



**Jet Propulsion Laboratory**  
California Institute of Technology

**Two-Phase Thermal Technology Laboratory**

# **Two-Phase Thermal Technology Development at JPL**

Taku Daimaru

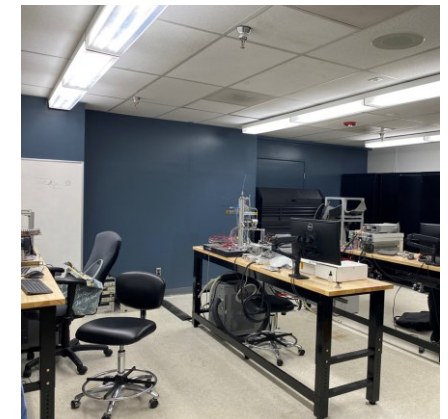
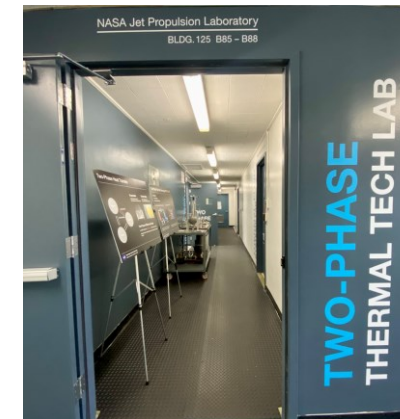
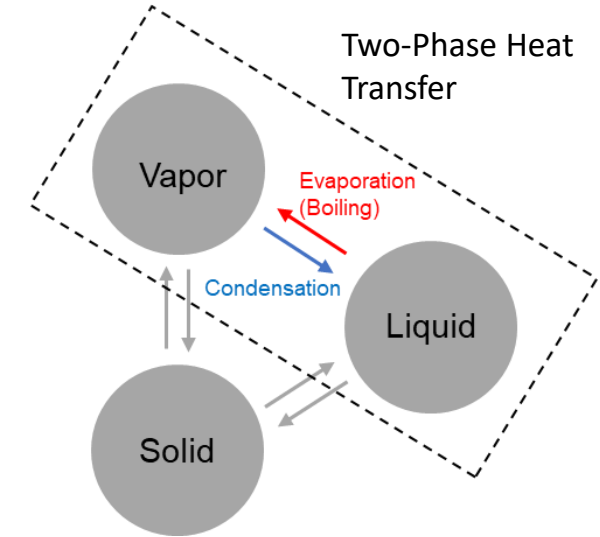
Thermal Engineer / PI of JPL Two-Phase Lab

# JPL Two-Phase Thermal Technology Laboratory

- Utilizing Evaporation (Boiling) and Condensation Heat Transfer
- More Efficient Comparing to Single (Liquid) Phase
  - 1g of Water: Evaporation >> **500 times** >> 1°C Temperature Increases
    - ➡ System can be much lighter or Larger heat rejection
  - Fluids stays at the constant temperature when Evaporation/Condensation happens.
    - ➡ Isothermal / Precise temperature control by controlling pressure

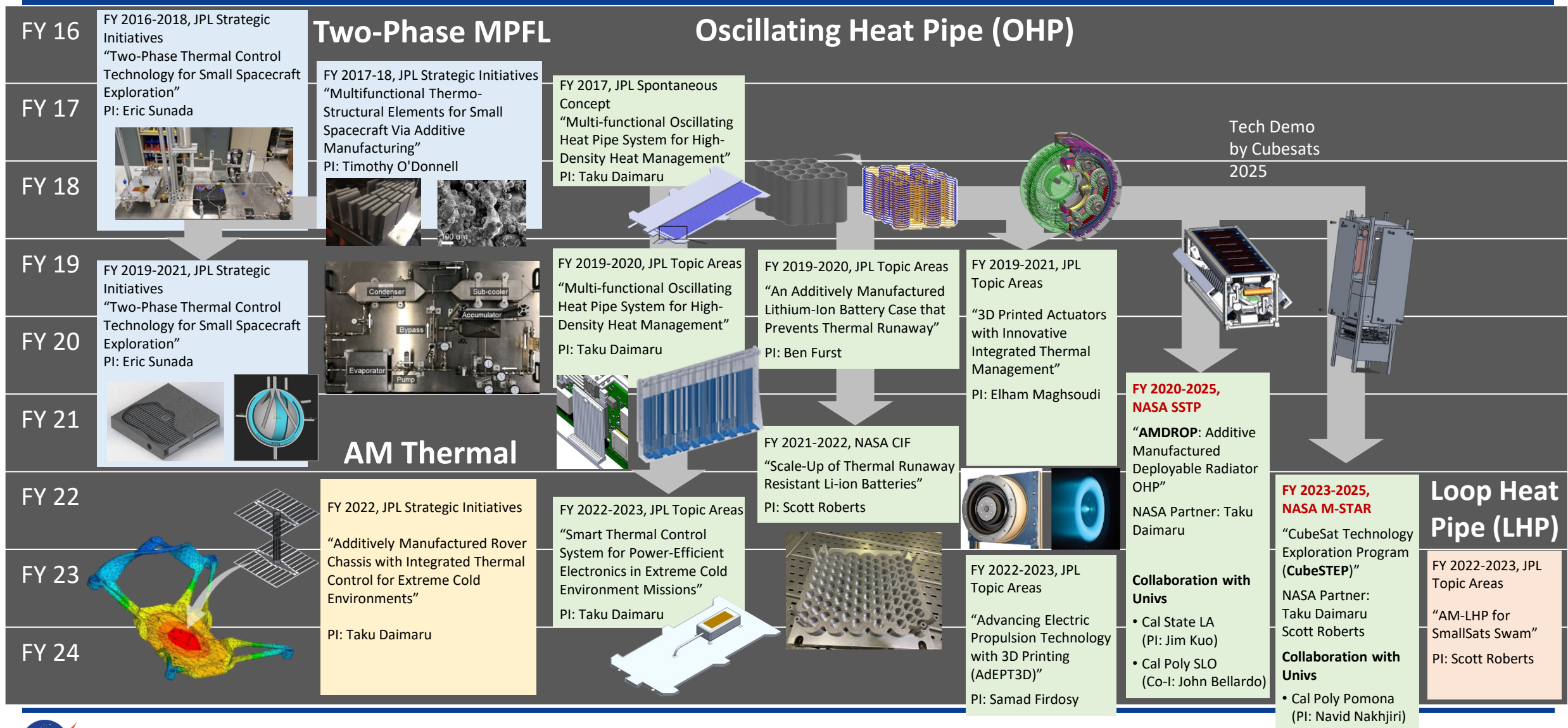
## Primary research areas in JPL

- Two-Phase Mechanically Pumped Fluid Loop (2016-2022)
- Oscillating Heat Pipe (2017-2025)
- AM Rover Chassis with Integrated Thermal Control (2022-2024)
- Loop Heat Pipe (2023-2025)



JPL Two-Phase Thermal Technology Laboratory (2024)

