

AePW-4 High-Angle Working Group Meeting



November 14, 2024

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Agenda, November 14



- November 8, DPW8-AePW4 joint meeting
- AIAA SciTech 2025: Orlando, FL, Mini Workshop 1, Thursday, January 9
- AIAA Aviation 2025 in-person meeting
- Summer 2025: New BSCW Experiment
- AePW-4 website: <https://nescacademy.nasa.gov/workshops/AePW4/public>
- Presentation today:
 - FUN3D flutter analysis status
- Next meeting, December 12
- AIAA Aviation 2026: DPW-8 and AePW-4 Workshop



High Angle Working Group July 2022

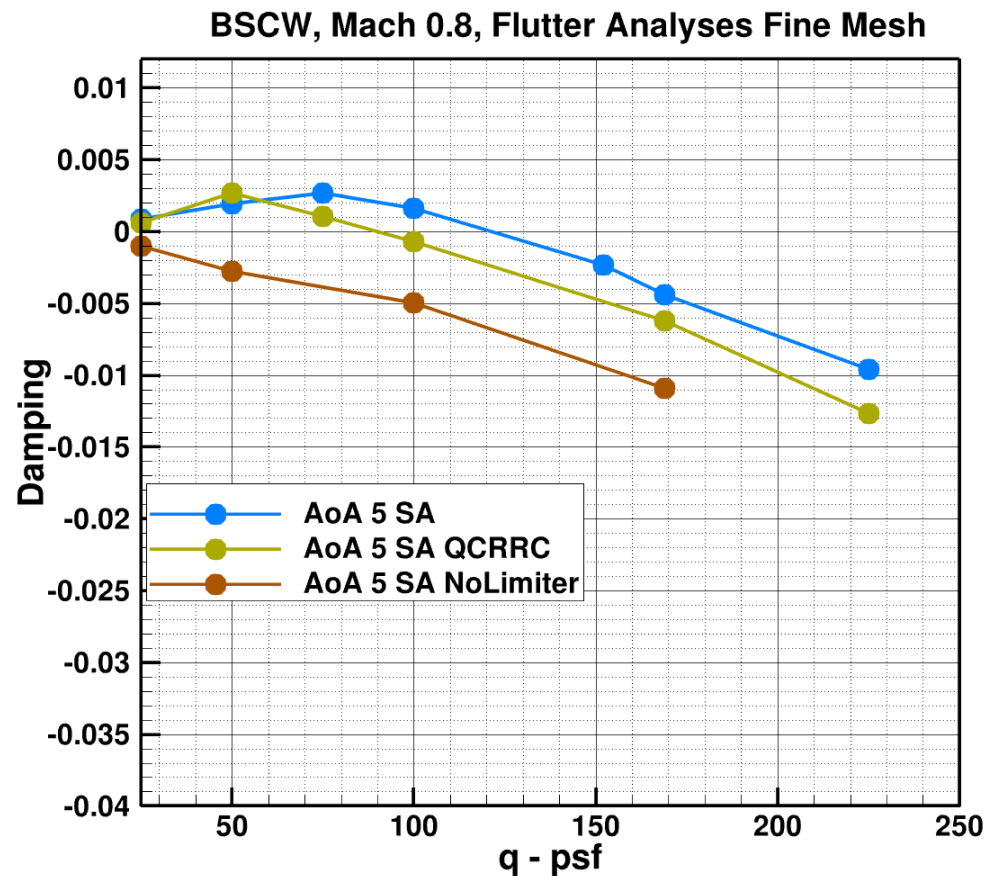
FUN3D BSCW flutter analysis at Mach 0.8, AoA = 5 deg

Turbulence model: SA

Flux Limiter: No

- Very good convergence
- Shock is pushed toward leading edge
- Flutter solutions are always unstable (negative damping) - SA and DDES

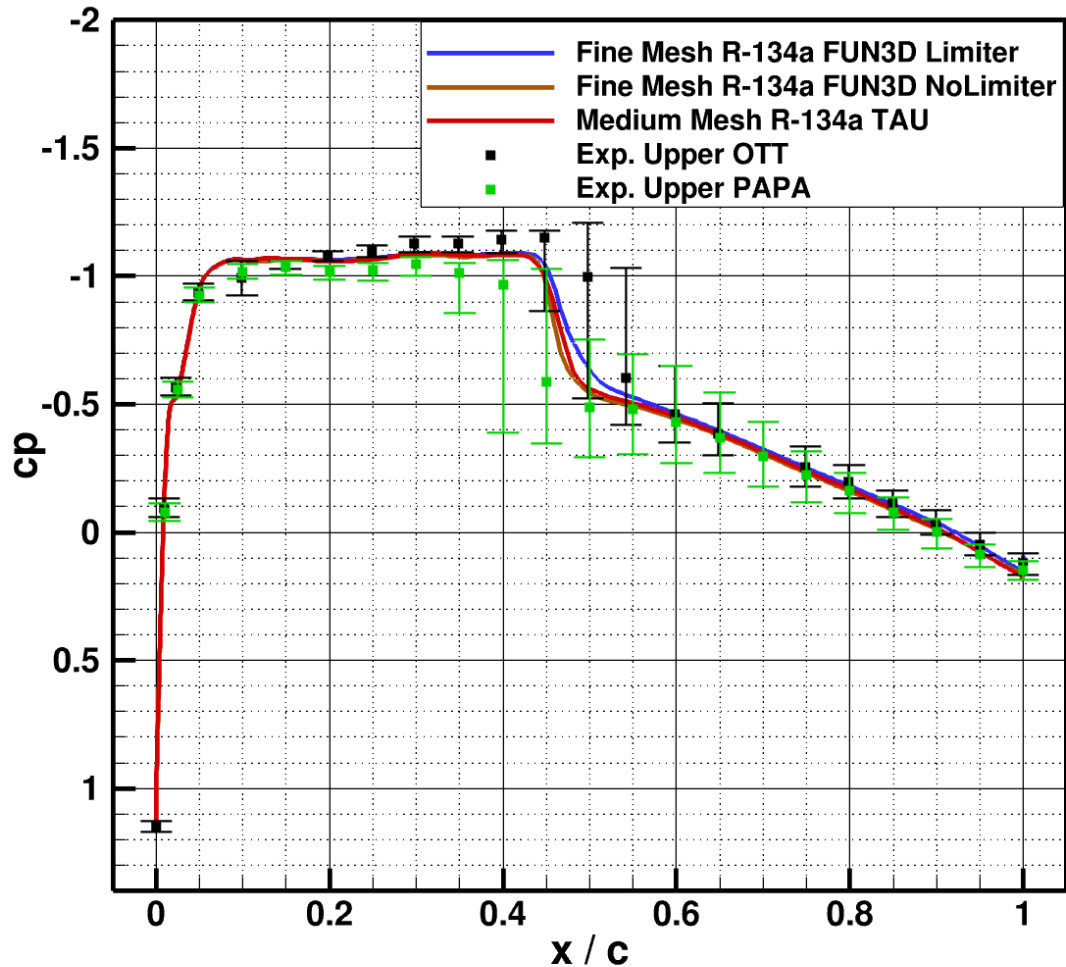
- SA w/ compressibility correction?



