

GRAVITATIONAL MULTIBODY DYNAMICS: A QUICK-START GUIDE

Course plan:

- 1) Choosing the right model
- 2) Going with the flow
- 3) Periodic motion
- 4) Stability and orbit manifolds
- 5) From CR3BP to ephemeris



Davide Guzzetti

May-June 2026, NESCA Academy Series



AUBURN

Lecture 4:

STABILITY AND ORBIT MANIFOLDS

Gravitational Multibody Dynamics: a Quick-start Guide

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A PERIODIC ORBIT AS A FIXED POINT OF THE FLOW MAP

Draw schematics



A PERIODIC ORBIT AS A FIXED POINT OF THE FLOW MAP

The flow map advances an initial state

$$\varphi_t : \mathbf{x}_0 \mapsto \mathbf{x}(t; \mathbf{x}_0)$$

A periodic orbit is a fixed point of the time-T map

$$\varphi_T(\mathbf{x}_0) = \mathbf{x}_0$$

Let Sigma be a codimension-one surface transverse to the flow

$$\Sigma = \{\mathbf{x} : h(\mathbf{x}) = 0\} \quad \text{and} \quad \nabla h(\mathbf{x})^T \mathbf{f}(\mathbf{x}) \neq 0$$

The Poincare' map advances a point on the section to its next return to the section

$$P : \Sigma \rightarrow \Sigma \quad P(\mathbf{x}_k) = \varphi_{\tau(\mathbf{x}_k)}(\mathbf{x}_k)$$

A periodic orbit that intersects the section once per period becomes a fixed point of the return map. If it intersects m times before repeating, it is a fixed point of the m-th iterate

$$P(\mathbf{x}^*) = \mathbf{x}^* \quad \text{or} \quad P^m(\mathbf{x}^*) = \mathbf{x}^*$$



VARIATIONAL DYNAMICS ALONG A PERIODIC ORBIT

Let the reference solution be periodic with period T , add a small perturbation

Linearize

Because the reference orbit is periodic, the Jacobian matrix along the orbit is also periodic

The STM over one period is the derivative of the time- T flow map at the orbit point

With the return map written in the same coordinates, this is the derivative DP at the fixed point

$$\mathbf{x}(t) = \mathbf{x}^*(t) + \delta\mathbf{x}(t) \quad \mathbf{x}^*(t + T) = \mathbf{x}^*(t)$$

$$\delta\dot{\mathbf{x}} = A(t)\delta\mathbf{x} \quad A(t) = D\mathbf{f}(\mathbf{x}^*(t))$$

$$A(t + T) = A(t)$$

$$D\varphi_T(\mathbf{x}_0^*) = \Phi(T, 0)$$

$$\dot{\Phi}(t, 0) = A(t)\Phi(t, 0), \quad \Phi(0, 0) = I$$

$$DP(\mathbf{x}_0^*) = D\varphi_T(\mathbf{x}_0^*) = \Phi(T, 0)$$



THE MONODROMY MATRIX

The monodromy matrix is the STM evaluated over one full period

$$M = \Phi(T, 0)$$

The monodromy matrix is the derivative of the period map at a periodic-orbit fixed point

$$DP(\mathbf{x}_0^*) = M = D\varphi_T(\mathbf{x}_0^*) = \Phi(T, 0)$$

It maps perturbations from one revolution to the next

$$\delta \mathbf{x}_{k+1} = M \delta \mathbf{x}_k$$

Repeated application describes the perturbation evolution over multiple periods

$$\delta \mathbf{x}_k = M^k \delta \mathbf{x}_0$$

Its eigenvalues are characteristic multipliers

$$M \mathbf{w}_i = \rho_i \mathbf{w}_i$$

Their magnitudes determine linear growth, decay, or neutral behavior over one period

$$|\rho_i| < 1 \quad |\rho_i| > 1 \quad |\rho_i| = 1$$



THE MONODROMY MATRIX

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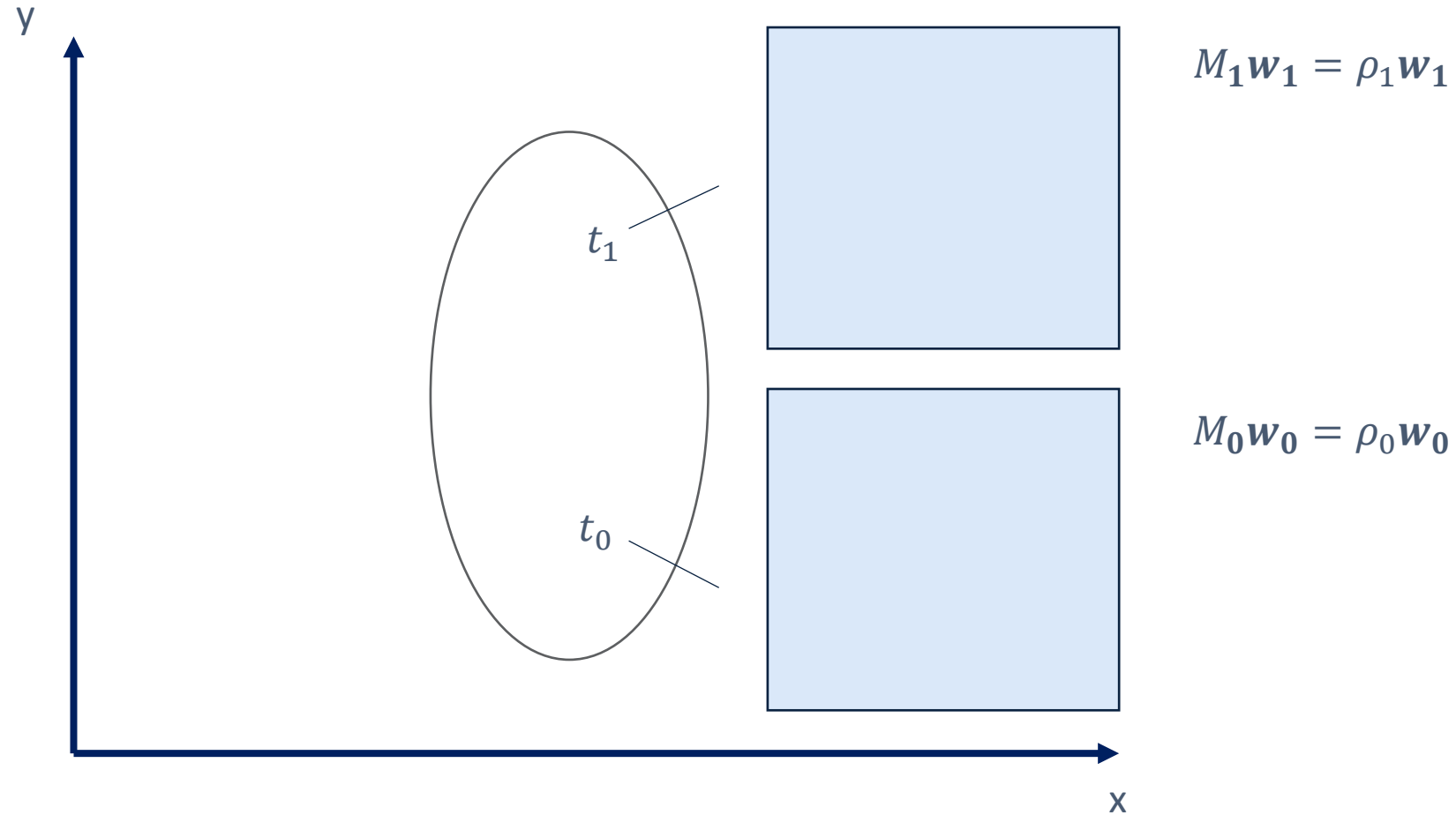
$$|\rho_i| = 1$$