# JPL Two-Phase Thermal Technology Lineage

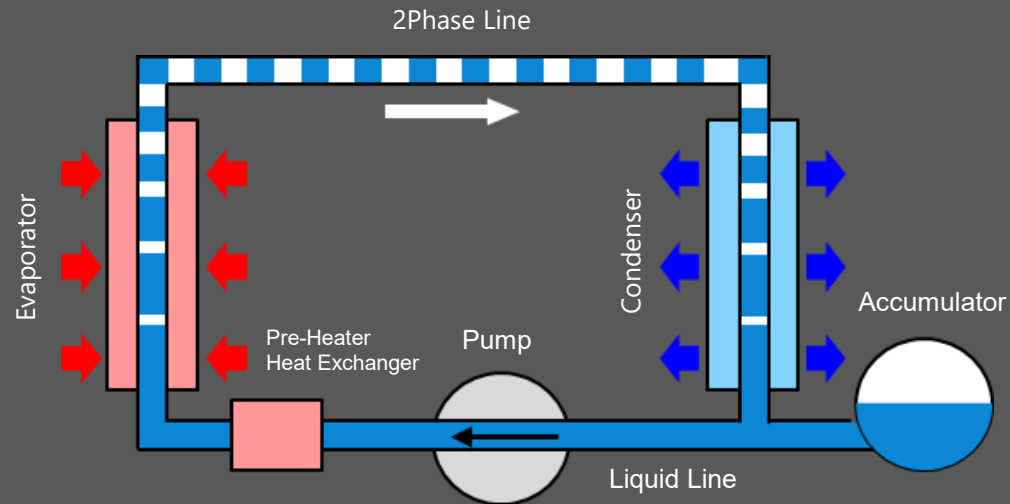


# Primary Two-Phase Thermal Technologies

## Key Words

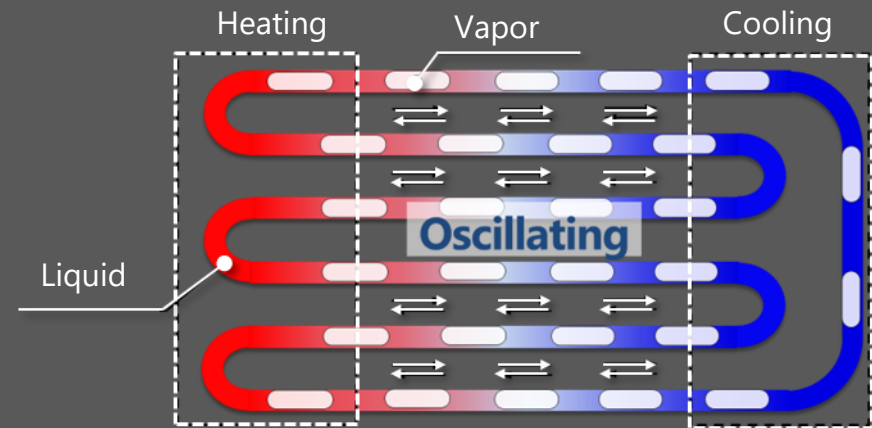
- Two-Phase Flow: More Efficient Heat Transport
- Additive Manufacturing: Integrate Thermal Control into Structure

## Two-Phase Mechanically Pump Fluid Loop



- Circulate fluid by a pump, heat transport in long distance > 10m
- Evaporation in Evaporator and Condensation in Condenser

## Oscillating Heat Pipes (OHPs)

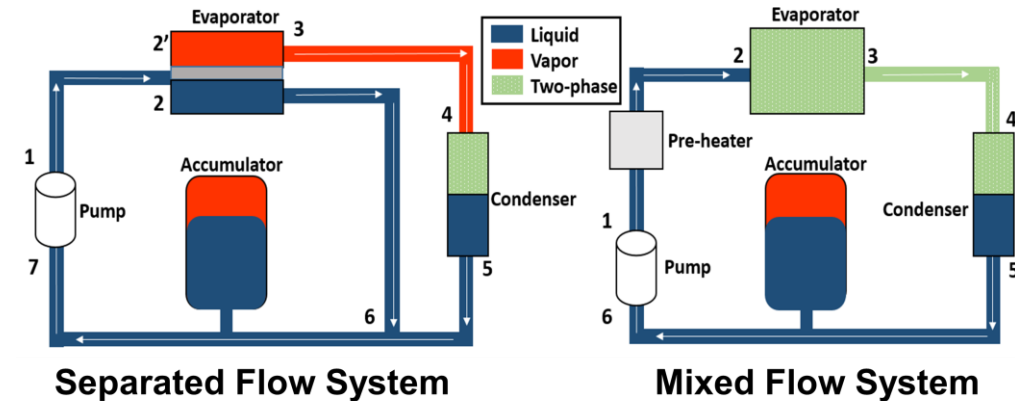


- Heat pipe using self-excited oscillation of fluid, heat transport < 1m
- Heat transport by evaporation/condensation of micro liquid film

# Two-Phase Mechanical Pump Fluid Loop

## System Architecture

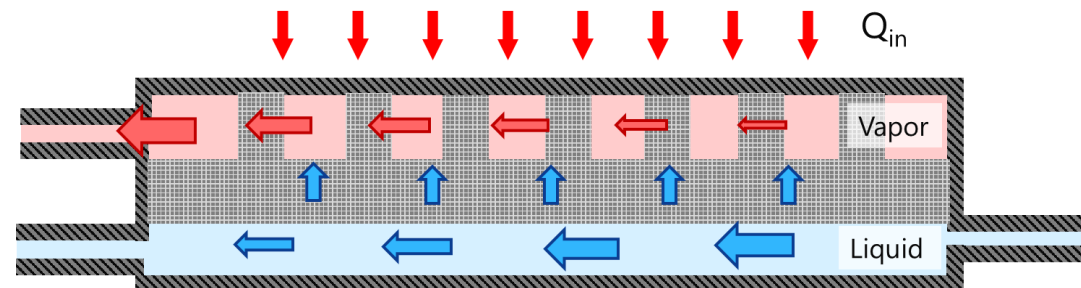
- Mixed Flow System
  - Preheating / Two-Phase flow
- **Separated Flow System**
  - No Preheating / Vapor Flow



System Architecture  
(Furst 2017)

## Wick-Type Evaporator

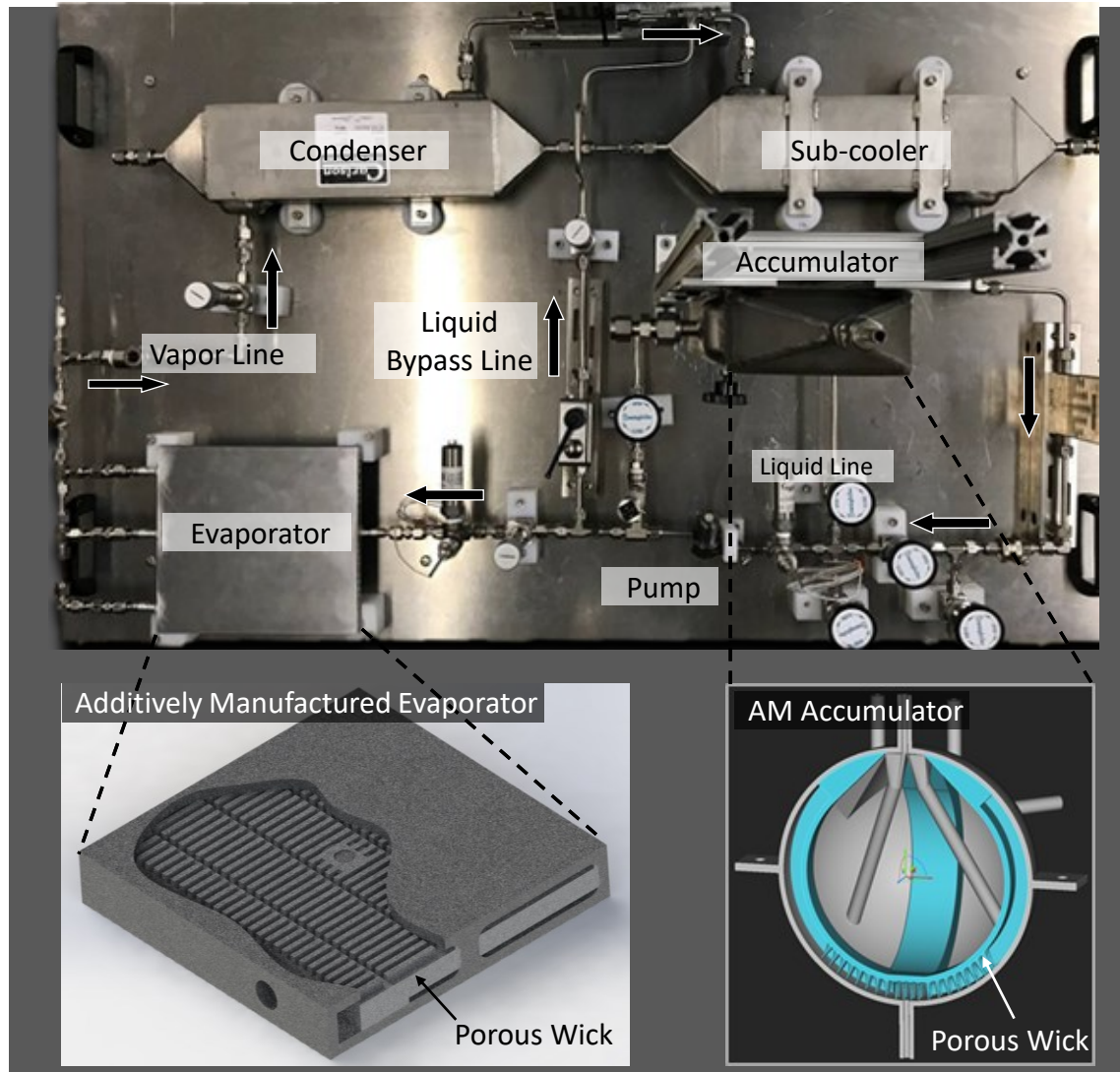
- Working in Separated Flow System
- Phase Separation
- Ability to Adjust Mass Flow Rate to  $Q$ 
  - Wick pulls liquid up as needed