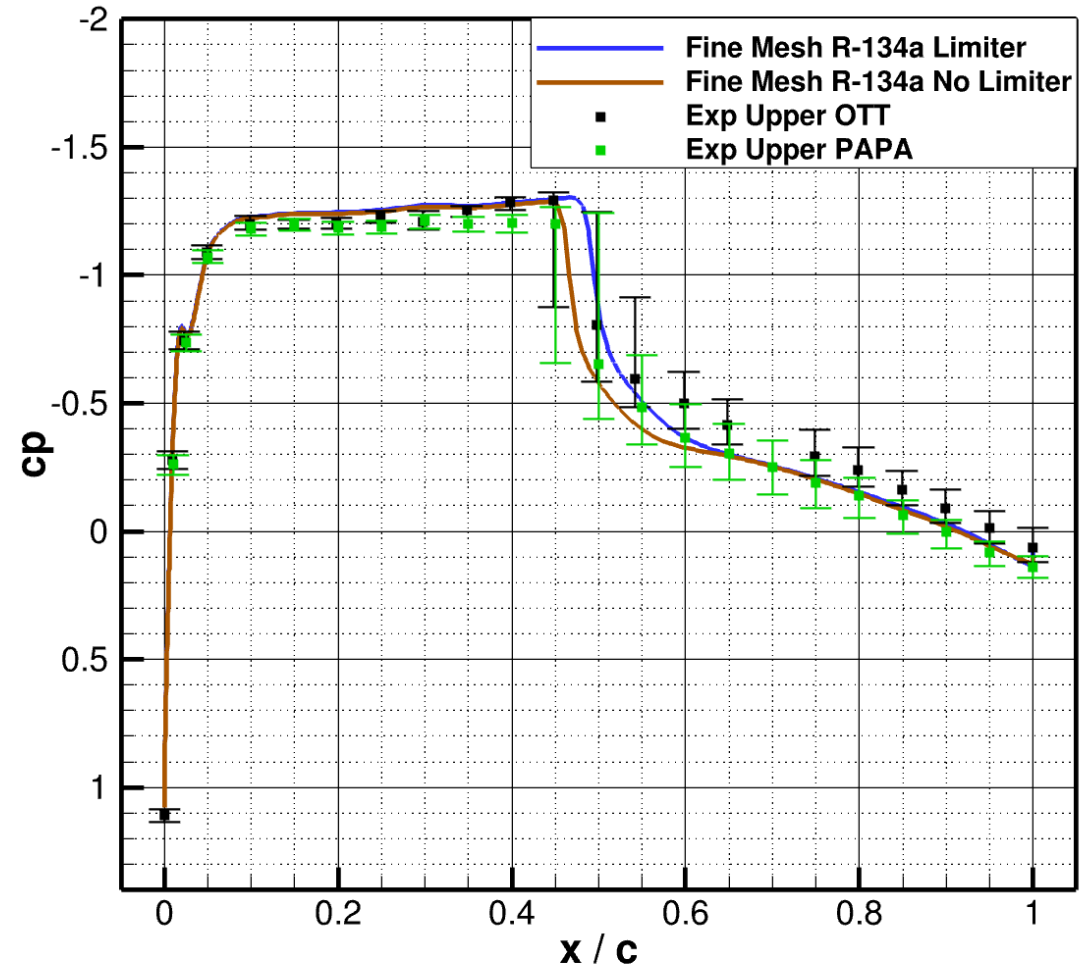
Rigid Steady FUN3D SA with and without Venkat Limiter