$$|\rho_i| > 1$$



EIGENSTRUCTURE ALONG AN ORBIT





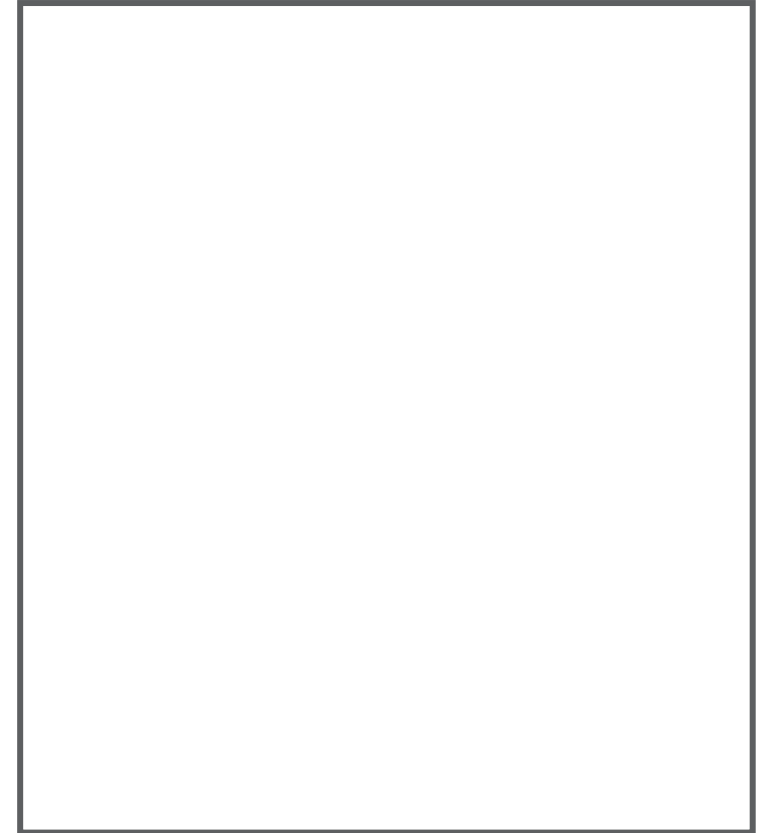
STABILITY CLASSIFICATION

- 1) $|\rho_i| < 1 \quad \forall i \rightarrow$ linearly asymptotically stable
- 2) $|\rho_i| \leq 1 \quad \forall i, \exists j: |\rho_j| = 1, \rho_j \text{ is semisimple} \rightarrow$ marginally stable
- 3) $\exists i: |\rho_i| > 1 \rightarrow$ linearly unstable

Notable fixed points

- 1) $\exists i, j: |\rho_i| > 1$ and $|\rho_j| < 1 \rightarrow$ saddle point
- 2) $\forall i |\rho_i| \neq 1 \rightarrow$ hyperbolic point (no neutral directions)

*Blank space to sample
Agar diagrams*

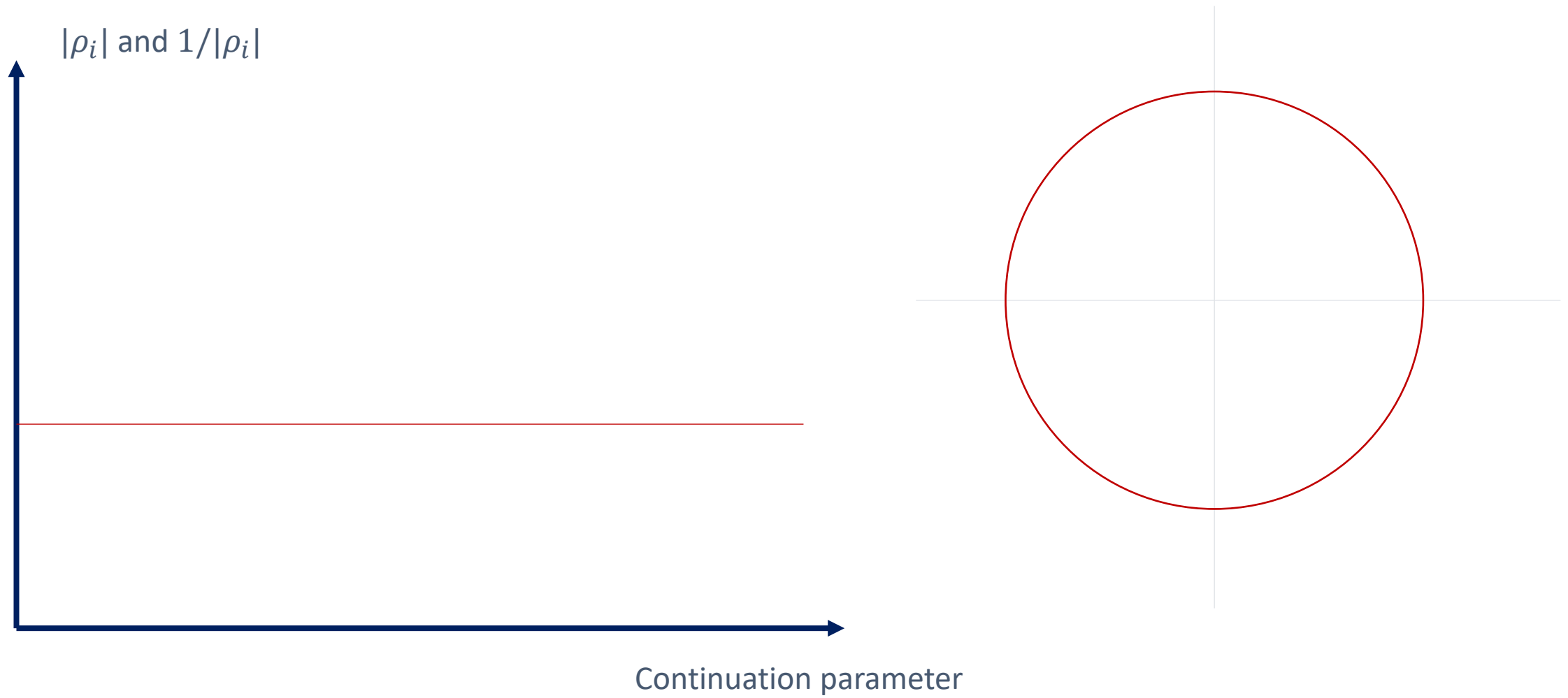




NOTABLE PROPERTY IN CR3BP

- Always a pair of unitary eigenvalues
- Always in reciprocal pairs $\left(\rho_i, \frac{1}{\rho_i}\right)$
- Complex eigenvalues always in complex conjugate pairs $(a + ib, a - ib)$