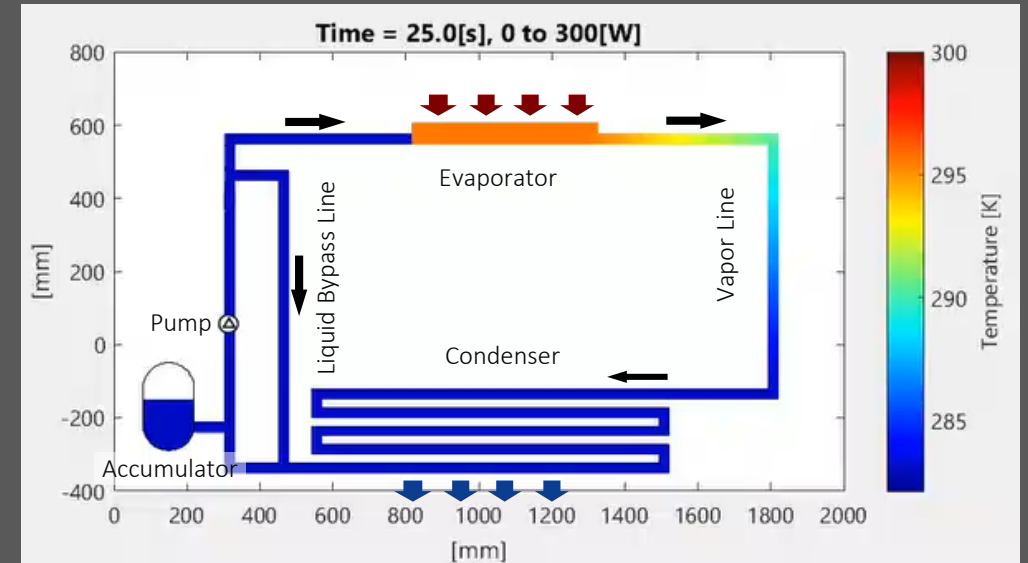
Wick-Type Evaporator



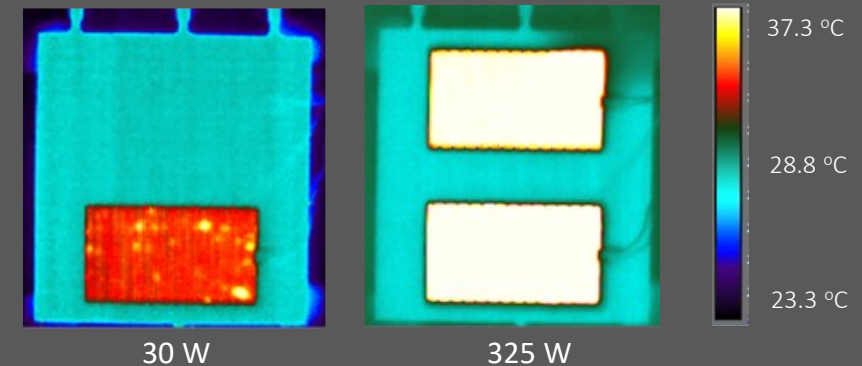
# Two-Phase Mechanical Pumped Fluid Loop (TRL 4~5)