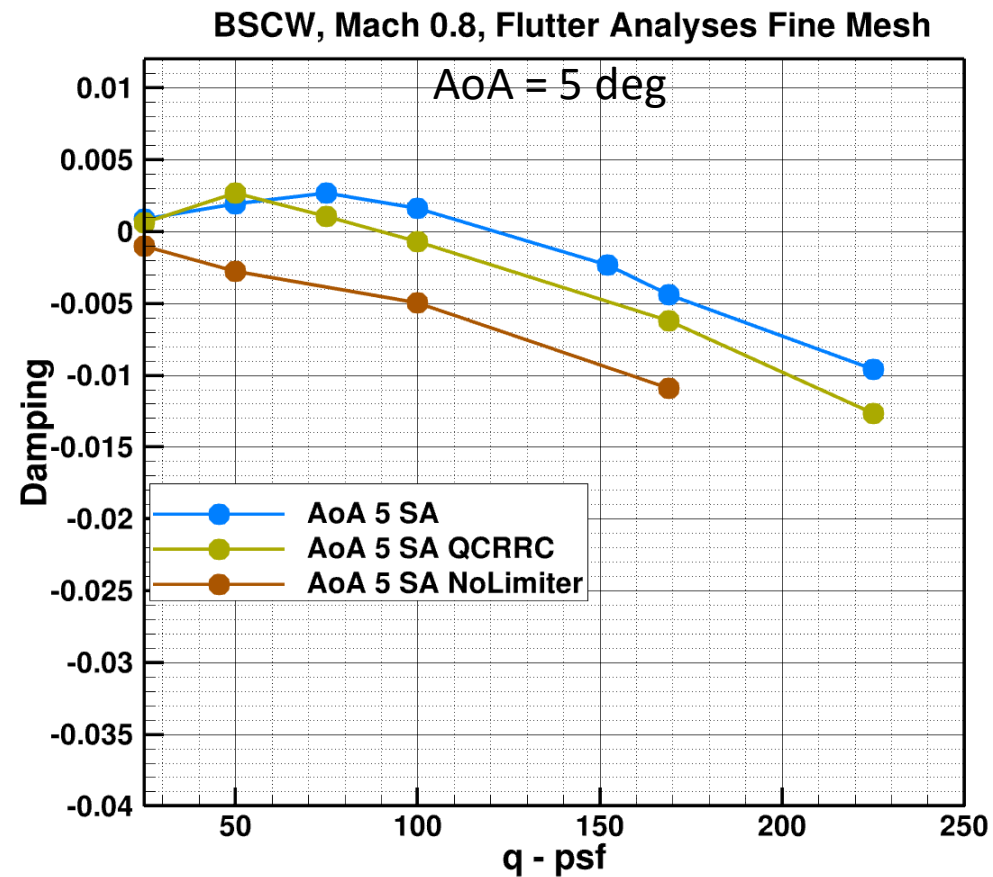
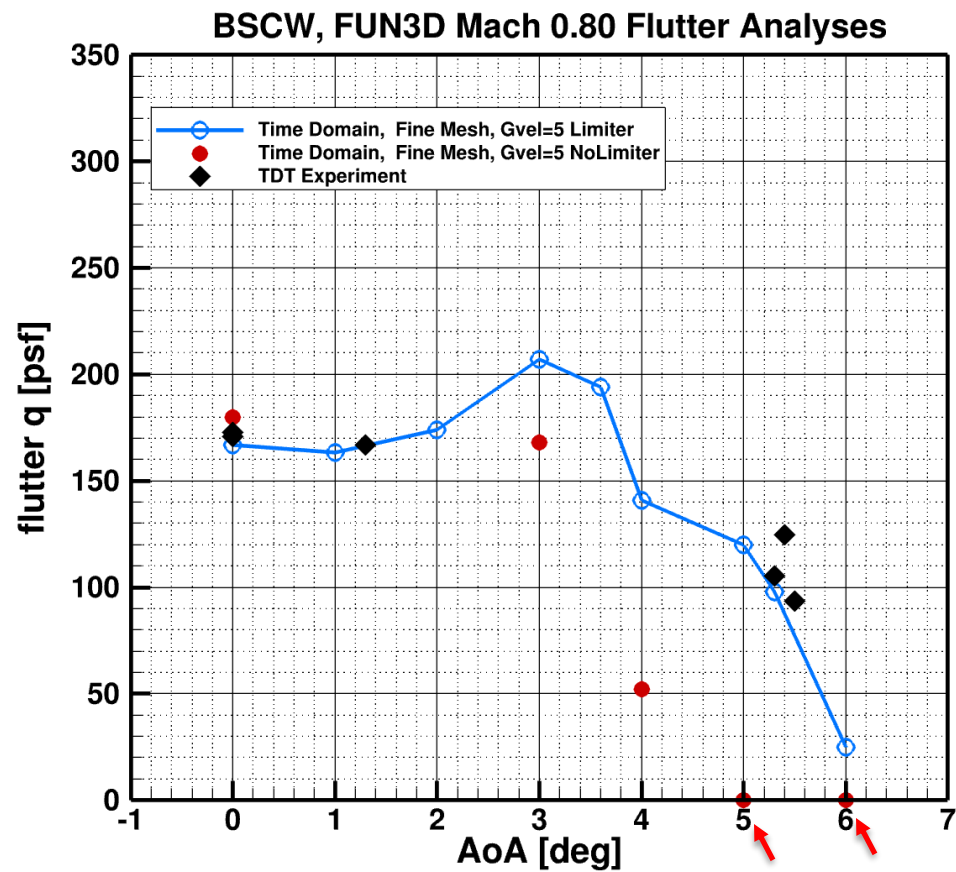
BSCW Mach = 0.8, AoA = 3 deg, Q = 169 psf
60% Span Station