EIGENSTRUCTURE AND ORBIT BIFURCATIONS





LOCAL MANIFOLDS

Its eigenvalues are characteristic multipliers

$$M\mathbf{w}_i = \rho_i\mathbf{w}_i$$

Their magnitudes determine linear growth, decay, or neutral behavior over one period

$$|\rho_i| < 1$$



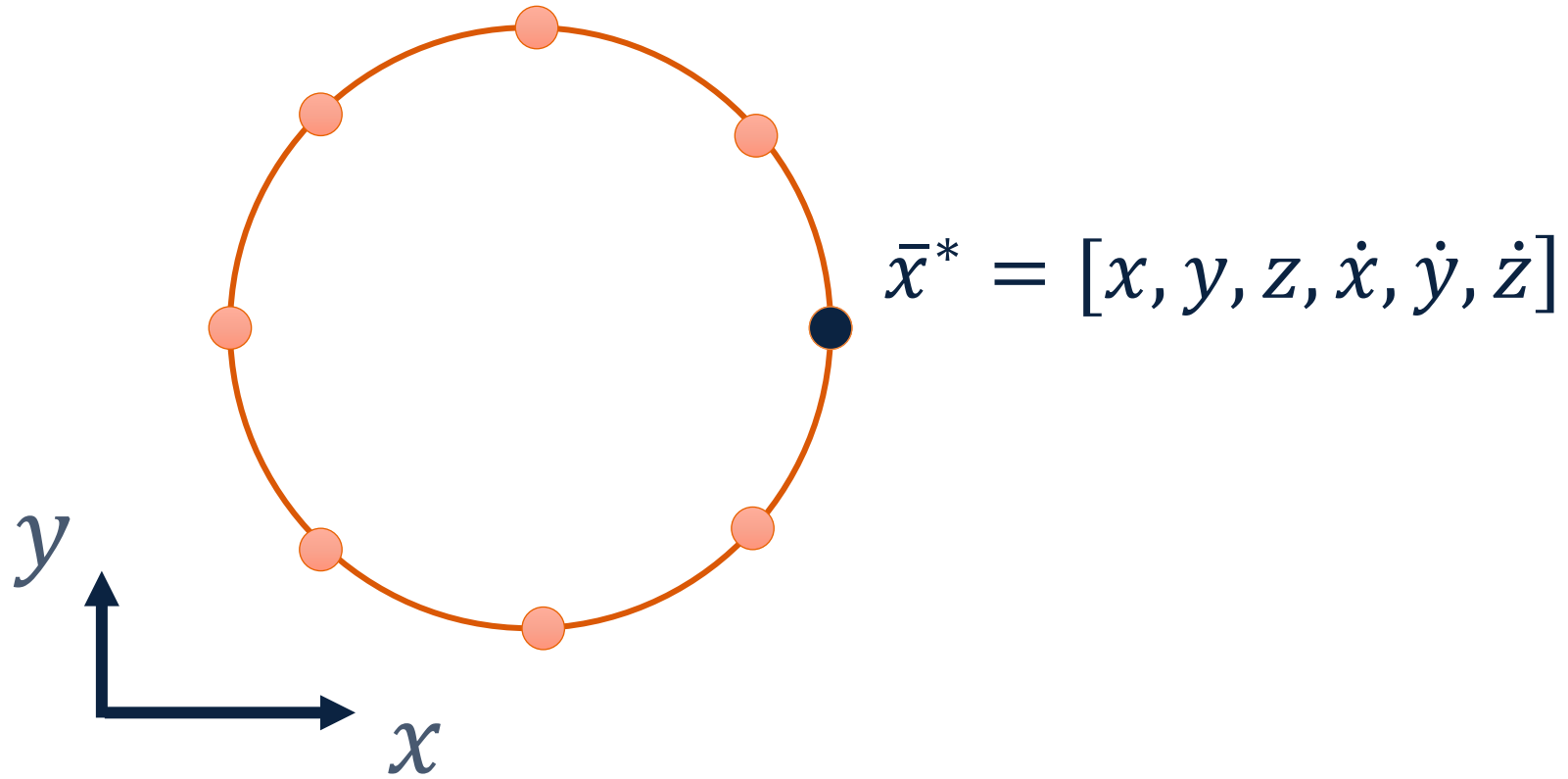
$$|\rho_i| = 1$$



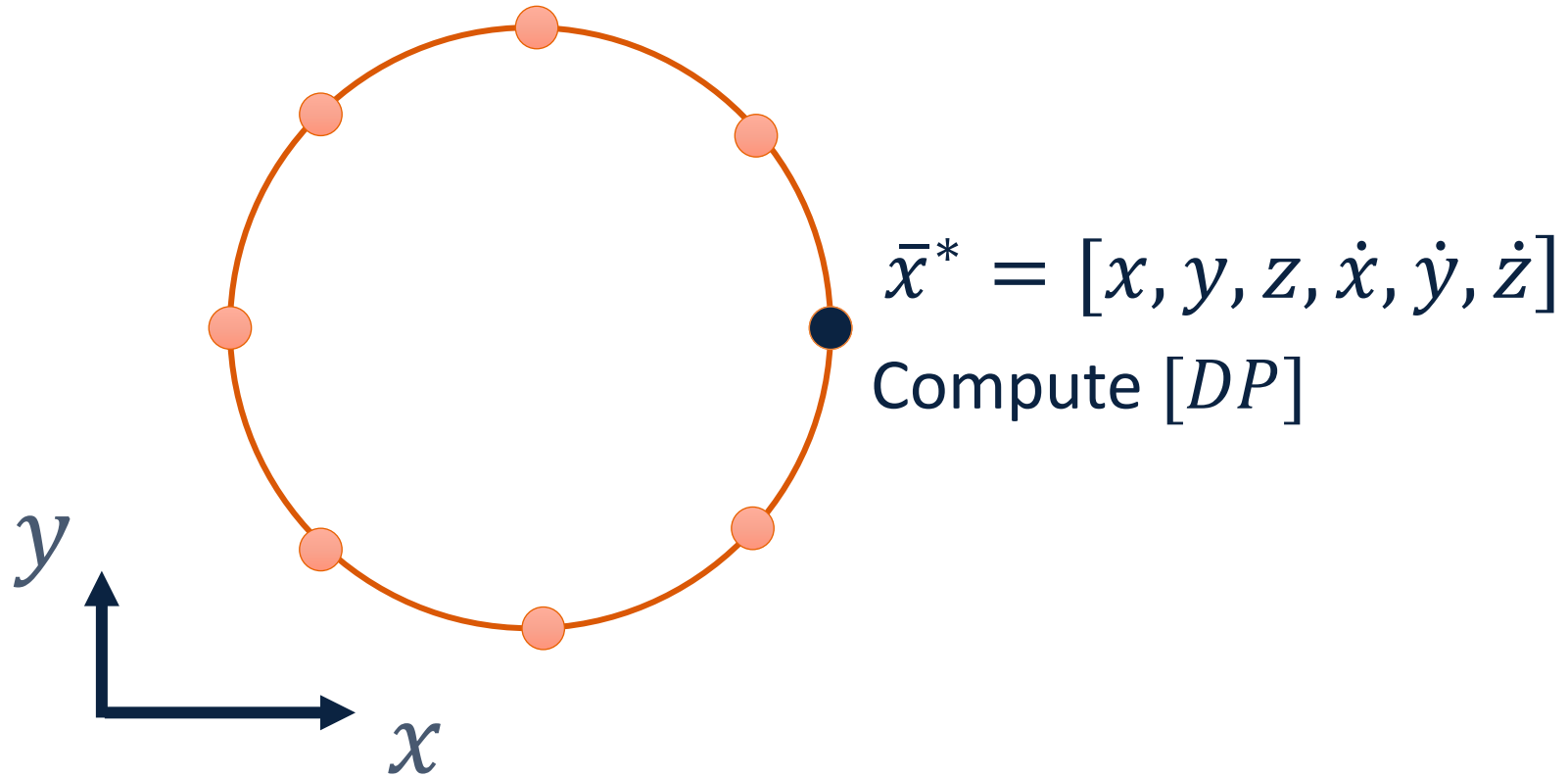
$$|\rho_i| > 1$$



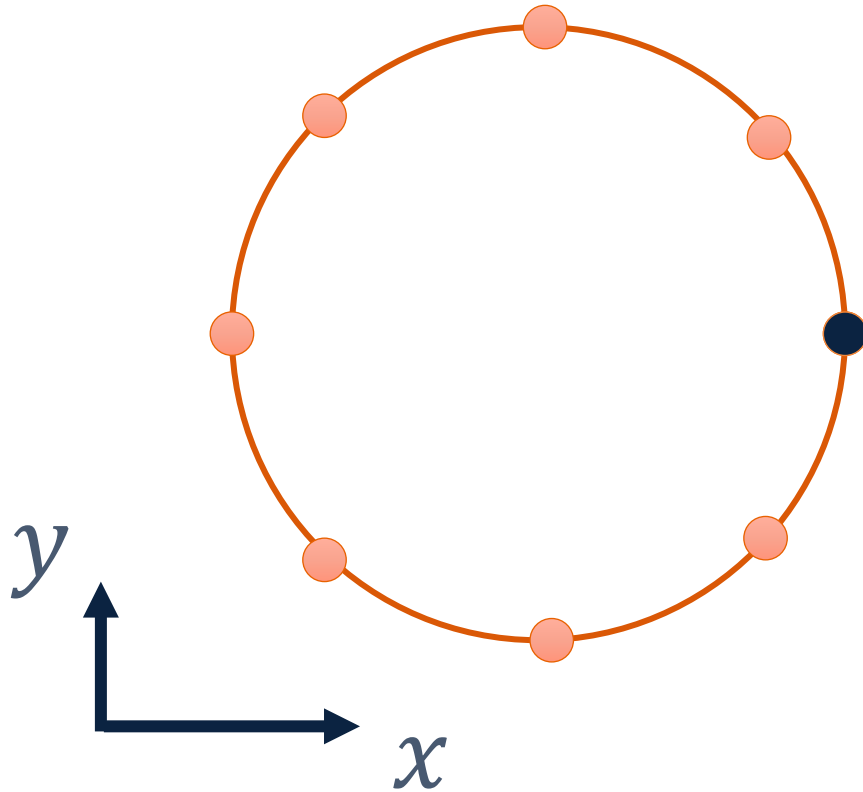
COMPUTATION OF GLOBAL STABLE/UNSTABLE MANIFOLDS