Numerical Simulation

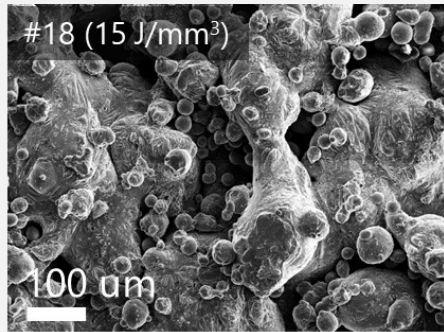


Evaporator IR Image



# Evaporator Development

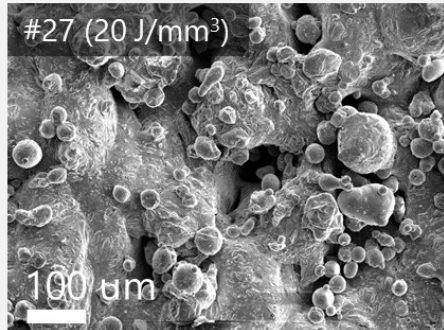
## Porous Material Development in AM



**Baseline**

**#23 (15J/mm<sup>3</sup>)**

Pore Radius: 18.0 μm  
Permeability:  $3.2 \times 10^{-13}$   
Porosity: 22.3 %



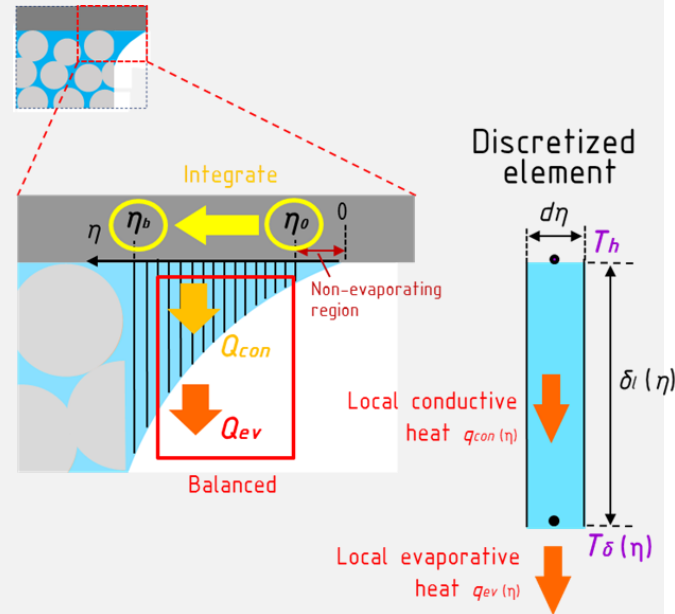
**Challenge**

**#30 (20J/mm<sup>3</sup>)**

Pore Radius: 2.9 μm  
Permeability:  $1.2 \times 10^{-14}$   
Porosity: 12.2 %

**Aluminum**

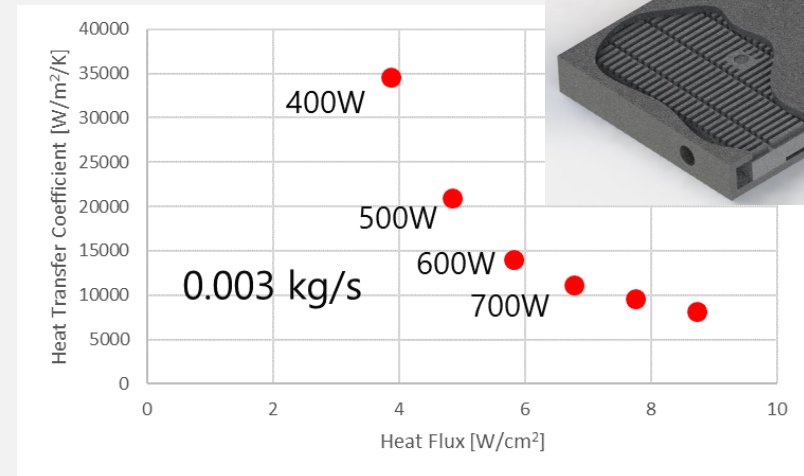
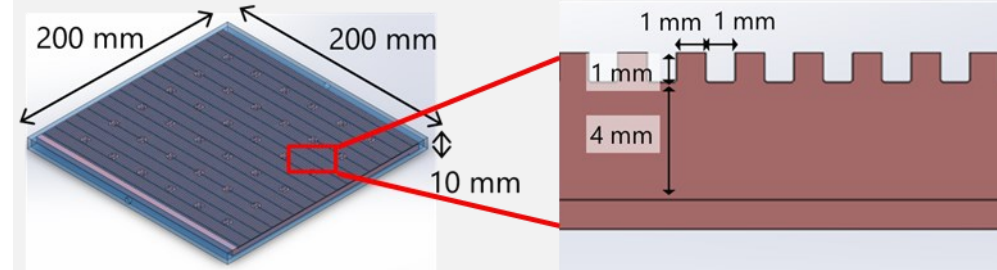
## Theoretical Modeling



## Heat Transfer Coefficient Model

Odagiri (2017)

## Designing

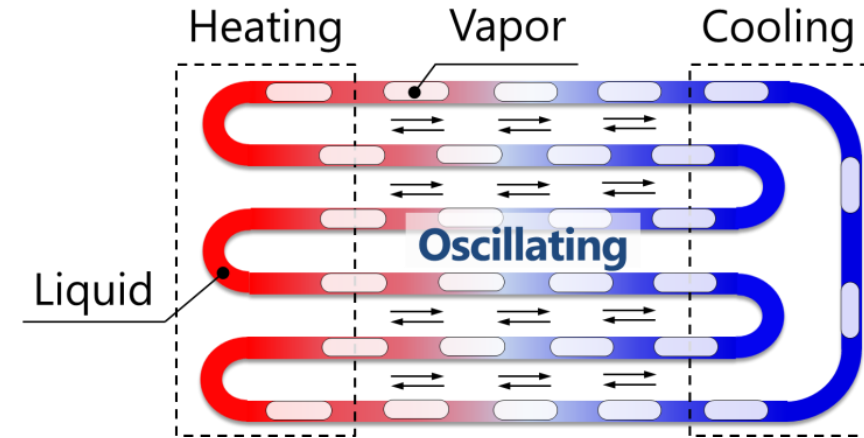


## Heat Transfer Coefficient

# Oscillating Heat Pipe (OHP)

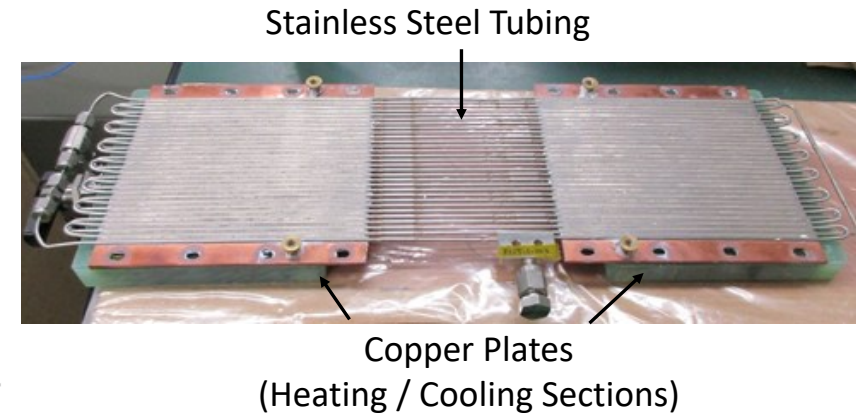
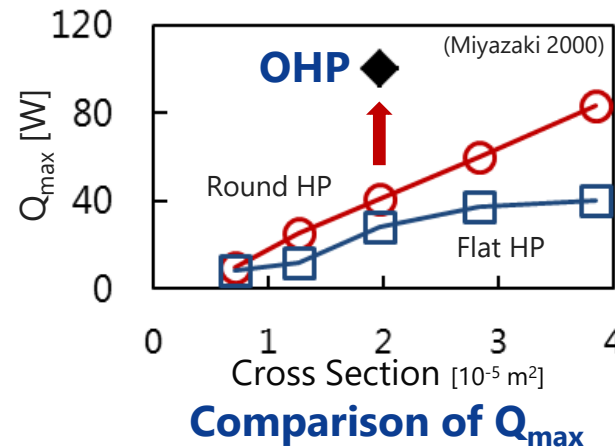
## Operational Principle

- Metal Pipe, Working Fluid (Liquid Slug – Vapor Plug)
- Transport heat by Self-Excited Oscillation of working fluid
- Utilizes pressure-driven, two-phase fluid flow to rapidly transfer heat between heat sources and heat sinks.
- Heat transferred by Phase Change in Liquid Films



## Merits and Challenge

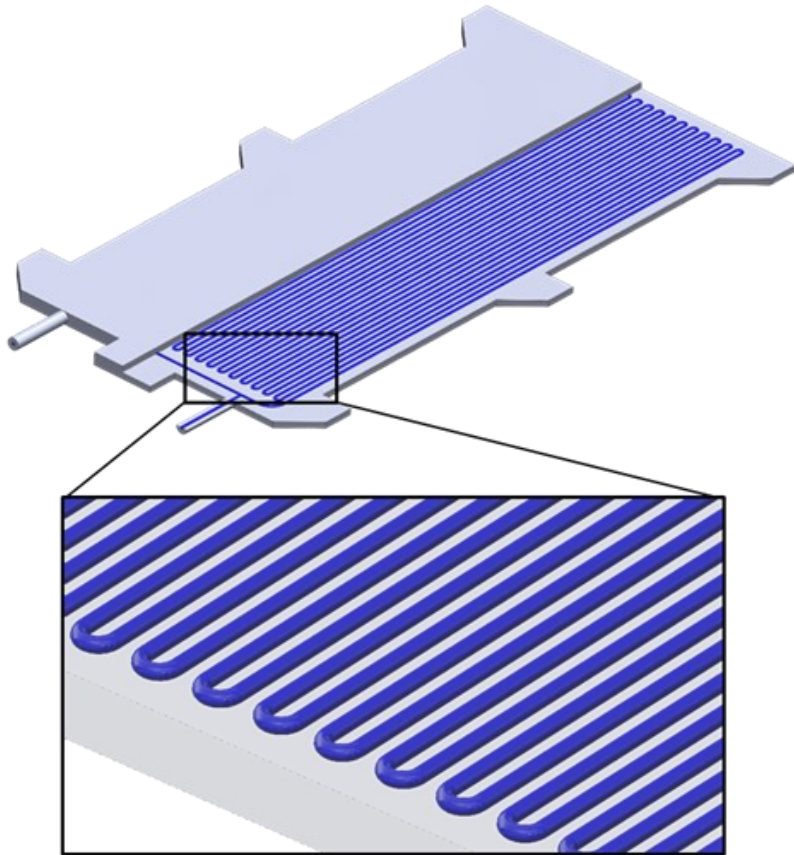
- Can handle higher Maximum Heat Input  $Q_{\max}$
- Thinness • Light Weight • Flat-Plate
- No Porous Structure (Wick)
- Flexibility in Channel Arrangement
- ➡ **Additive Manufacturing (AM)**
- Complex Physics
- ➡ **Numerical Simulation**



