The background of the slide is a composite image. The upper portion shows a large, pale, icy celestial body, likely Europa, with visible surface features like ridges and grooves. The lower portion shows a dark, rocky, and cratered surface, possibly the ground of a moon or planet. The sky is a deep black with scattered white stars.

# **ELSA**

## **Europa Lidar Sensor Assembly**

**Luke Skelly**





# Outline

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- **Europa lander mission**
  - **Goals**
  - **Challenges**
  - **Concept of Operations**
- **Lidar system design**
  - **Challenges**
  - **Key technologies and approaches**
- **Brassboard prototype design**
- **ELSA Elsewhere**
- **Summary**





# Europa Lander Mission Goals

## 1. Life

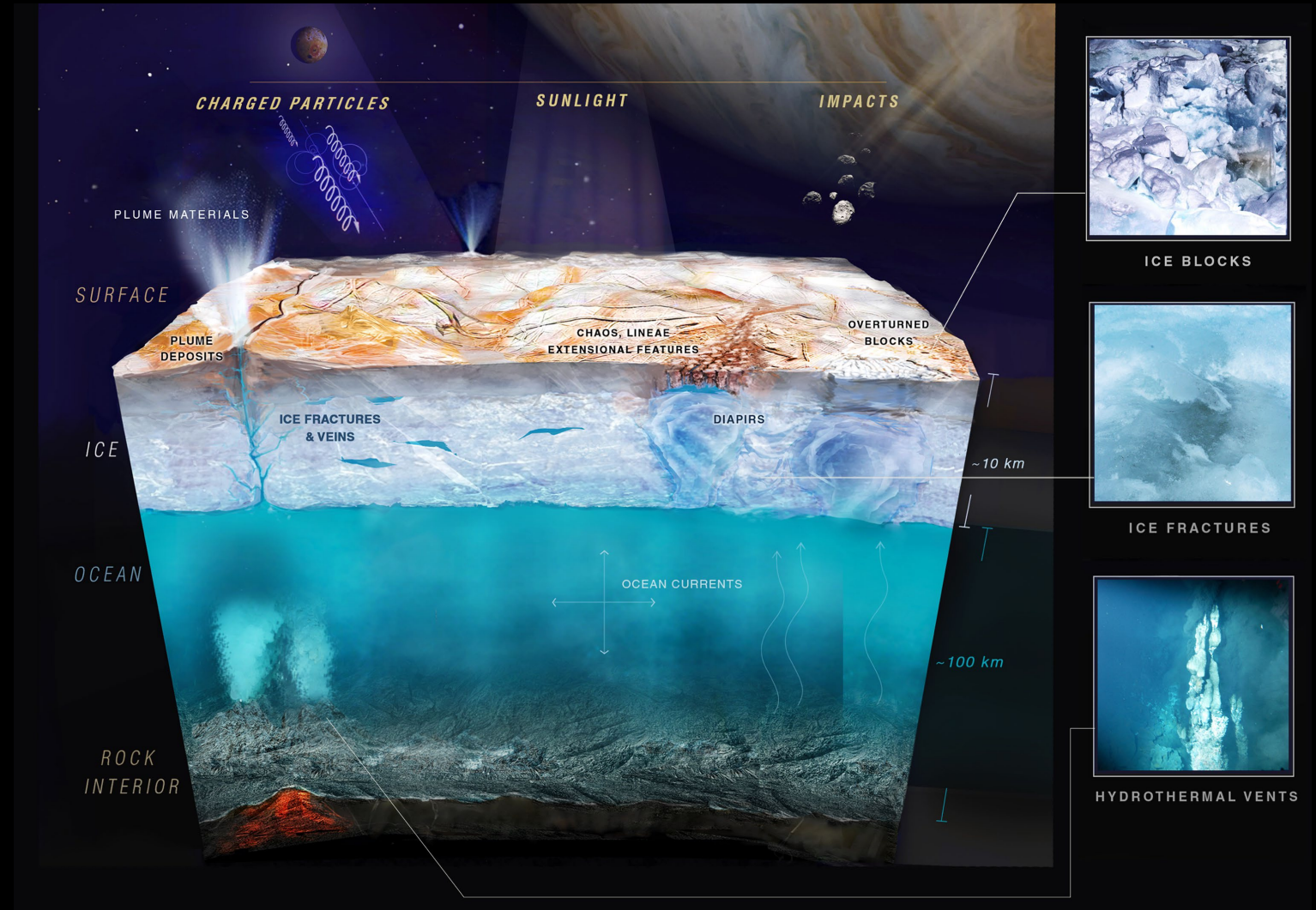
- Search for evidence of life

## 2. Habitability

- Assess the habitability of Europa via in situ techniques

## 3. Context

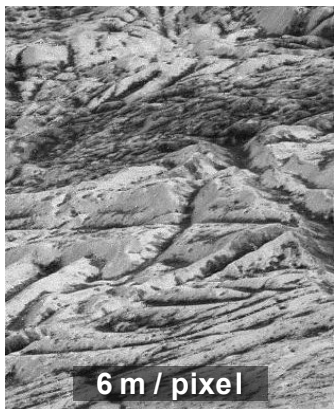
- Reconnaissance for future robotic exploration