BSCW Mach = 0.8, AoA = 5 deg, Q = 169 psf
60% Span Station



Flutter, FUN3D SA with and without Venkat Limiter

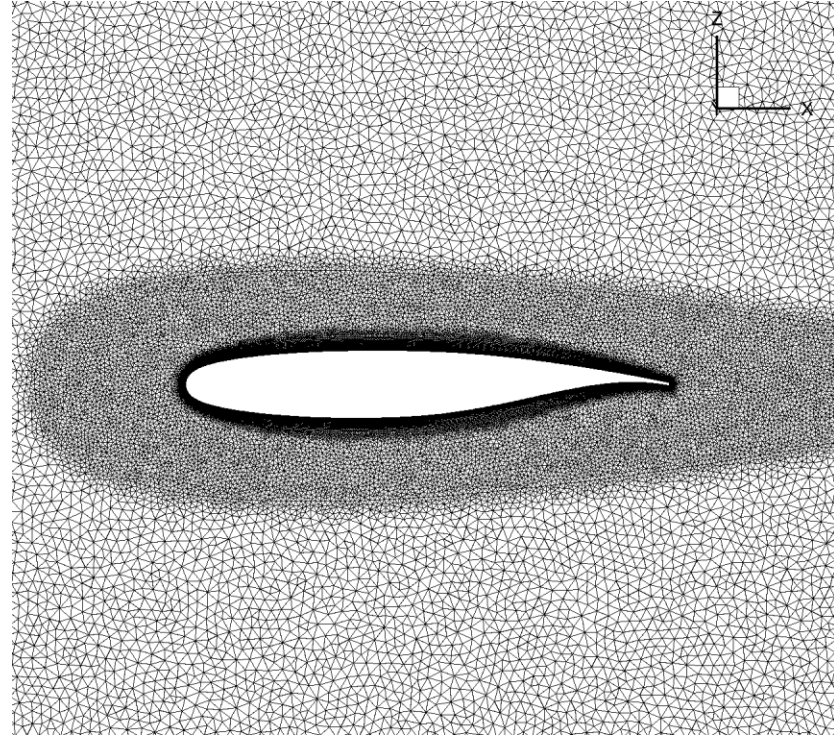


This only implies that flutter q cannot be computed

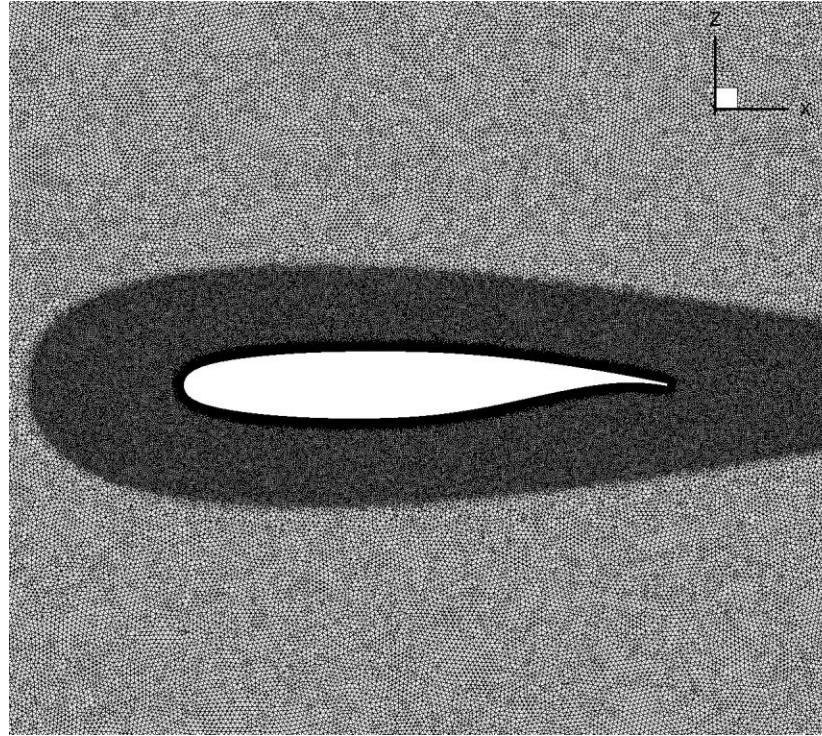
2D Analysis



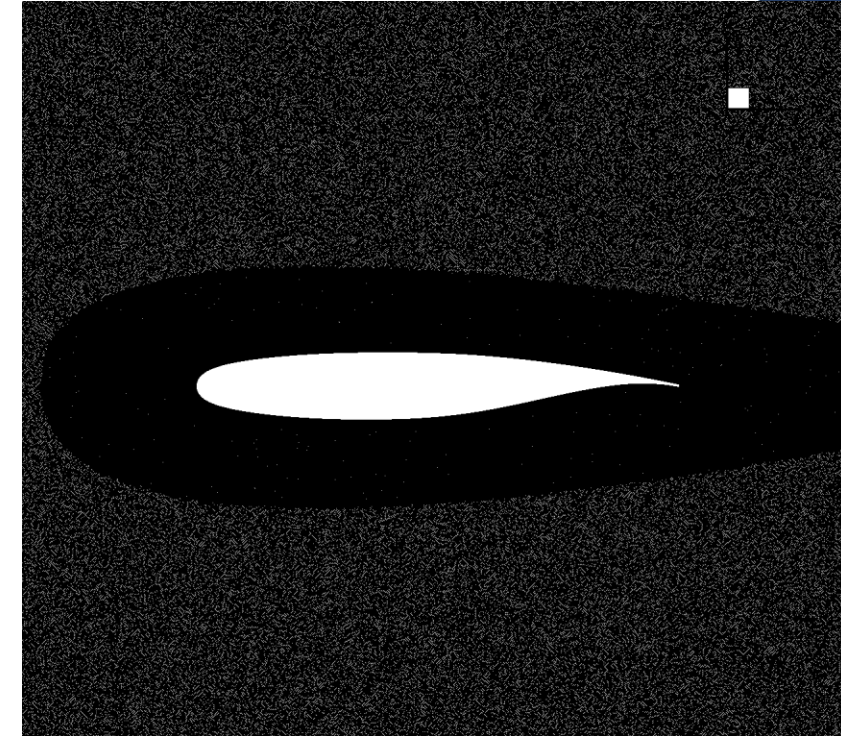
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1794326