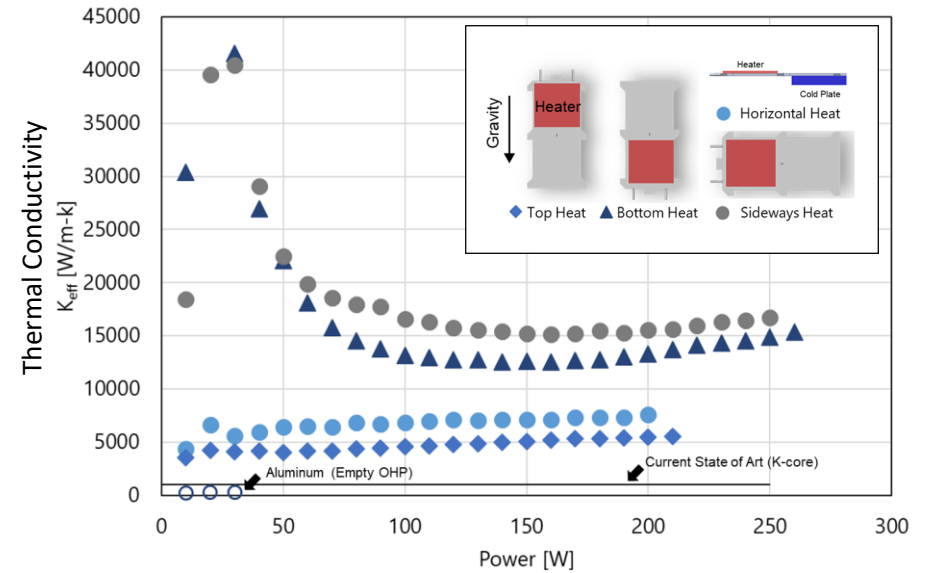
**Example of Conventional OHP (JAXA)**



# Additive Manufactured OHP (AMOHP) (TRL 5)



- Material : AlSi10Mg
- Number of Turns : 42
- Channel Diameter : 1 mm
- Turn Radius : 1 mm
- Plate Length : 20 cm
- Plate Width : 9 cm
- Plate Thickness : 4 mm

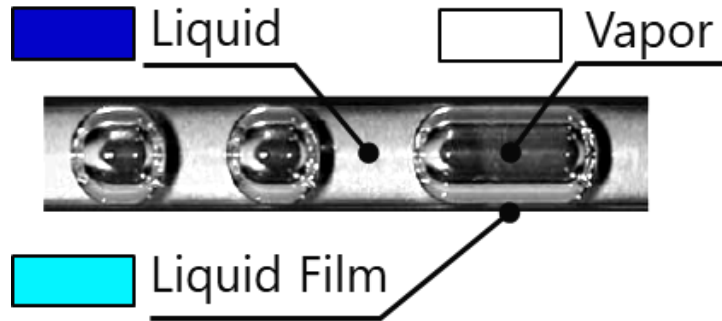


- Collaboration with JPL Additive Manufacturing Center
- High Thermal Conductivity
  - 7,500 W/m/K (Horizontal Heat)
  - 40,000 W/m/K (Bottom Heat)
  - 5-15X higher than Current State of the Art (K-core)

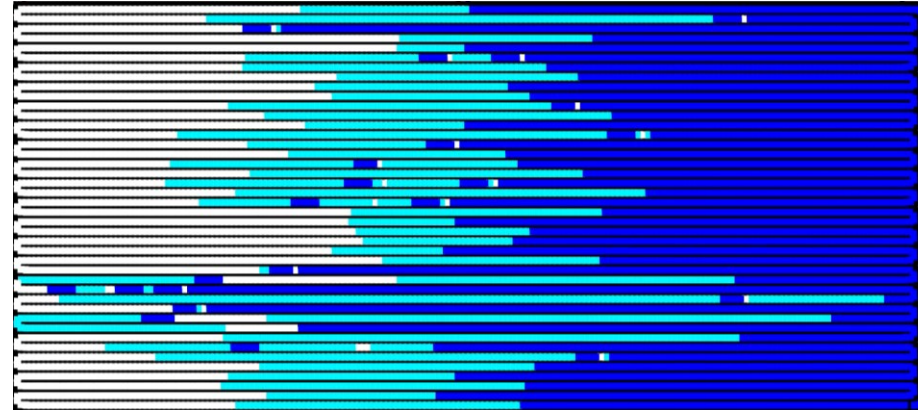
# OHP Numerical Simulation Model

**100 W**

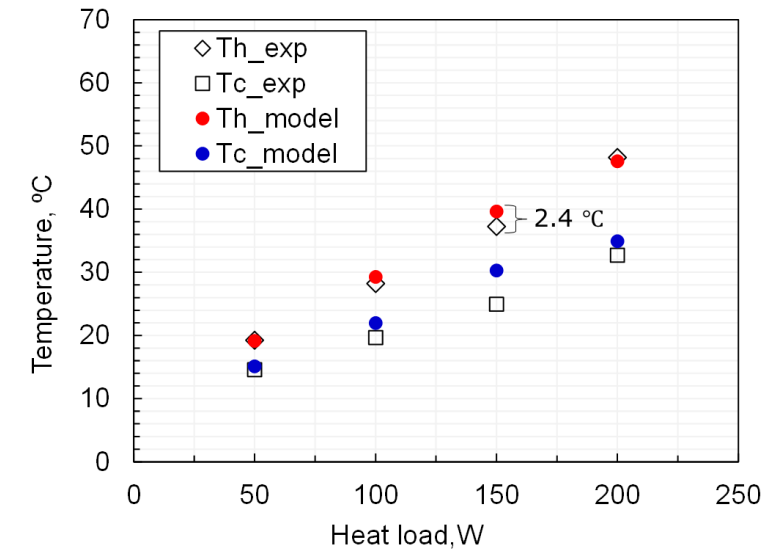
Time = 30 sec  
Steady-state  
1.0 x speed



## Change in liquid-vapor distribution



## Change in temperature distribution

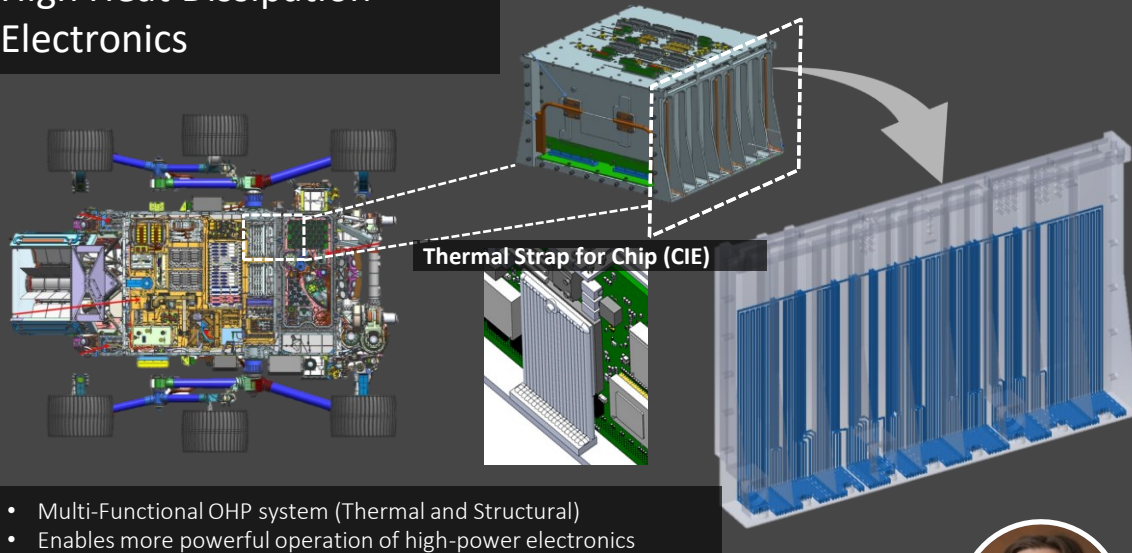


- High precision numerical model based on Two-Phase Physics
  - Movement of Liquid/Vapor
  - Boiling, Liquid Film Evaporation
  - Heat Transfer in Structure
- Being used as OHP design tool