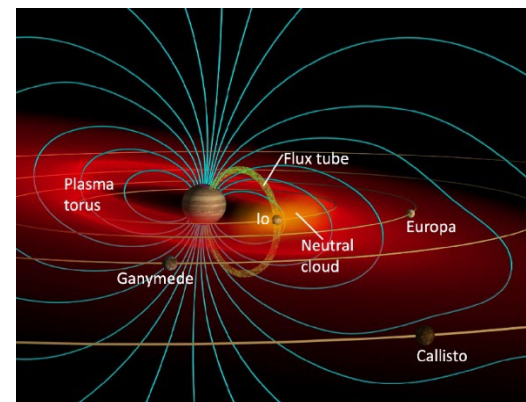
# Challenges to Landing on Europa

## Unknown Terrain



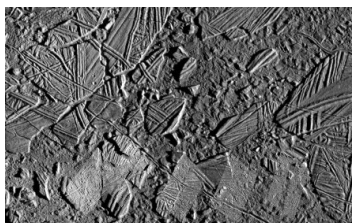
- Current highest resolution image is 6 meters per pixel
- Clipper mission could provide 50 cm imagery over 2 km swaths
- Not enough resolution for lander hazard avoidance

## Ultra-high Radiation



- Ionizing radiation of electrons and protons generated by Jupiter's magnetosphere
- 1.5 Mrads with 25 mm Al shielding
- 150 krads in "vault"

## Hazardous Terrain



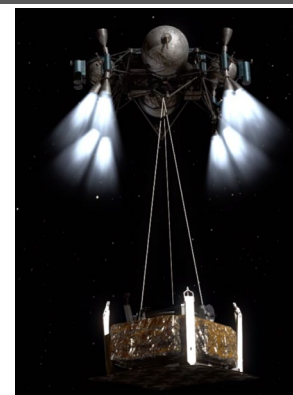
Conamara Chaos,  
Europa



Atacama Desert,  
Chile

- Penitents (blades of ice) could be up to 50 meters high

## Limited SWaP



- 40+ kg of science equipment
- Sky crane without atmosphere to parachute with
- Limited battery power

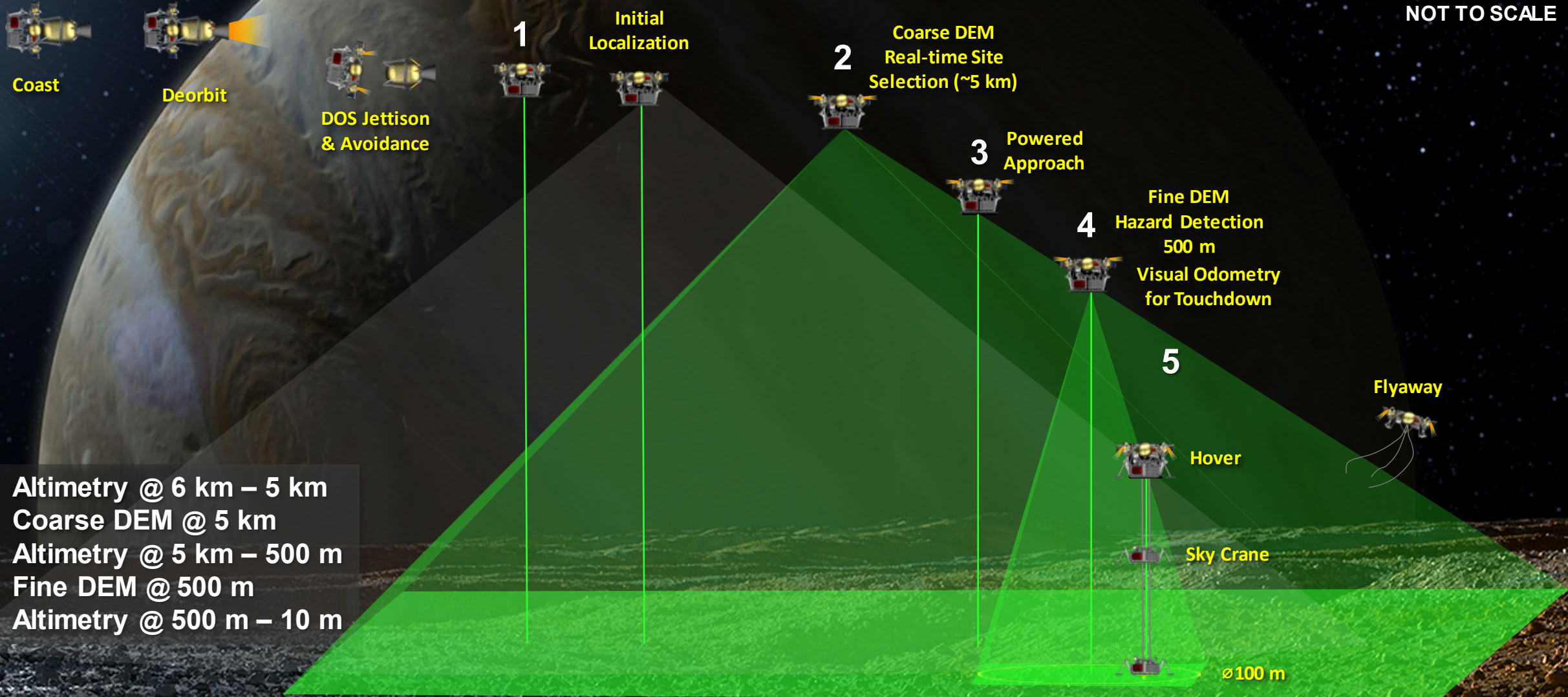
**Lidar is the best technology for rapid high-resolution imaging in ultra-high radiation environment**