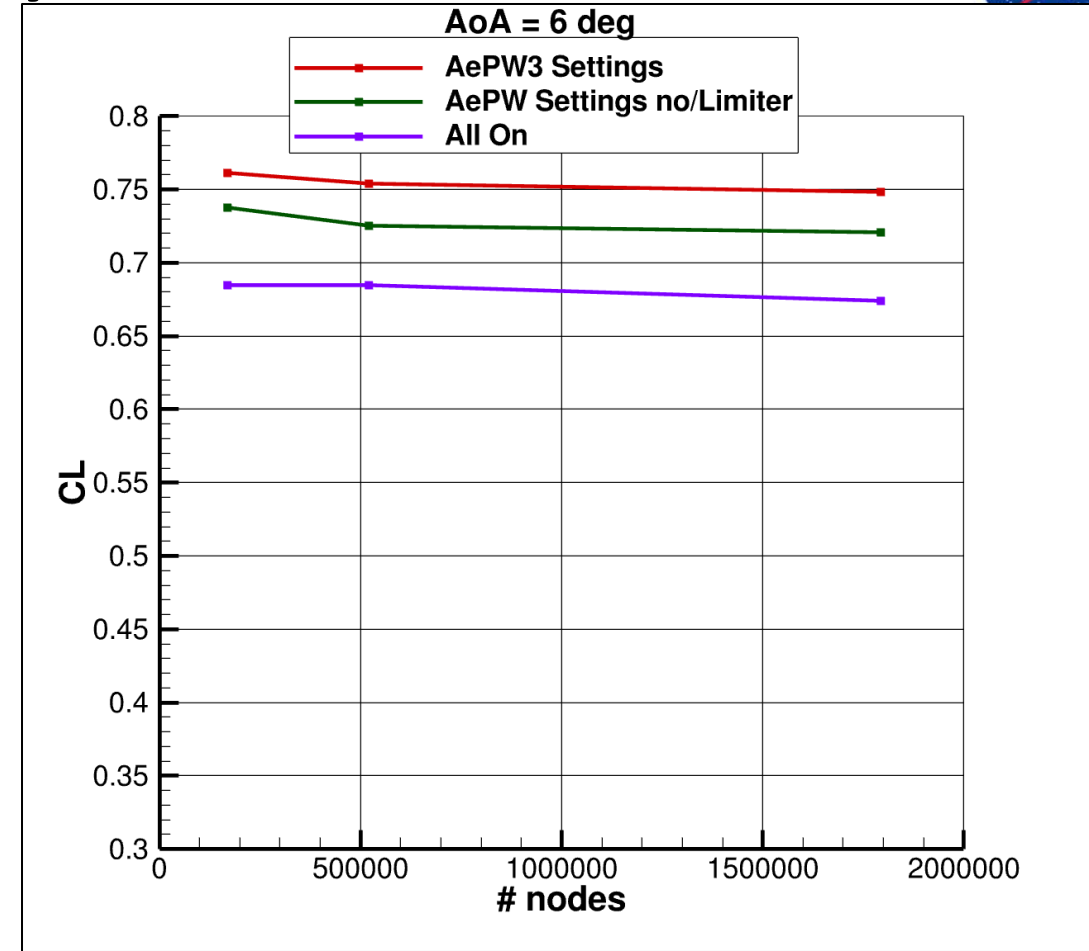
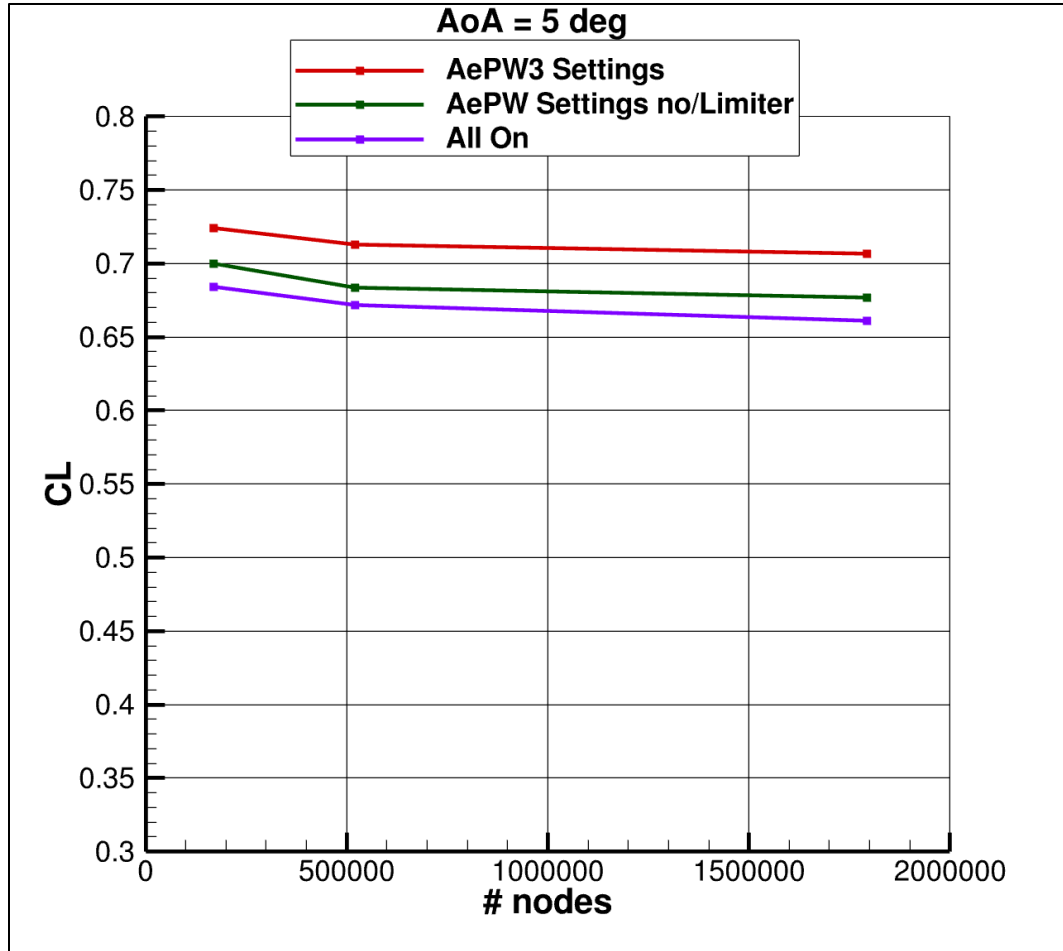


AePW-3 Settings: SA, w/Limiter

AePW-3 Settings: SA, no/Limiter

All Settings: SA-neg, QCR, RC, Compressibility Correction, 2nd order Turbulence, Frozen Limiter, Hanim