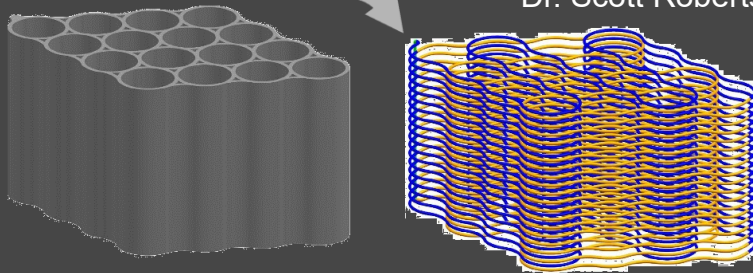
# Applications of AMOHPs in JPL

## High Heat Dissipation Electronics



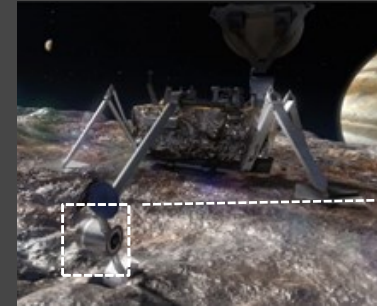
## Li-Ion Battery Case

- Thermal Management of entire Battery Unit
- Preventing Thermal Run-away which generate high heat-density dissipation

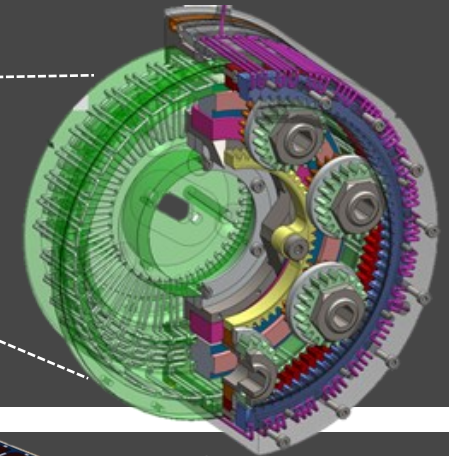


Dr. Scott Roberts

## Actuator Thermal Control For Planetary Sampling



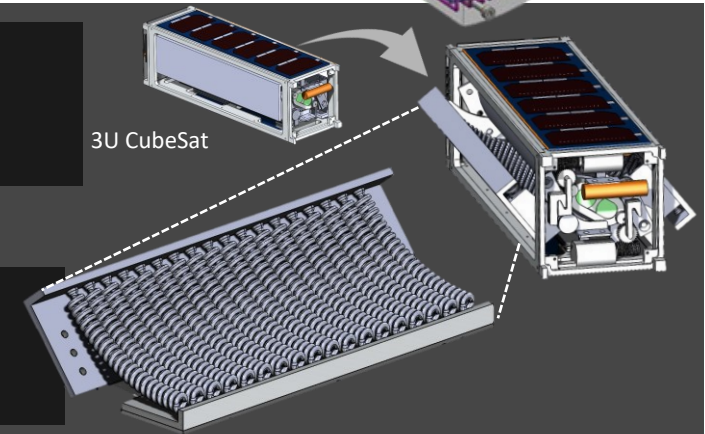
- More efficient warming-up in cold environment
- Enables high-power heat dissipation during actuator operation