# Europa Lander OV-1

NOT TO SCALE

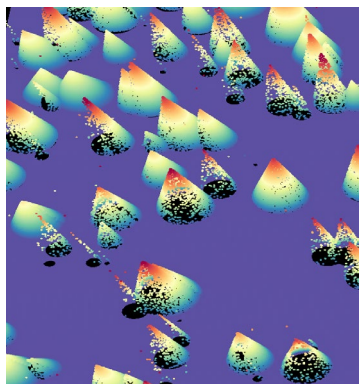


1. Altimetry @ 6 km – 5 km
2. Coarse DEM @ 5 km
3. Altimetry @ 5 km – 500 m
4. Fine DEM @ 500 m
5. Altimetry @ 500 m – 10 m



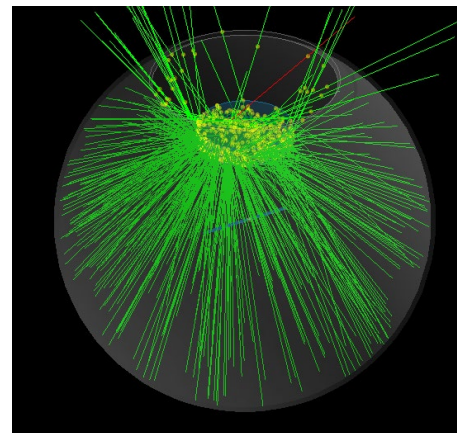
# Challenges to ELSA Design

## Performance Requirements



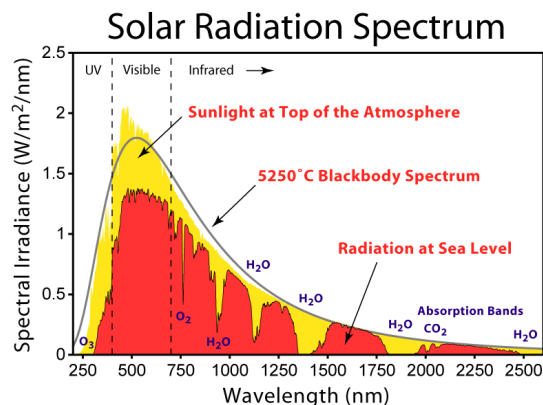
- **Dynamic range:**
  - Altimetry: 10 m to 10,000 m
  - Mapping: 500 m to 5,000 m
- **Fine mapping:**
  - Precision: 5 cm xyz
  - Accuracy: 100 cm
- $\pm 6$  degree field-of-view

## Ultra-high Radiation



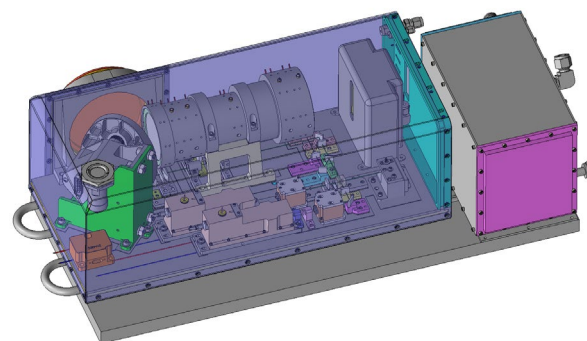
- **Optical system must have line-of-sight to surface**
  - Optical head must be outside vault
- **Radiation damage concerns to laser, optics, and detector**
- **Potential radioluminescence and Cherenkov radiation**

## Solar Background Noise



- **Radiation hardness pushes us to wavelengths of peak levels**
- **Competitive with signal link**
- **Narrowband filter design limited by wide field-of-view**

## Limited SWaP



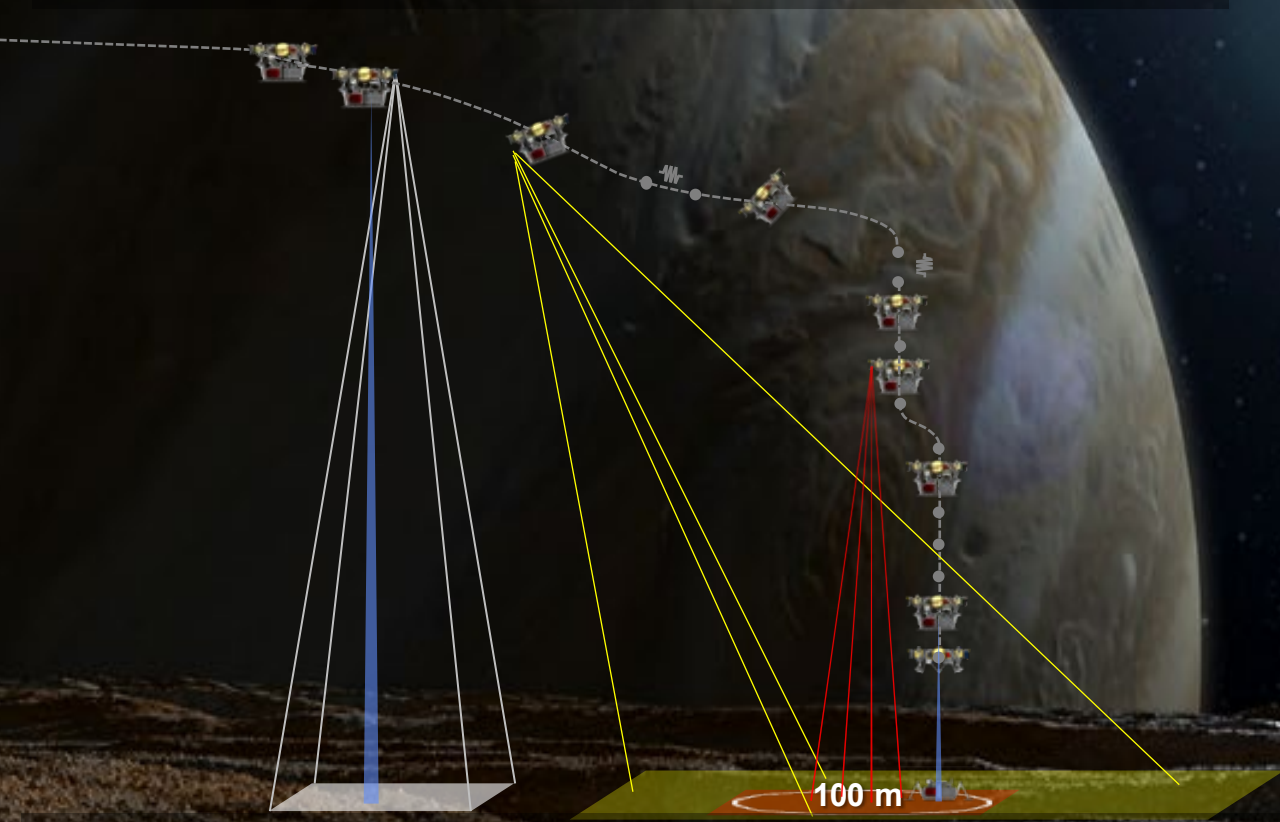
- **Weight: 7 kg**
- **Size: 15,625 cm³ = (25 cm)³**
- **Compact optics**
- **Limited power:**
  - Laser
  - Data processing