2D Analysis

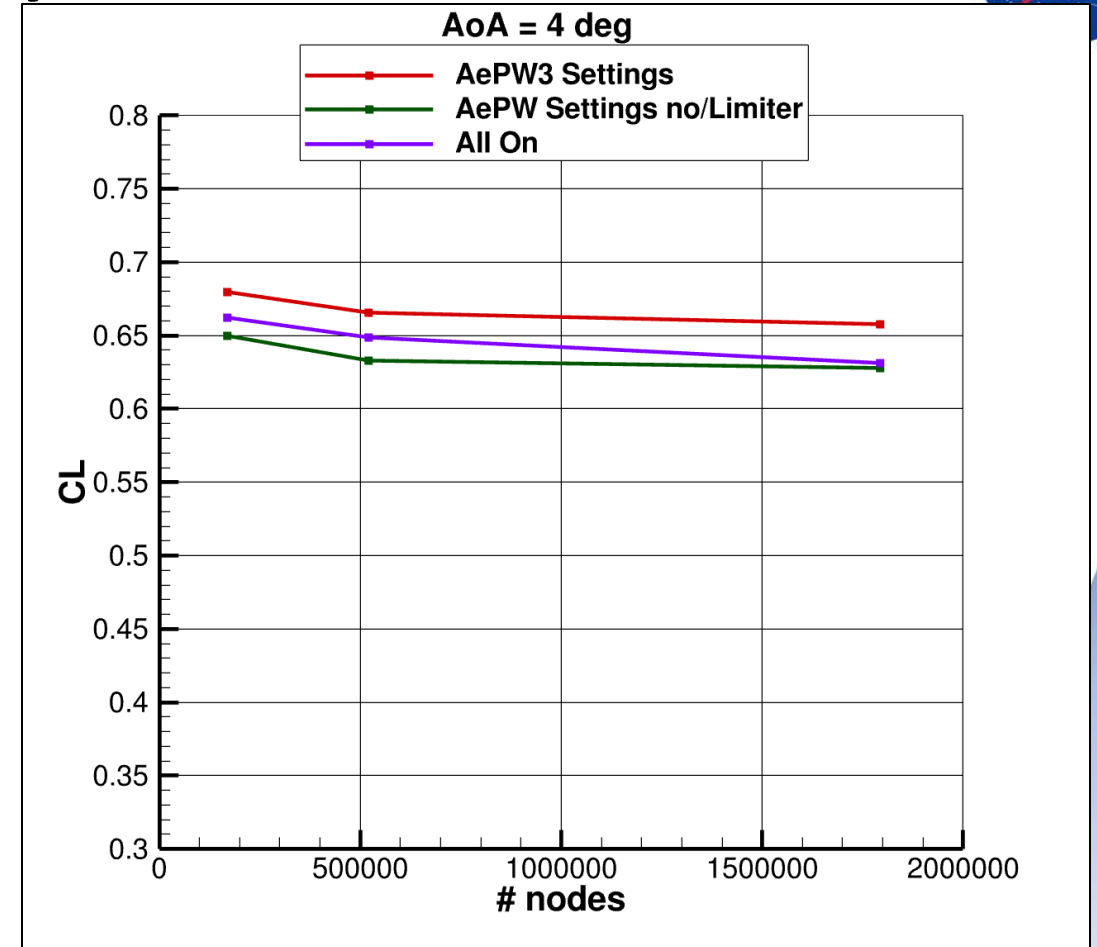
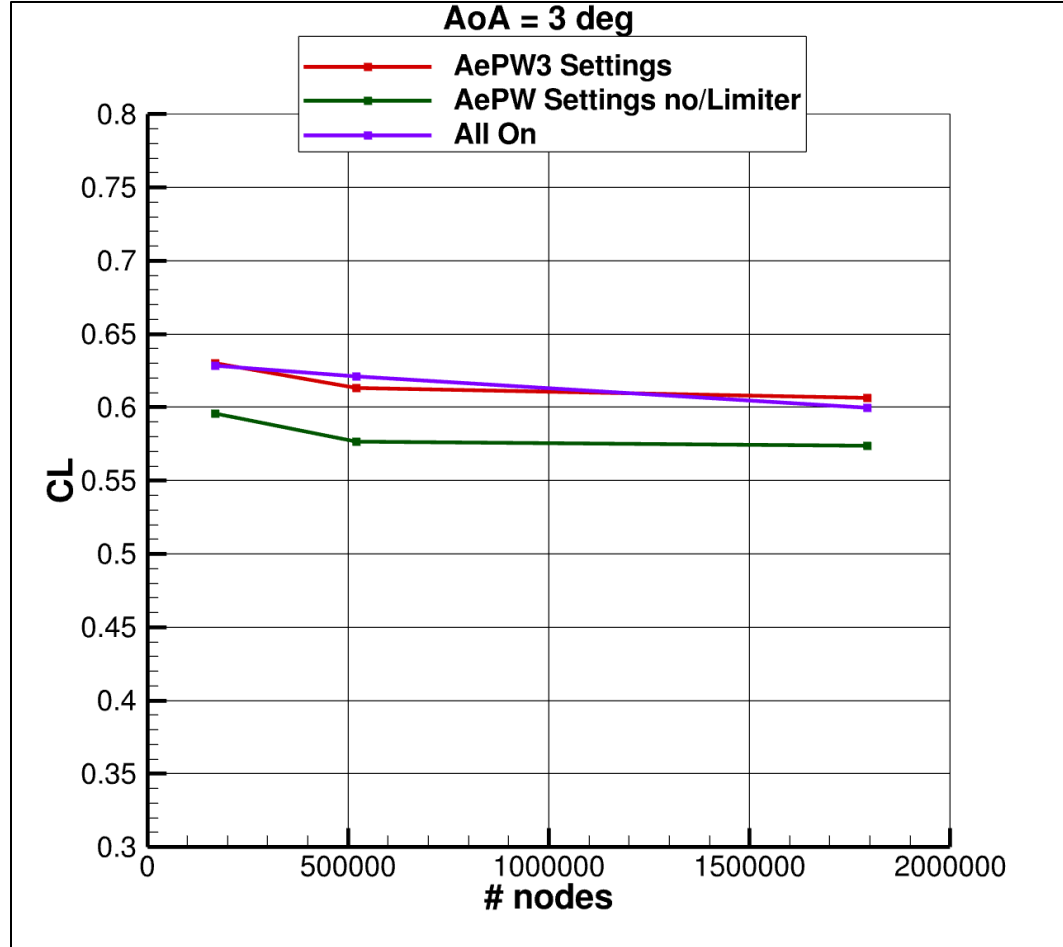