COMPUTATION OF STABLE/UNSTABLE MANIFOLDS



COMPUTATION OF STABLE/UNSTABLE MANIFOLDS



$$\bar{x}^* = [x, y, z, \dot{x}, \dot{y}, \dot{z}]$$

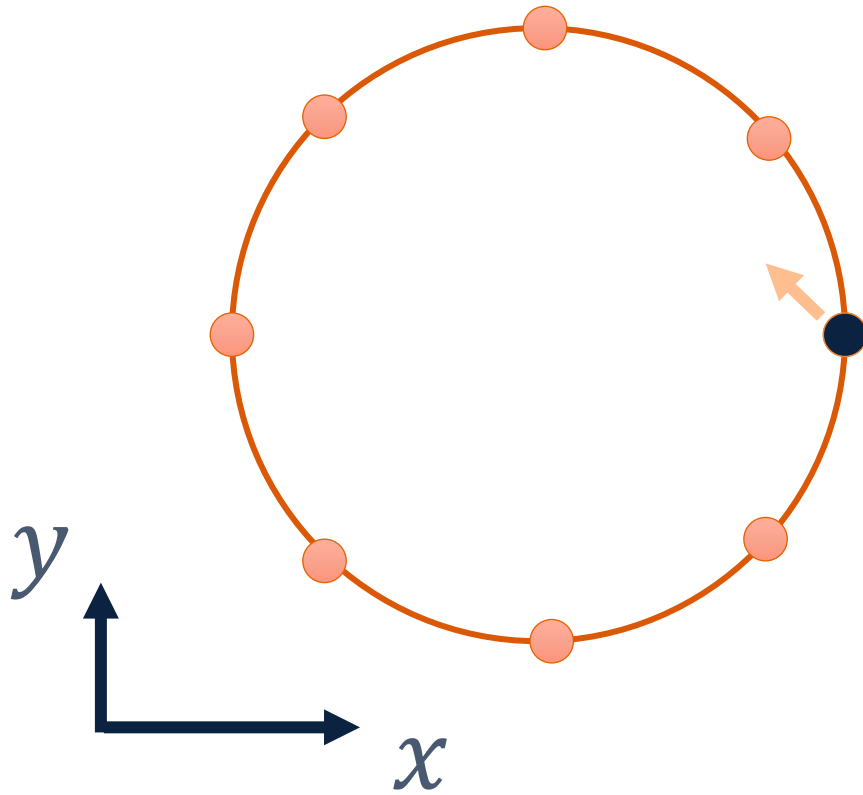
Compute $[DP]$

Compute eigenvalues
and eigenvectors

$$w_s: \quad |\rho_s| < 1$$

$$w_u: \quad |\rho_u| > 1$$

COMPUTATION OF STABLE/UNSTABLE MANIFOLDS



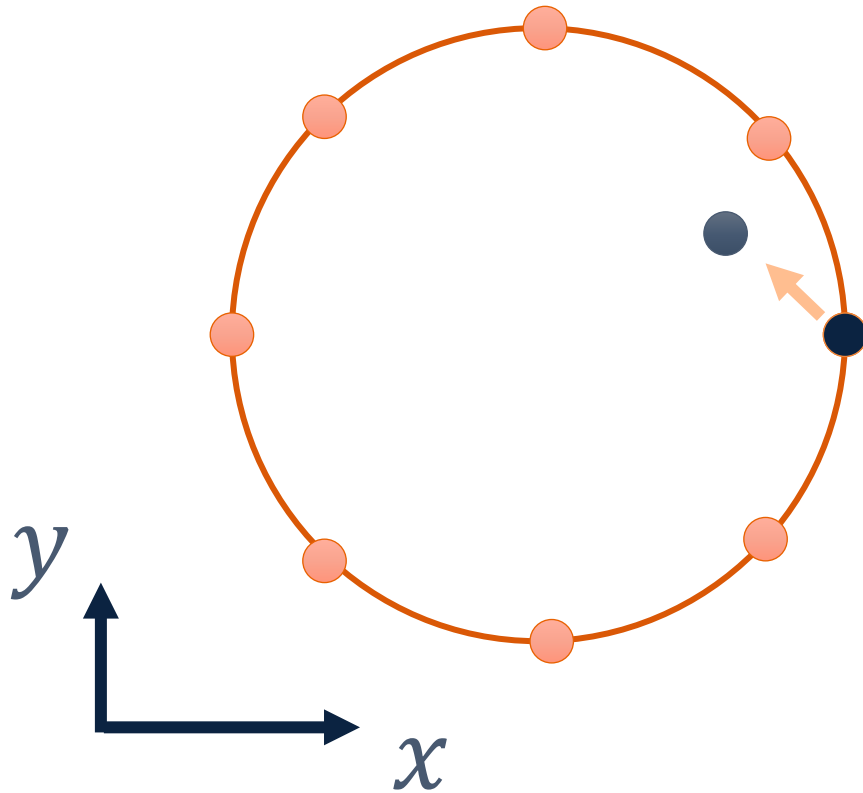
$$\bar{x}^* = [x, y, z, \dot{x}, \dot{y}, \dot{z}]$$

$$w_u: |\rho_u| > 1$$