# ELSA Overview

## Real-time 3D Mapping & Hazard Avoidance

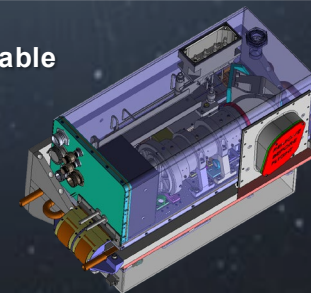


- Goal: Land 1-m<sup>2</sup> lander on unknown rough terrain in extreme radiation environment
- Multi-function ladar: Altimetry, course & fine 3D mapping
- *Real-time mapping 100 m x 100 m at 5-cm resolution in 1 second and image processing in 1 second*
- Current phase of program is to develop, build, and test a brassboard prototype

## Key Technologies & Approaches

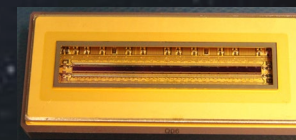
### Low-SWaP Ladar System

Flight-traceable design



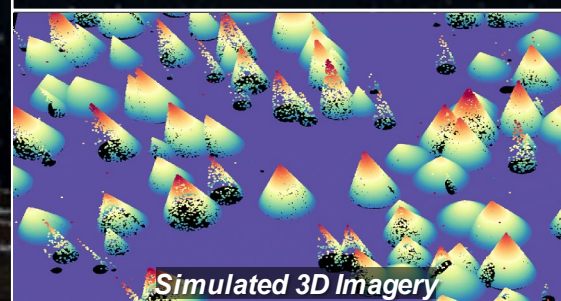
### Novel FPA Technology

32 x 2048 Si APDs



Rad-hard

### Embedded Real-time Processing



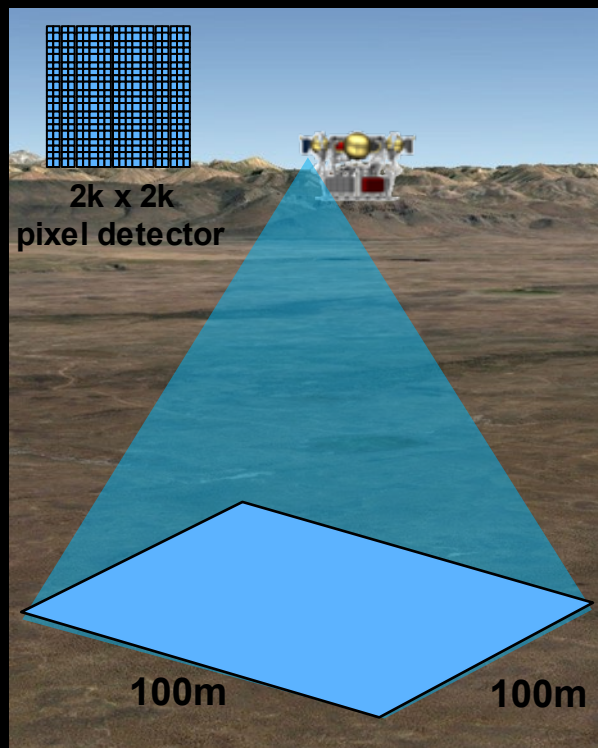
### Simplified Scan Architecture





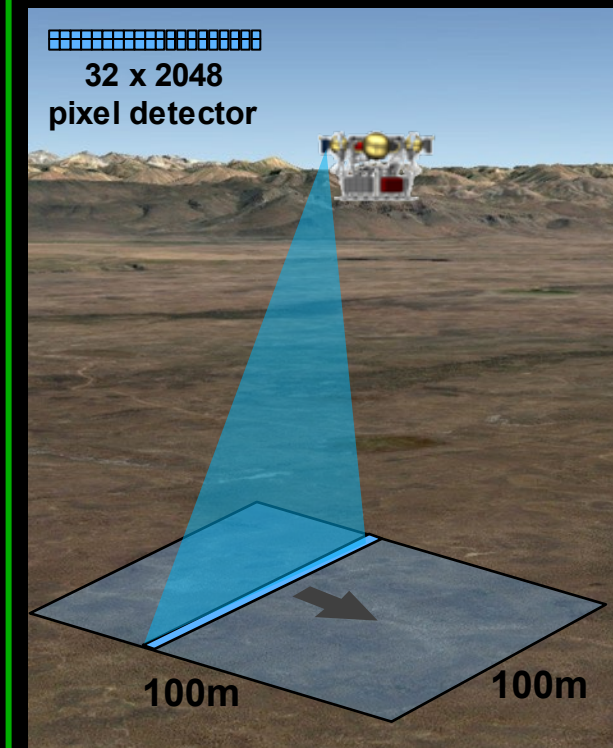
# Lidar Scan and Array Format

## Flood Illumination



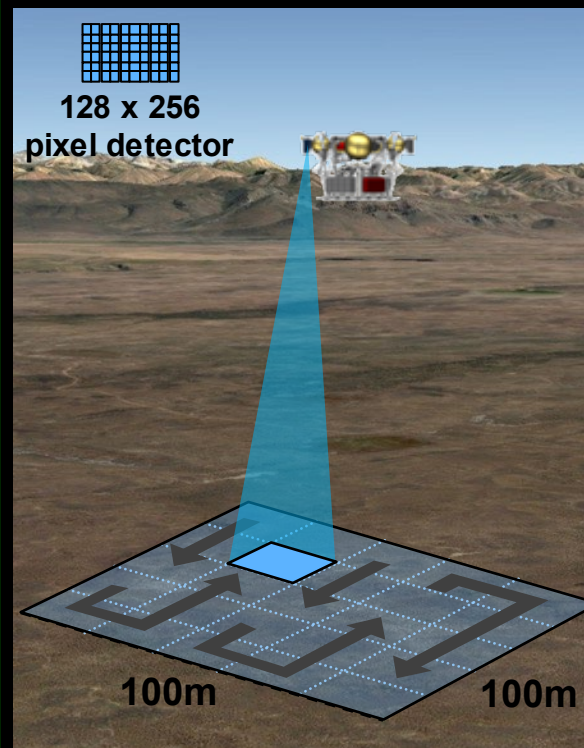
- Simple optics, scan mirrors & data processing
- Very challenging array design
