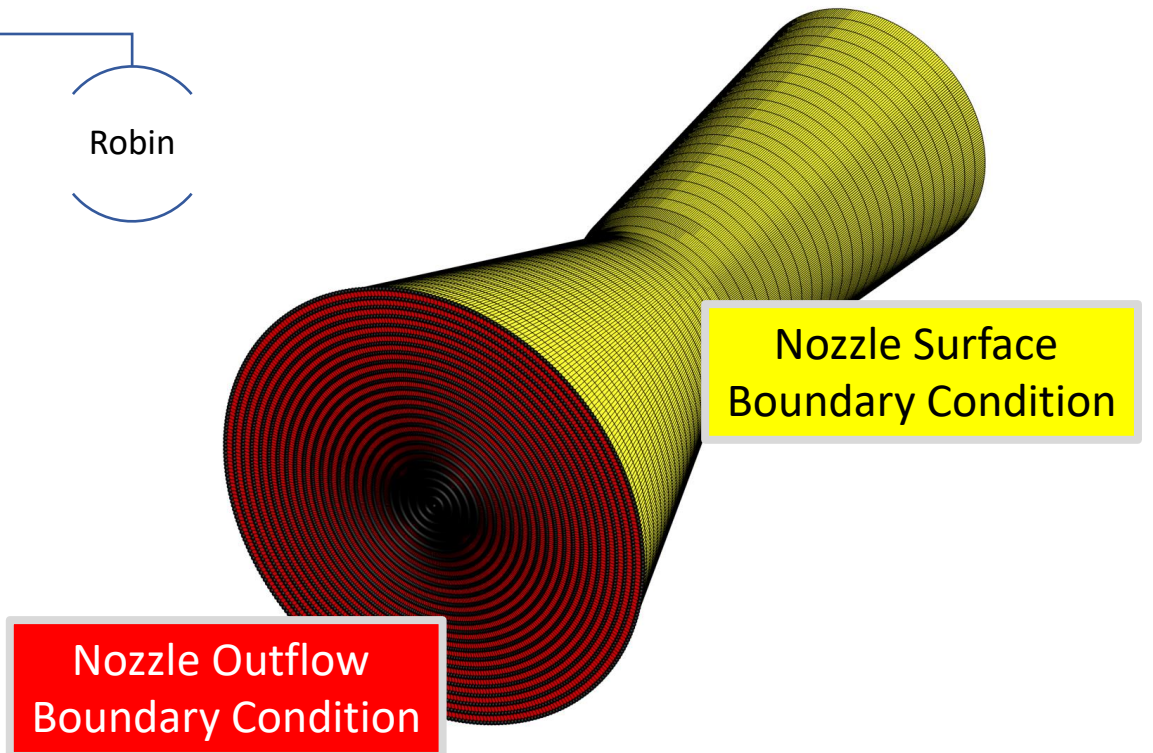


# Boundary Conditions



- The final piece needed to define a CFD simulation
- These are defined by the user and associated with parts of the mesh

Example Nozzle Mesh



# CFD Resources

- This has barely scratched the surface of CFD, but there are many great resources available on this subject:
  - Anderson, John D. *Computational Fluid Dynamics: the Basics with Applications*.
  - Anderson, Dale A., et al. *Computational Fluid Mechanics and Heat Transfer*.
  - Press, William H., et al. *Numerical Recipes: the Art of Scientific Computing*.
  - "I Do Like CFD" - <http://www.cfdbooks.com>