$$w_u = [x_u, y_u, z_u, \dot{x}_u, \dot{y}_u, \dot{z}_u]$$

$$\hat{w}_u = \frac{w_u}{\sqrt{x_u^2 + y_u^2 + z_u^2}}$$

COMPUTATION OF STABLE/UNSTABLE MANIFOLDS



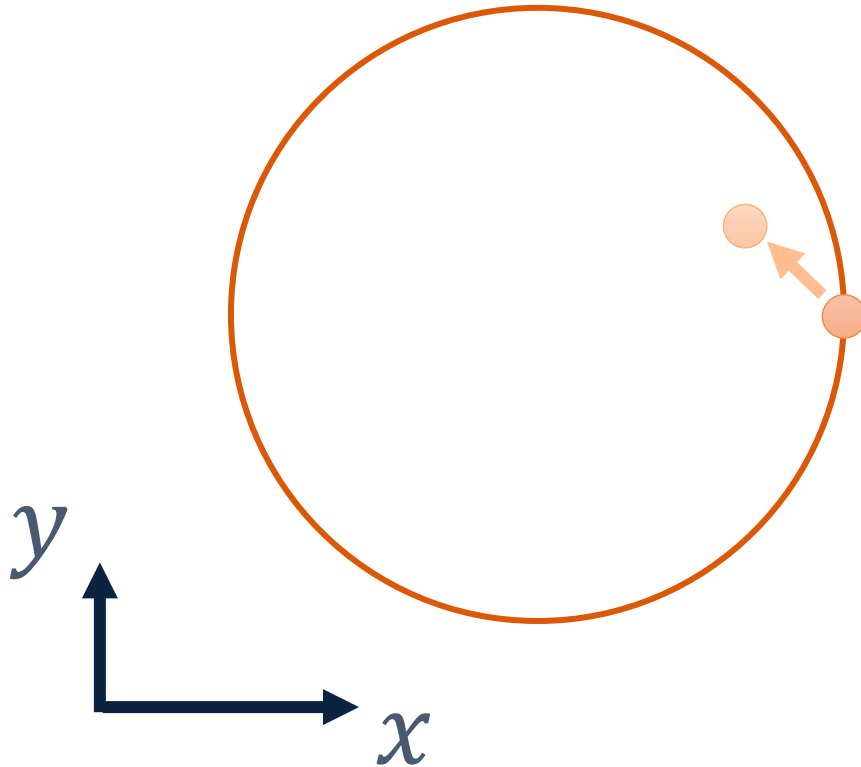
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$$\bar{x}_u = \bar{x}^* \pm d\hat{w}_u$$

COMPUTATION OF STABLE/UNSTABLE MANIFOLDS



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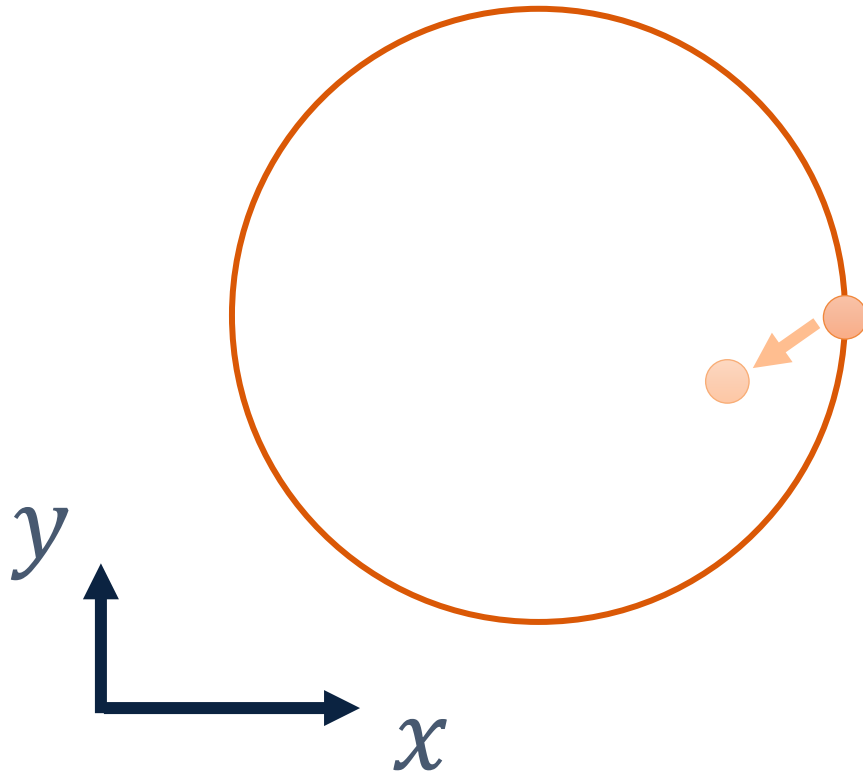
$$\mathbf{w}_u: \quad |\rho_u| > 1$$