- Link challenges

## Line Scan



- Fairly simple optics, 1 D scan mirrors
- Limited tolerance to pixel loss
- Improved link

## 2D Scan



- Array similar to already fielded lidar systems
- More complex scanning and data processing
- Best link

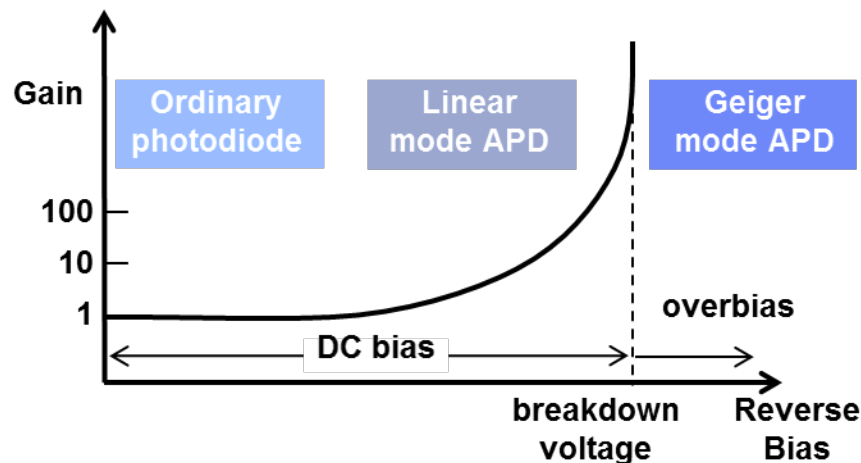
Line scanner selected as best balance between array design, link, and processing complexity



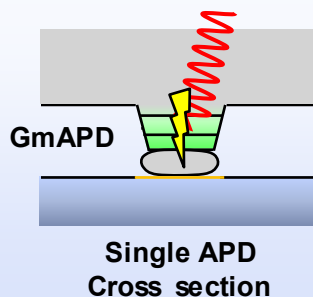


# Geiger-Mode Avalanche Photodiode (GmAPD) Detector Technology

## Photodiode Modes of Operation for MIT/LL Digital Pixel FPAs

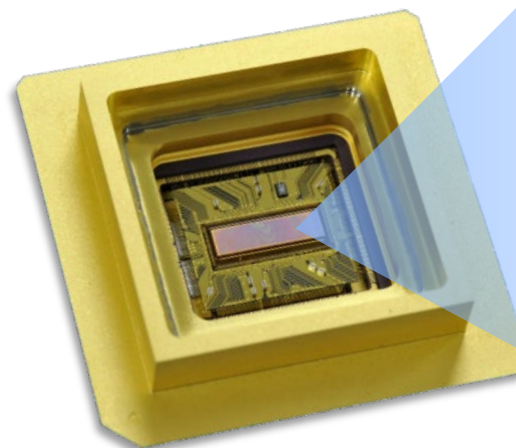


A single photon absorbed by the overbiased APD generates a fast rush of current

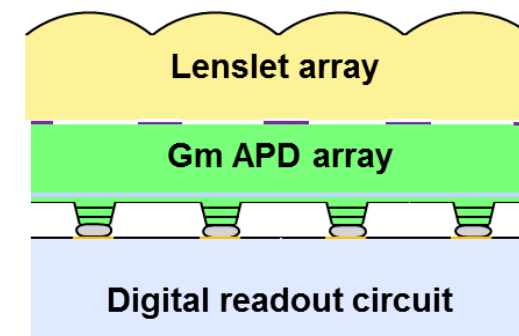


## Geiger-mode APDs provide:

- Single-photon sensitivity
- Lots of current → easy digitization
- Fast breakdown → excellent (sub-ns) time resolution
- TEC accessible temperatures → low SWAP
- Large format arrays
- Stable operation compared to linear mode APD