## Deployable Radiator for CubeSat (AMDROP)

- NASA SSTP CSLA/Cal Poly SLO/JPL

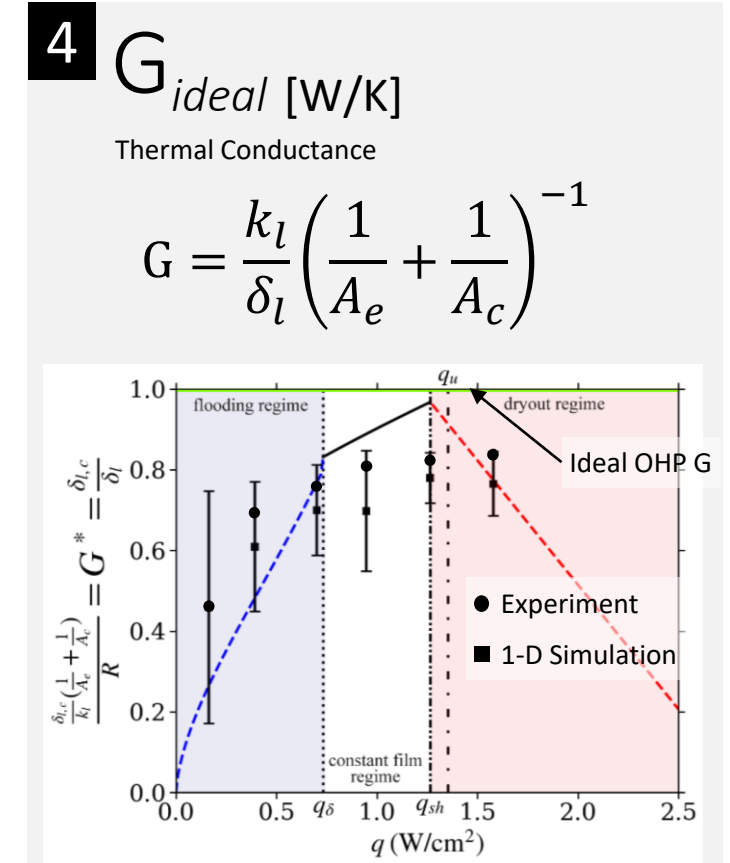
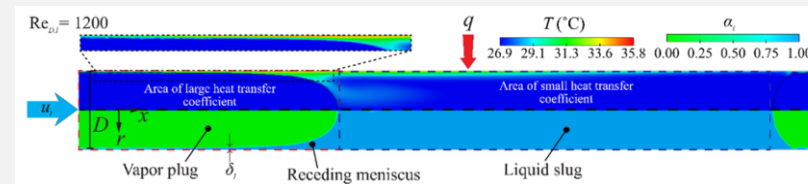
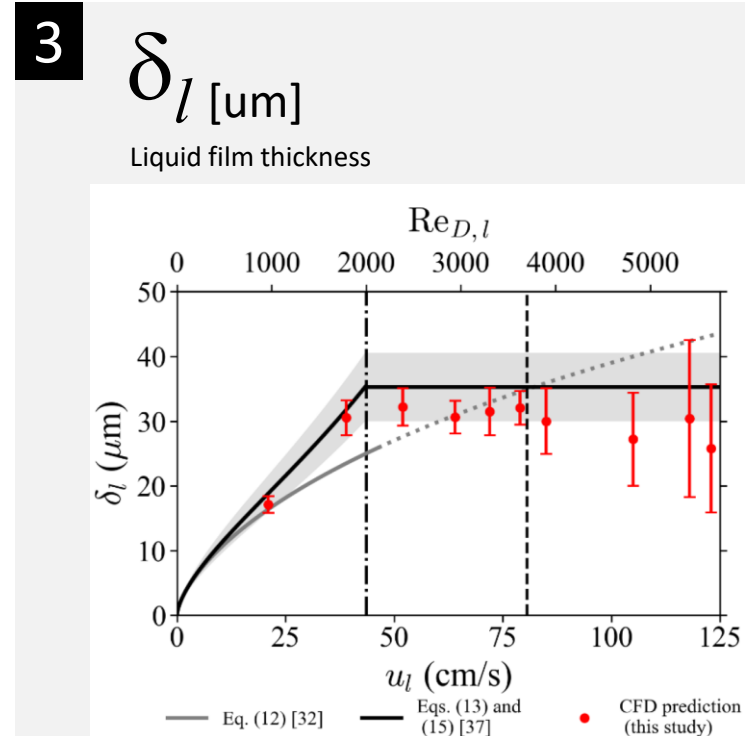
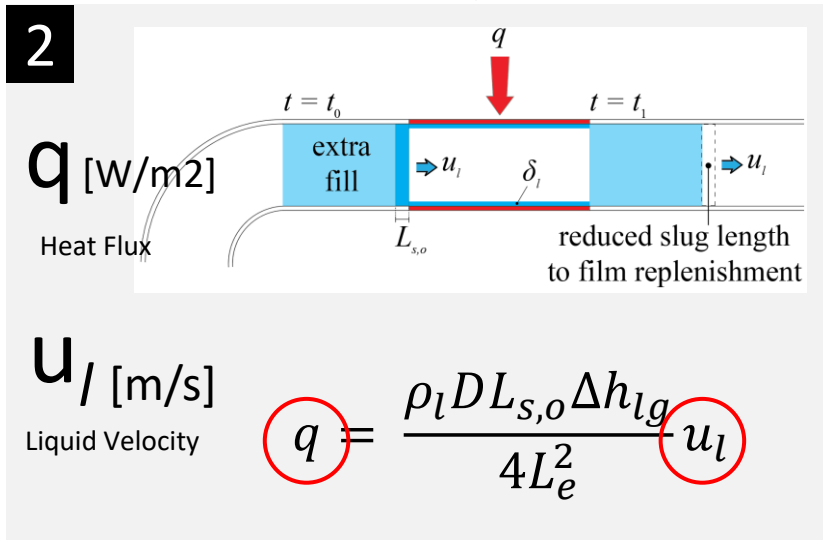
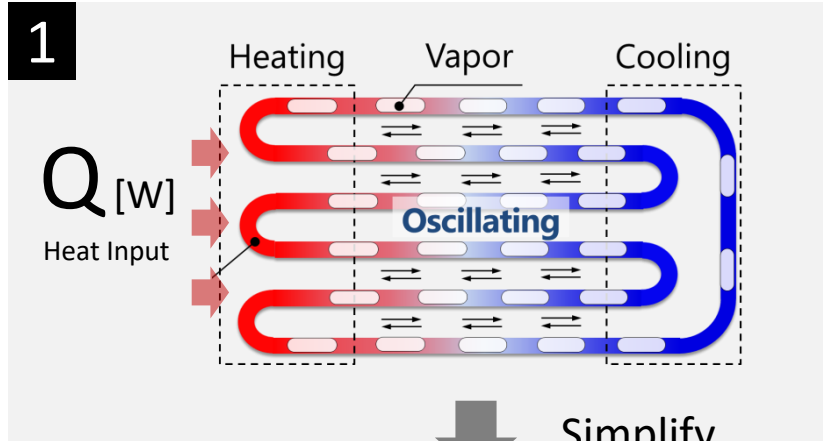
- AMDROP allows high power instrumentation/computation in CubeSats by enabling 100W class heat removal