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$$\bar{x}_u = \bar{x}^* \pm d\hat{\mathbf{w}}_u$$

Propagate forward [0 t]

COMPUTATION OF STABLE/UNSTABLE MANIFOLDS



$$\bar{x}^* = [x, y, z, \dot{x}, \dot{y}, \dot{z}]$$

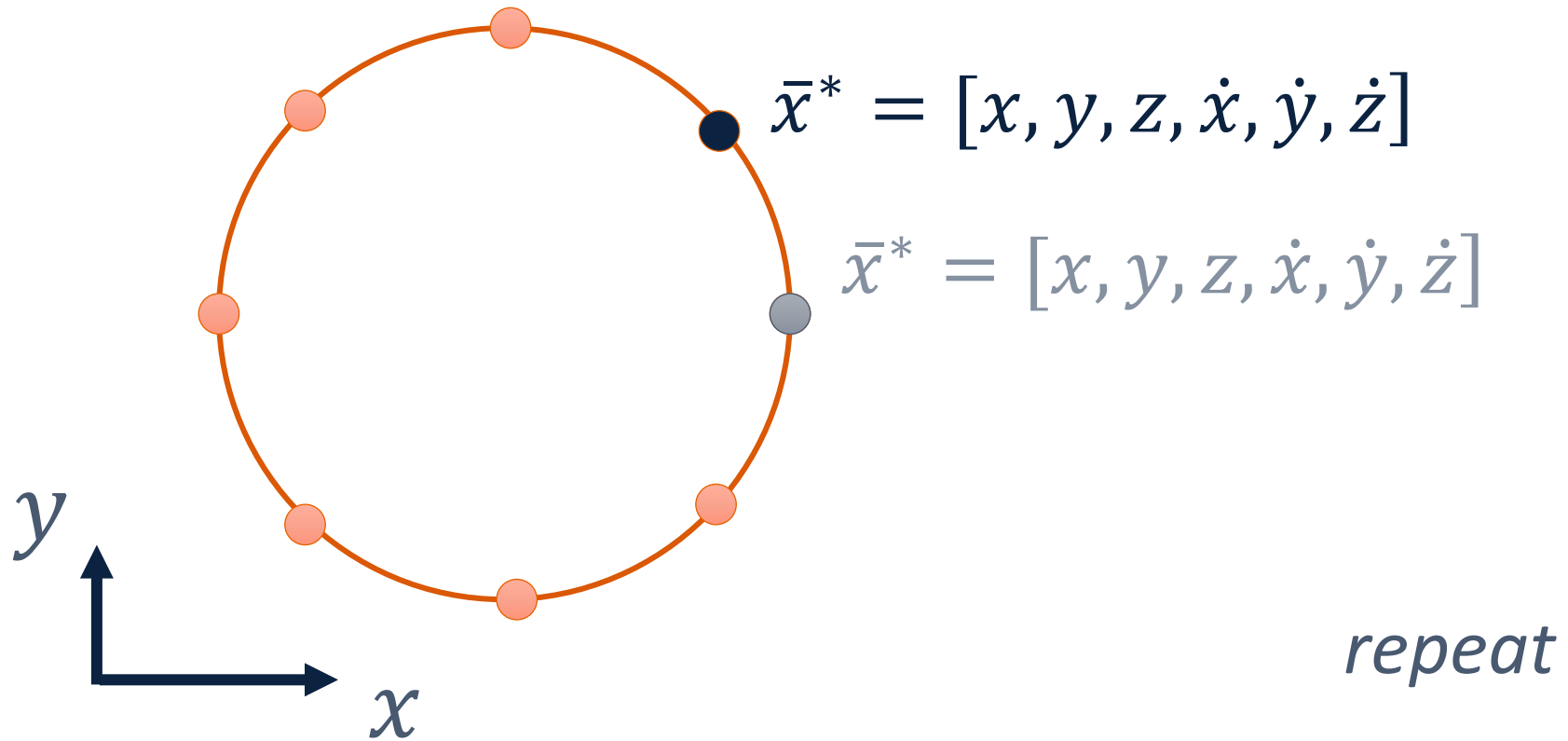
$$w_s: |\rho_s| < 1$$

$$\hat{w}_s = \frac{w_s}{\sqrt{x_s^2 + y_s^2 + z_s^2}}$$

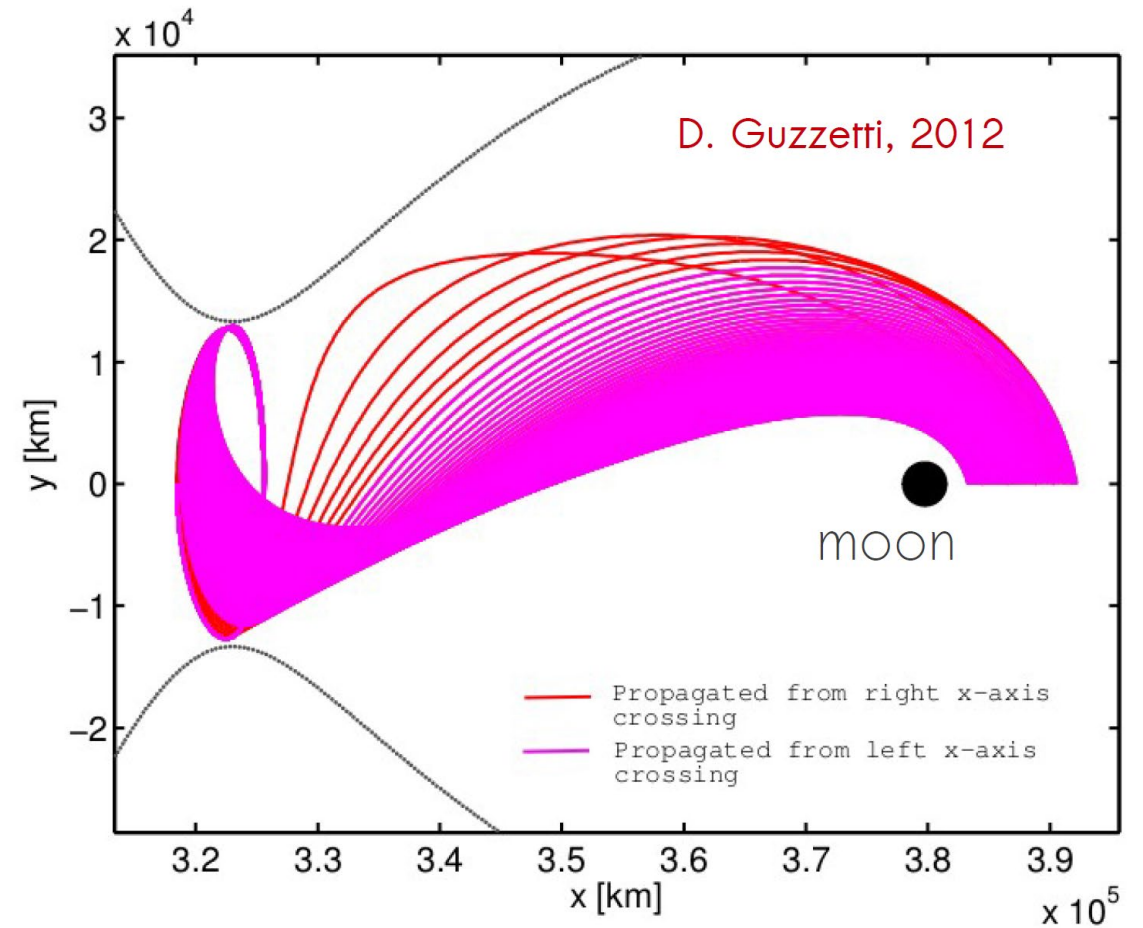