### Device Cross-section

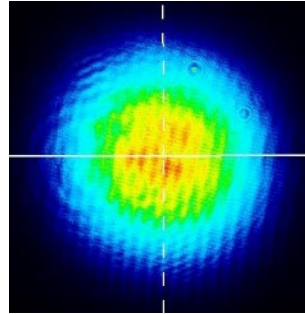
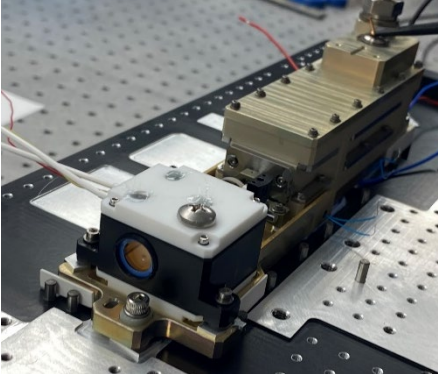




# Ongoing ELSA Assembly and Integration

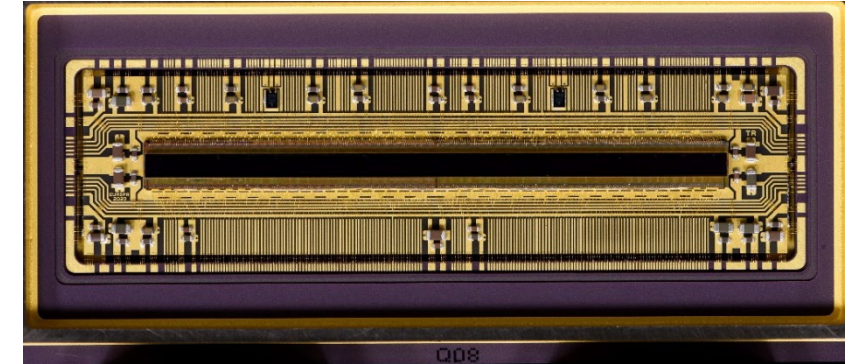
## Lasers

- 532 nm, 10 kHz, 250 mW, 490 ps
- Space-traceable
- **Completed**



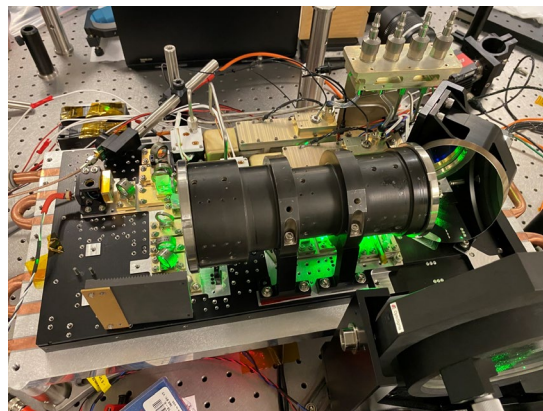
## Detector

- 32 x 2048 pix
- 500ps timing
- Rad-hard design
- **Completed**



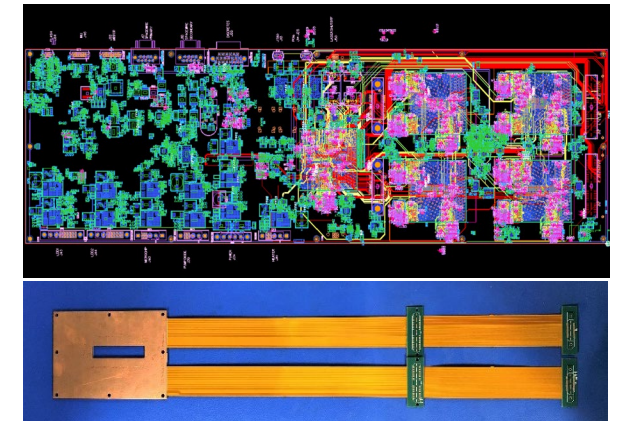
## Opto-mechanical

- Integrated w/ lasers
- Tx & Rx assembled and aligned
- Next: integrate camera



## Electronics & Firmware

- Real-time mapping on 4x rad-tolerant FPGAs
- PCBs in fabrication
- FPGA processing implemented