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2D Analysis



AePW-3 Settings: SA, w/Limiter

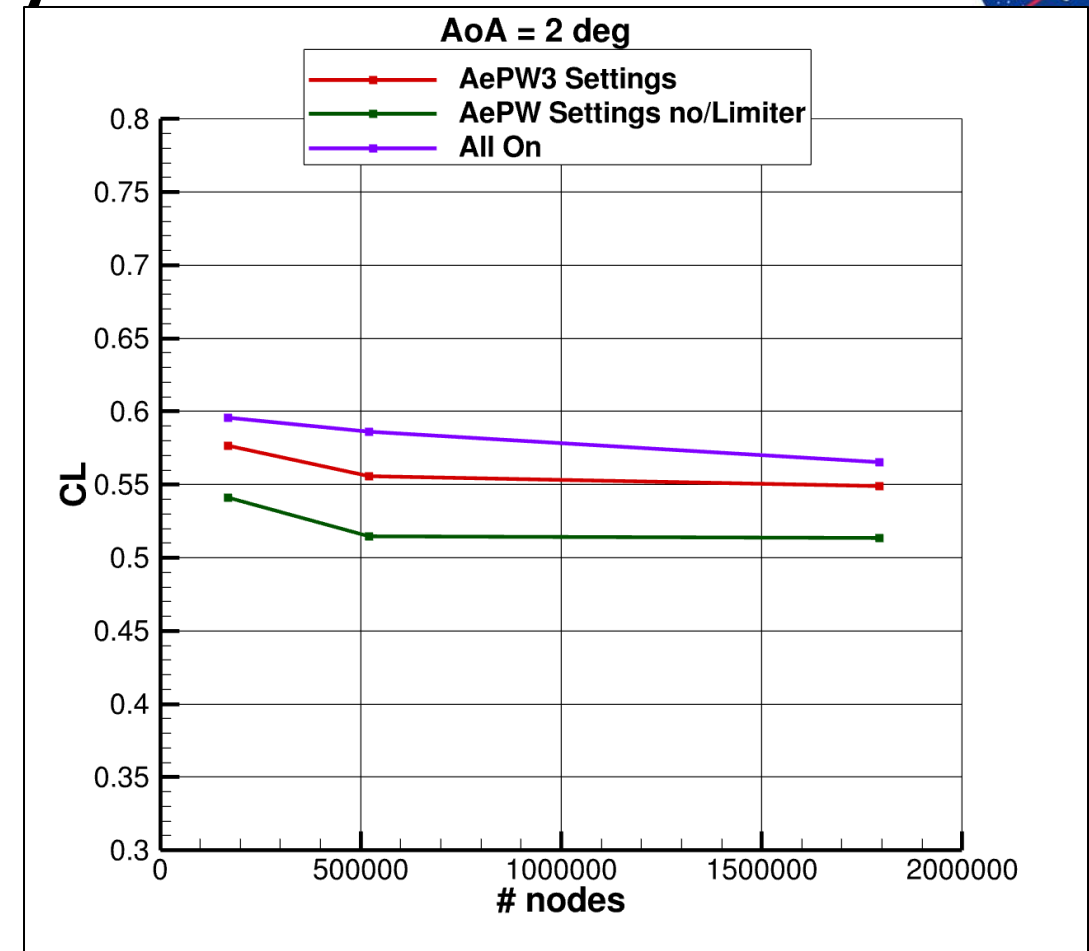
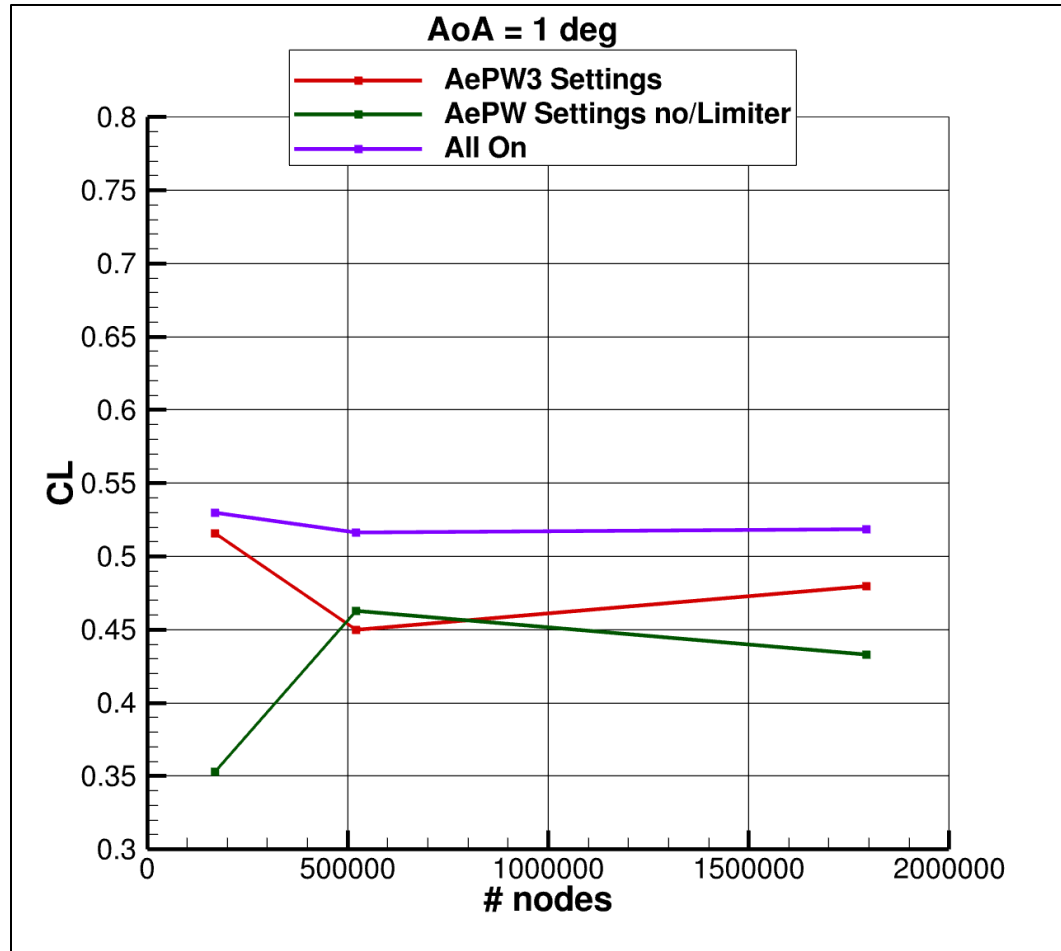
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Poor Convergence, AePW-3 Settings !