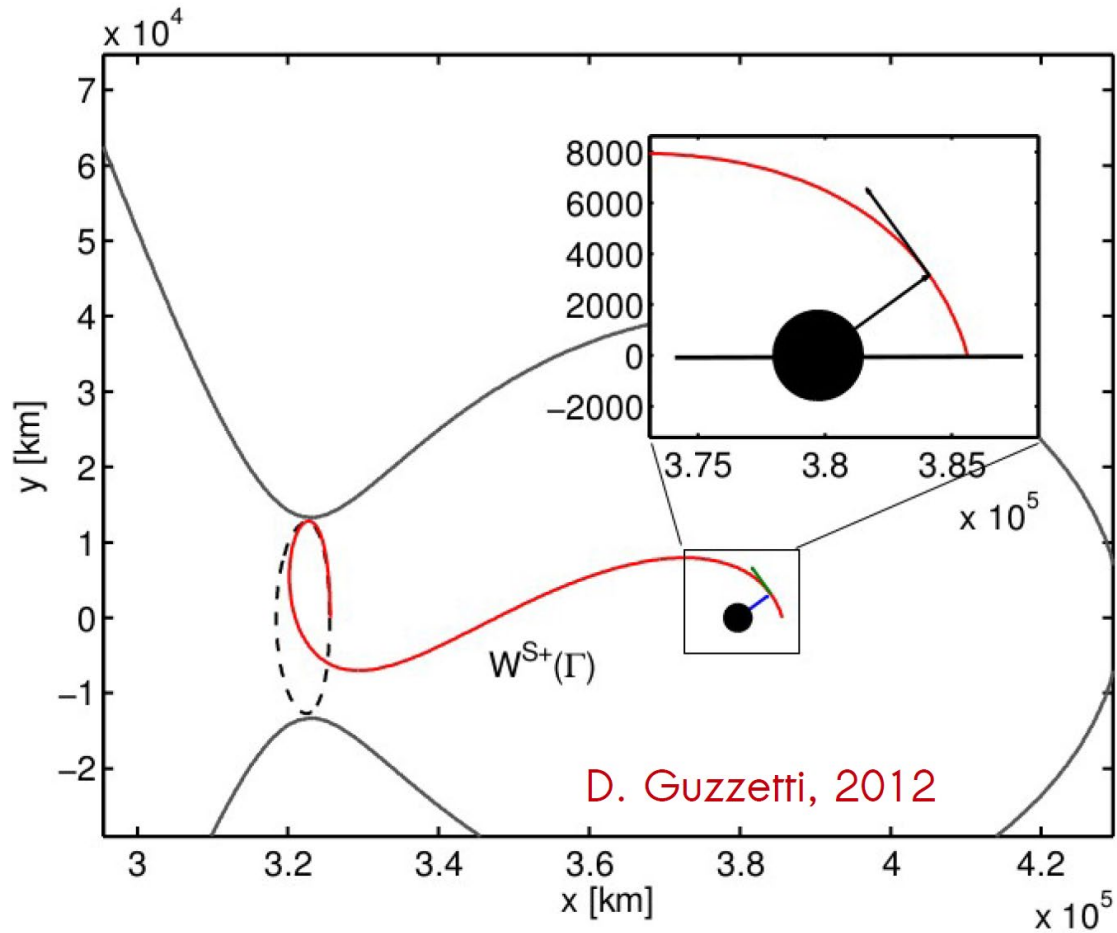
$$\bar{x}_s = \bar{x}^* \pm d\hat{w}_s$$

Propagate backward [0 -t]

COMPUTATION OF STABLE/UNSTABLE MANIFOLDS



GLOBAL INVARIANT ORBIT MANIFOLDS





STABILITY INDEX

For each reciprocal multiplier pair,

define a stability index as the half-sum of the pair

For an elliptic unit-circle pair (exclude +1 and -1)