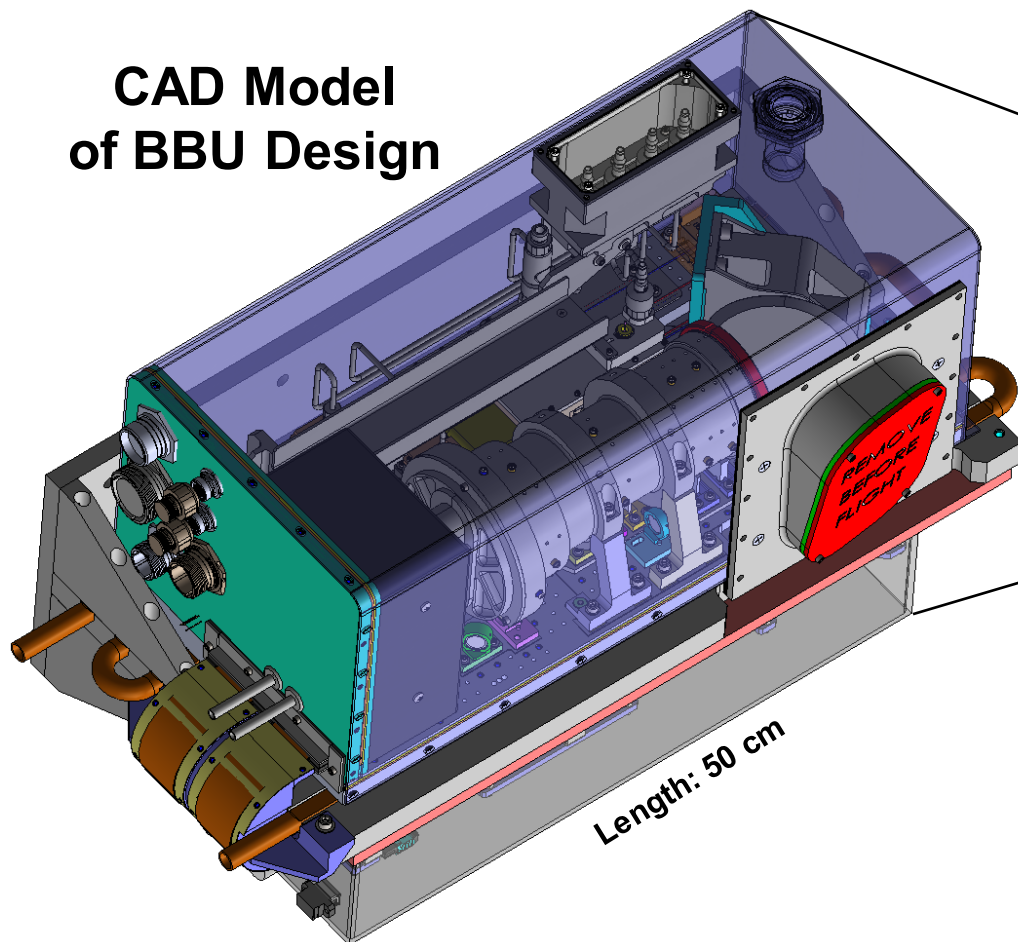






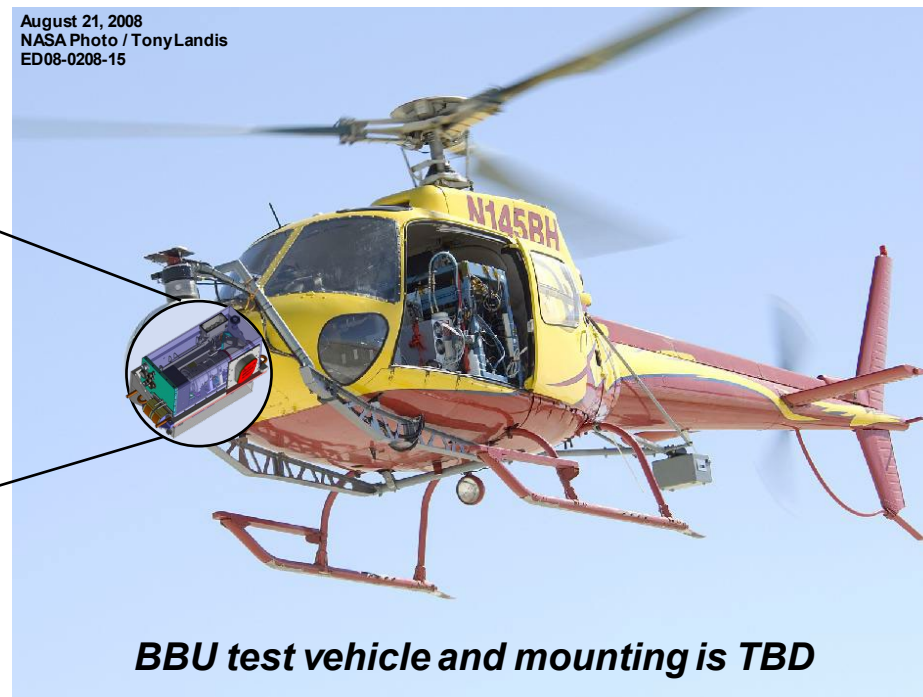
# ELSA Brassboard Validation Testing

**CAD Model  
of BBU Design**



**ELSA brassboard will demonstrate flight-traceable design with real-time FPGA processing and rad hard GmAPD camera**

August 21, 2008  
NASA Photo / TonyLandis  
ED08-0208-15

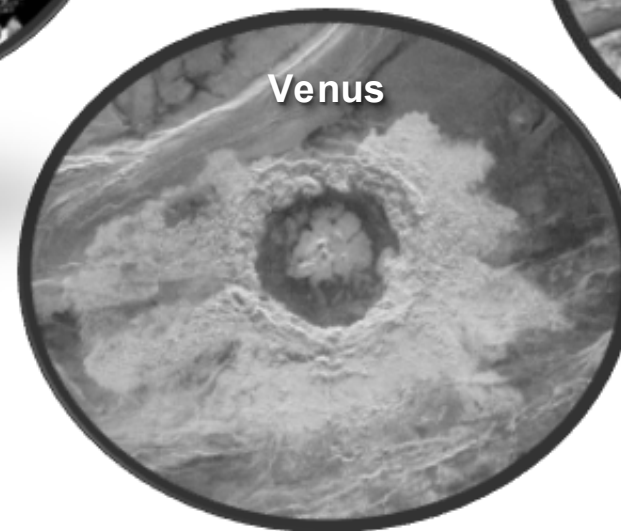
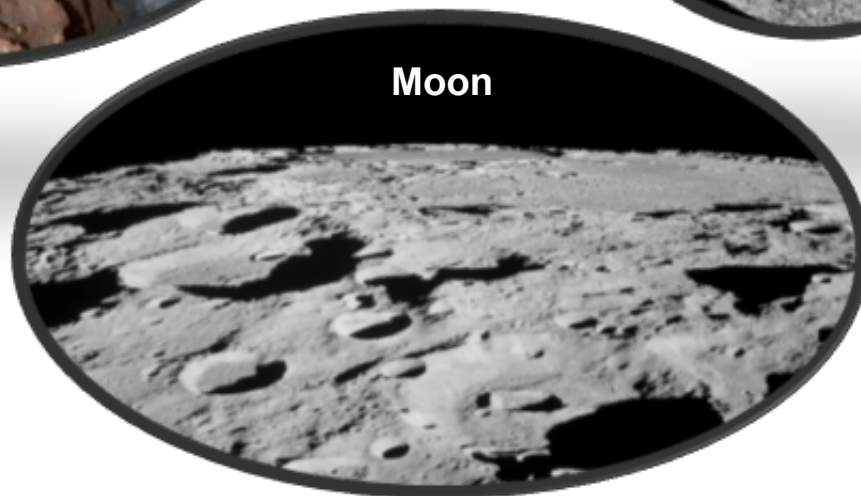
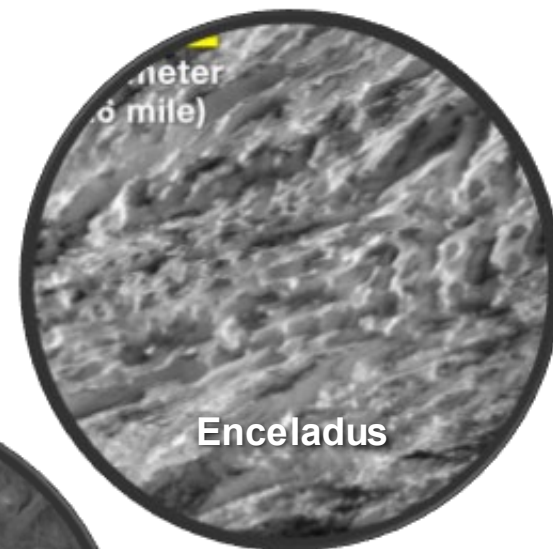
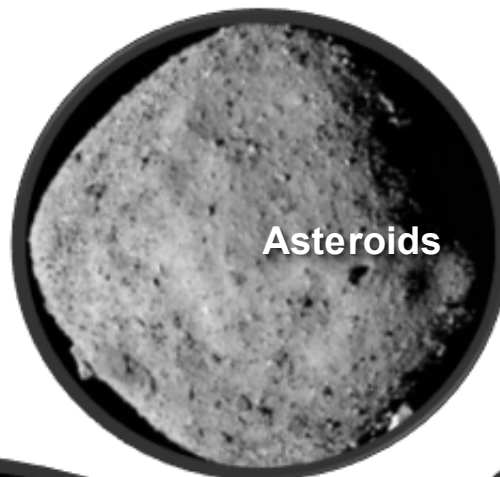
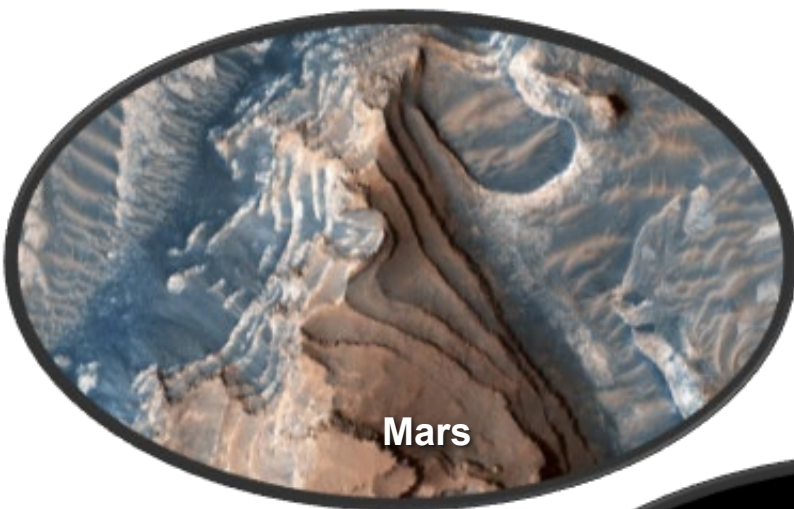


*BBU test vehicle and mounting is TBD*

- Final BBU testing to cover all modes of operation, as test vehicle permits
- Daylight testing limited from significantly more solar background than Europa
- Radiation testing to be done on specific components



# ELSA Elsewhere



Optimized for a Europa mission, ELSA design could be utilized for other space missions





# Summary

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- **NASA has a strong interest to explore Europa for signs of life**
- **Existing imagery of Europa lacking enough detail for site selection and hazard avoidance**
- **ELSA's high resolution imagery and altimetry enables autonomous hazard avoidance and safe landing on Europa**
- **The ELSA design brings significant technical challenges**
- **MIT Lincoln Laboratory has pushed technology and engineering to meet these challenges**
- **ELSA brassboard prototype is scheduled to be tested in 2023**
- **ELSA design general enough to be used for other missions (Enceladus, Moon, etc.)**



# Contact Information

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