2D Analysis

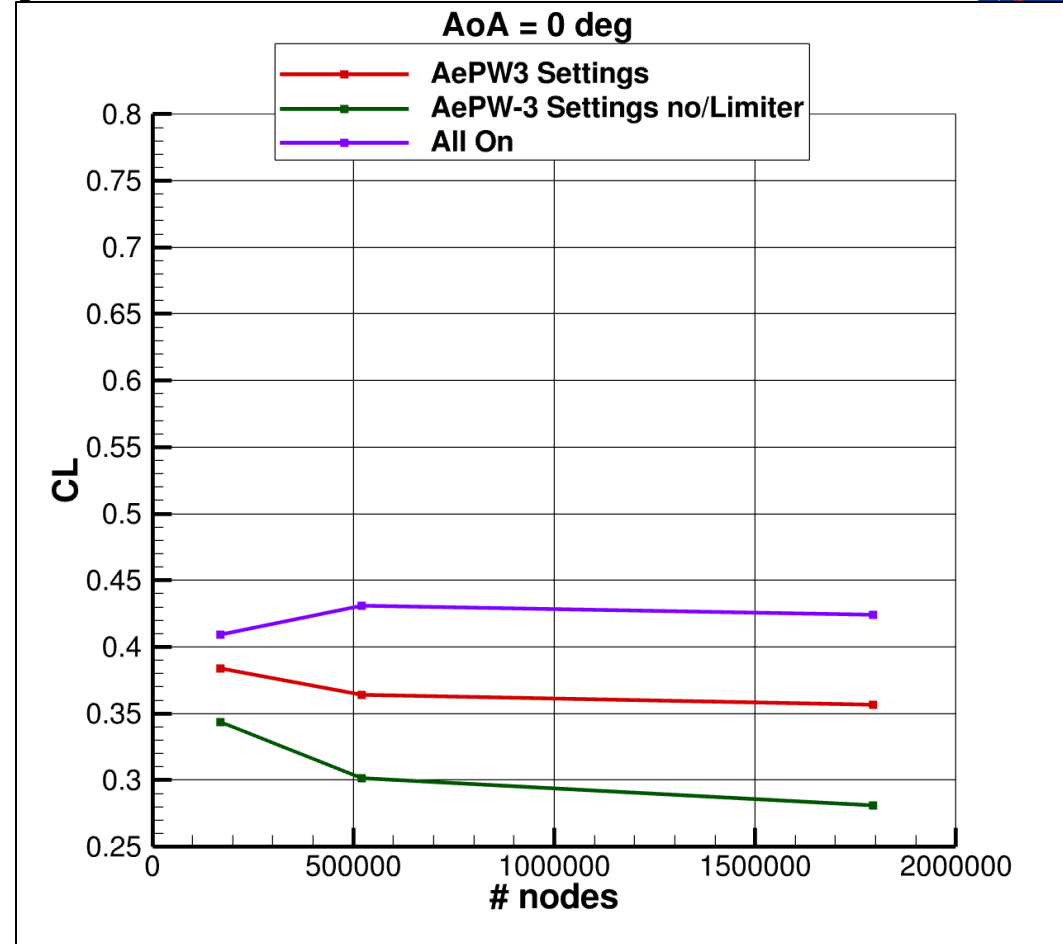


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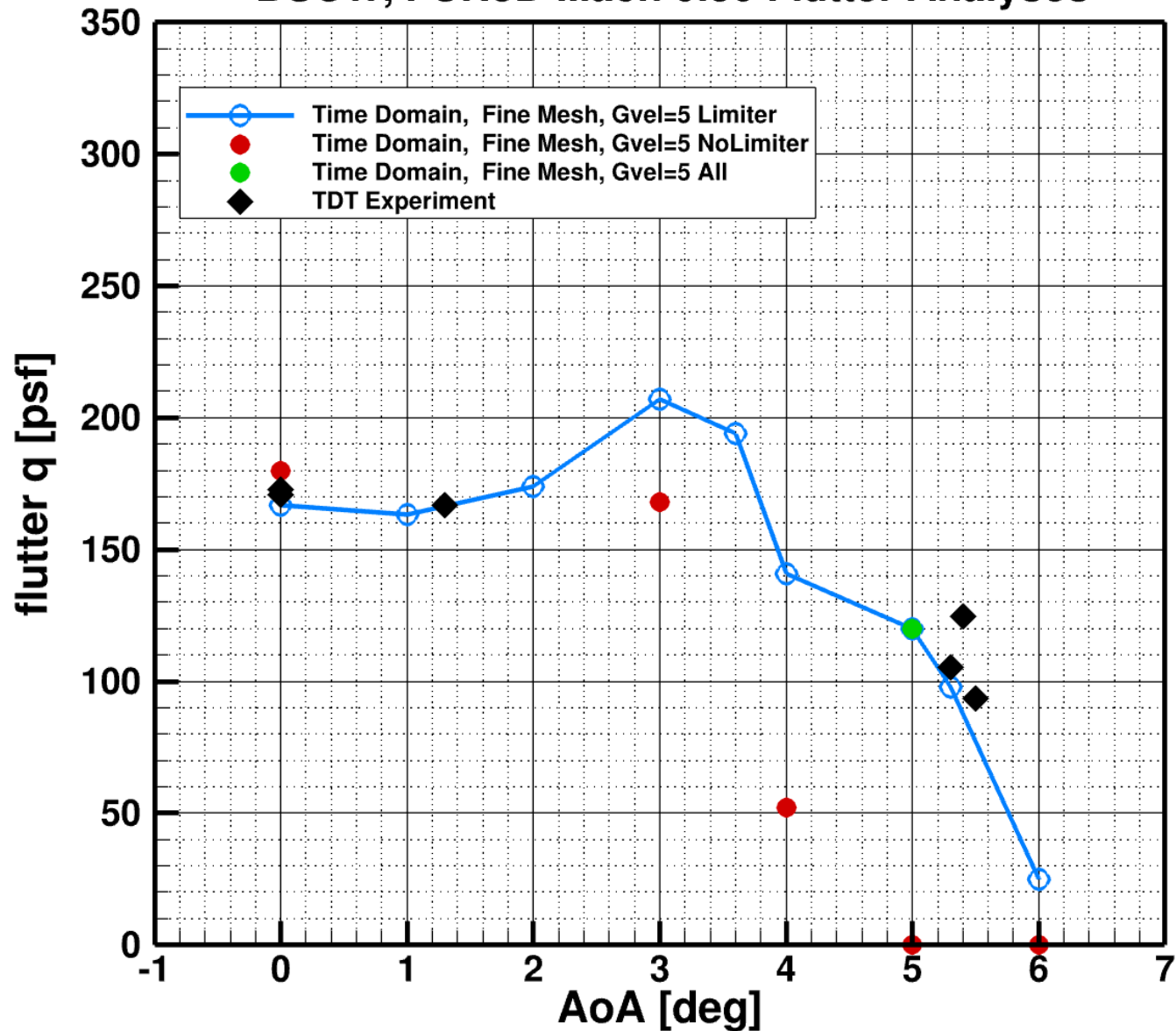


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BSCW, FUN3D Mach 0.80 Flutter Analyses



- Similarly to ONERA OAT15A, do we need in this working group a 2D BSCW rigid steady analysis across [0, 1, 2, 3, 4, 5, 6] (deg) and three meshes to establish a scatter in data among workshop participants ?
- Considering scatter in rigid steady results do we need to design a 2D BSCW flutter case?