Sometime appears as computed from the eigenvalue of maximum modulus

This quantity is always at least one and equals one for all unit-circle pairs

$$\rho_i, \quad \rho_i^{-1}$$

$$a_i = \frac{1}{2}(\rho_i + \rho_i^{-1})$$

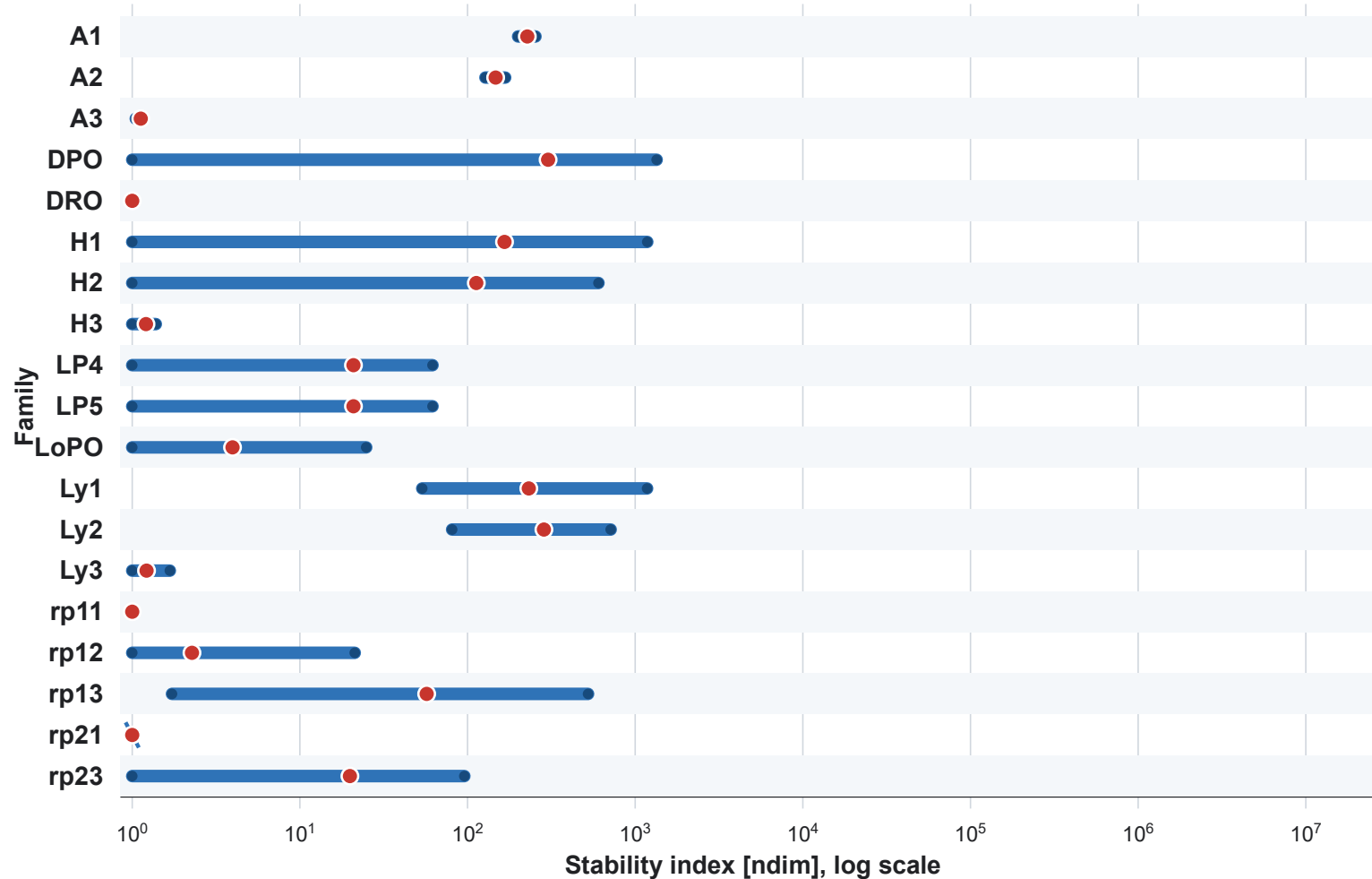
$$\rho_i = e^{i\theta} \quad \rho_i^{-1} = e^{-i\theta}$$

$$a_i = \cos \theta \quad |a_i| < 1$$

$$s_{max} = \frac{1}{2}(|\rho_{max}| + |\rho_{max}^{-1}|)$$

$$s_{max} \geq 1, \quad s_{max} = 1 \text{ if } |\rho_{max}| = 1$$

STABILITY ACROSS ORBIT FAMILIES



$$s_{\max} = \frac{1}{2} (|\lambda_{\max}| + |\lambda_{\max}^{-1}|)$$

Guzzetti, D., Bosanac, N., Folta, D. and Howell, K. "A Framework for Efficient Trajectory Comparisons in the Earth-Moon Design Space," AIAA/AAS Astrodynamics Specialist Conference, San Diego, CA, August 2014.

A READING

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AIAA JOURNAL

Stability of Periodic Orbits in the Elliptic, Restricted Three-Body Problem

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Jet Propulsion Laboratory, Pasadena, Calif.

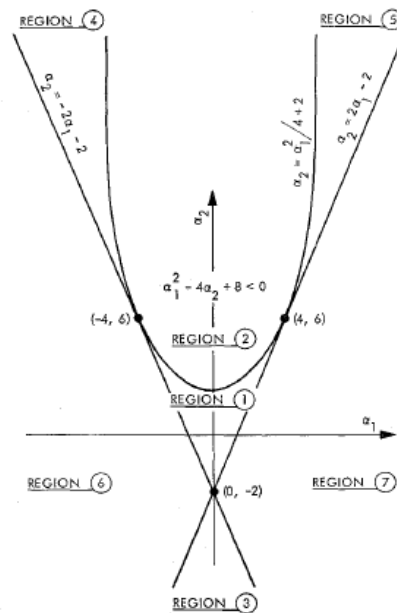


Fig. 1 Seven stability regions.

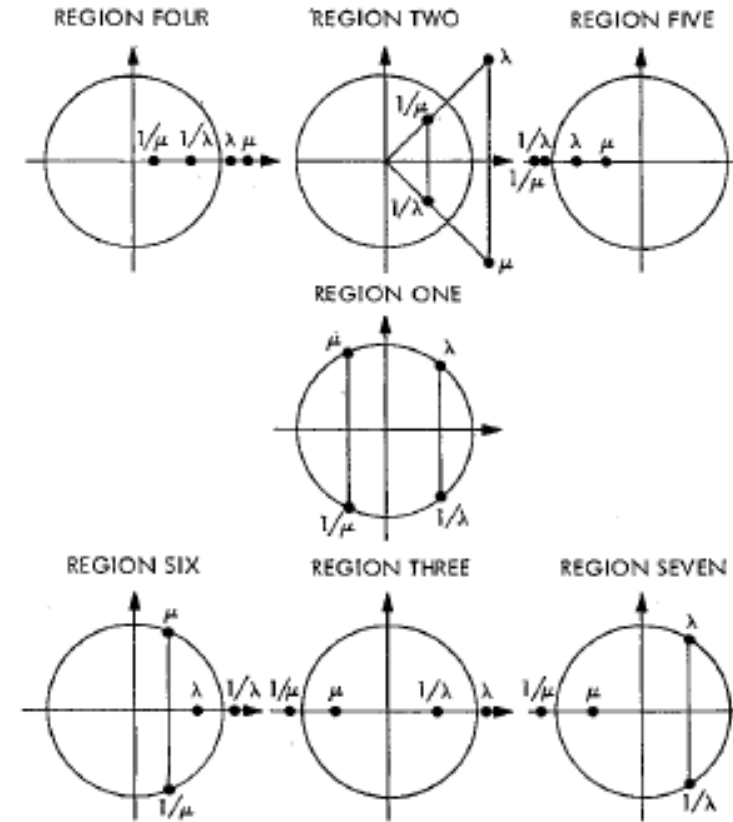


Fig. 2 Configuration of roots for each stability region.