# OHP Performance Prediction Model (Collaboration with University of Michigan)



Prof. Massoud Kaviany



Experimental Result is about 80% of  $G_{ideal}$

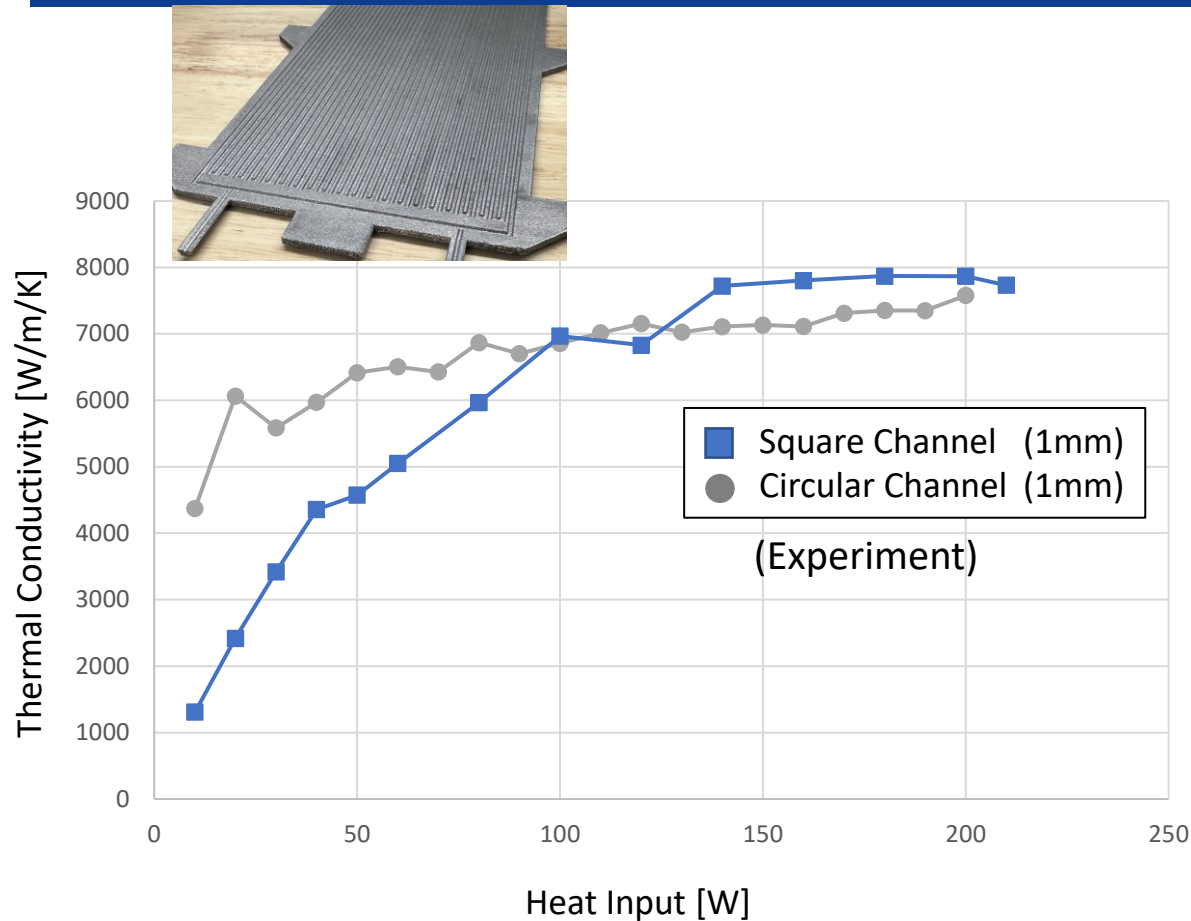


# OHP Channel Geometry

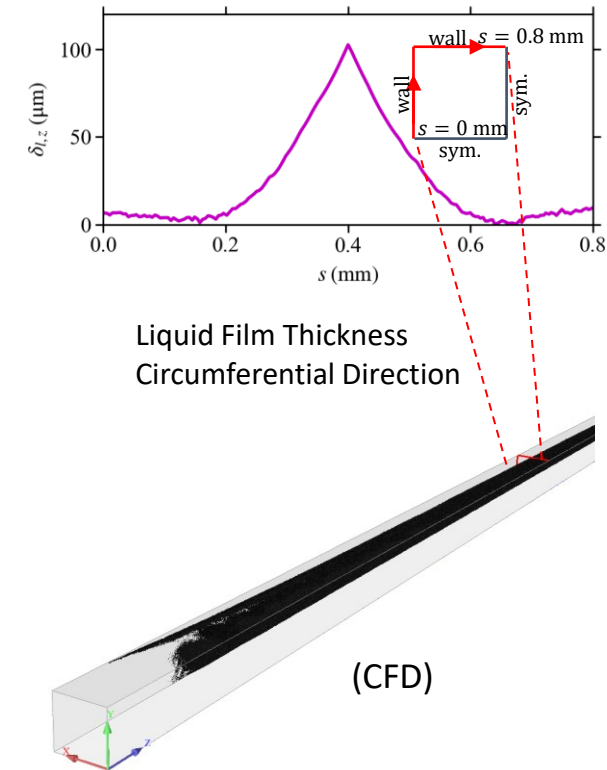
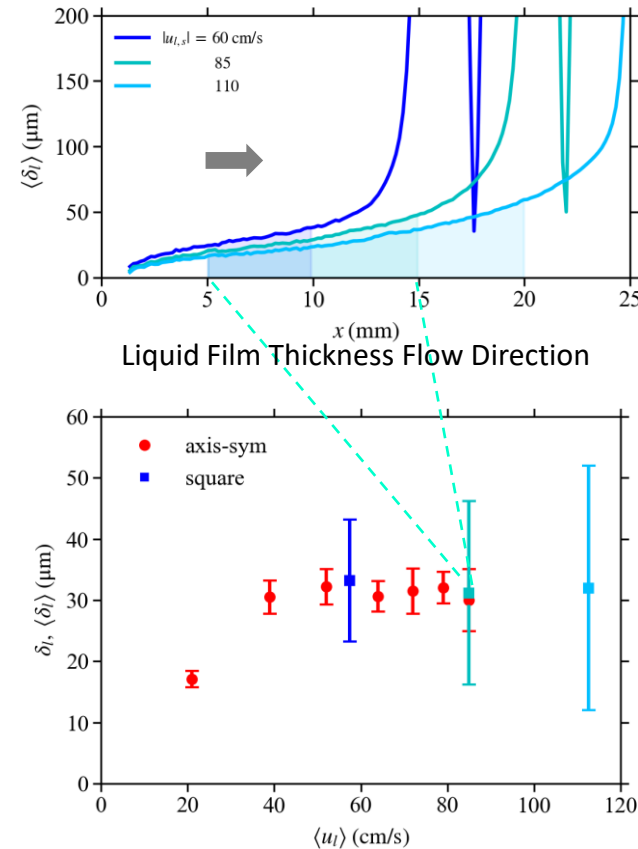
(Collaboration with University of Michigan)



Prof. Massoud Kaviany



Square channel is low-performing at lower heat, but slightly higher at high heat



- More Liquid Mass -> Higher Pressure Difference
- Thinner Liquid Film Section, Larger Surface Area

# OHP in Various Material (All 3D Printed)

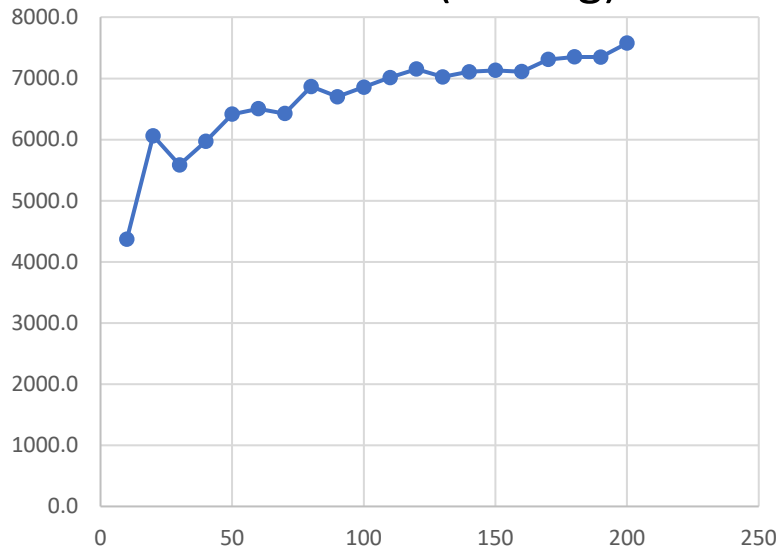


Michael Cox



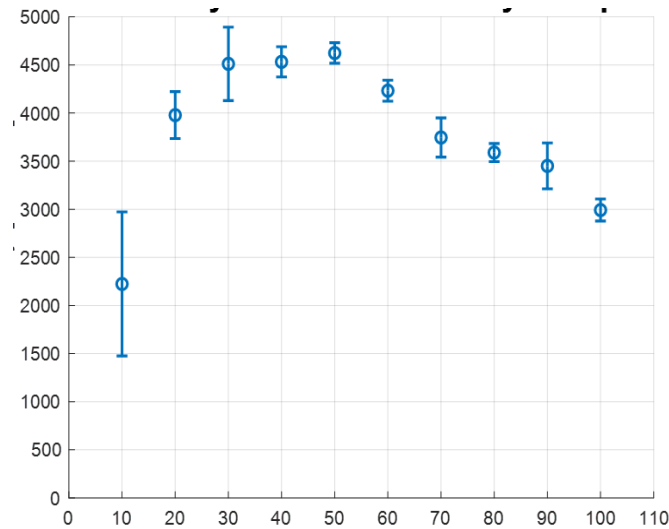
Tomas Wexler

## Aluminum (AlSiMg)



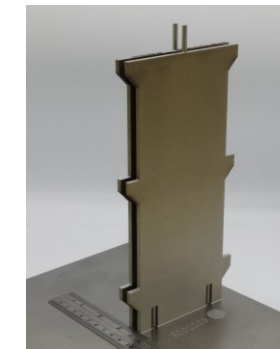
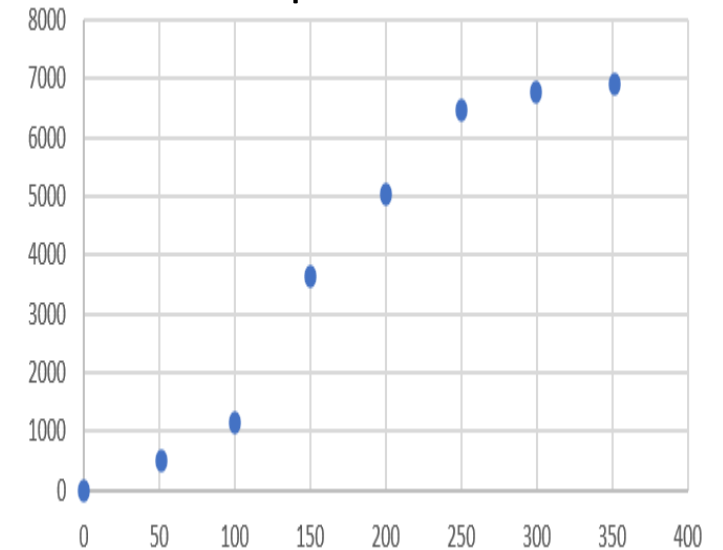
- Working Fluid: R134a
- 1mm Diameter Channel
- Operation Range -10 ~ 60C
- Material k: 180 W/m/K

## Titanium



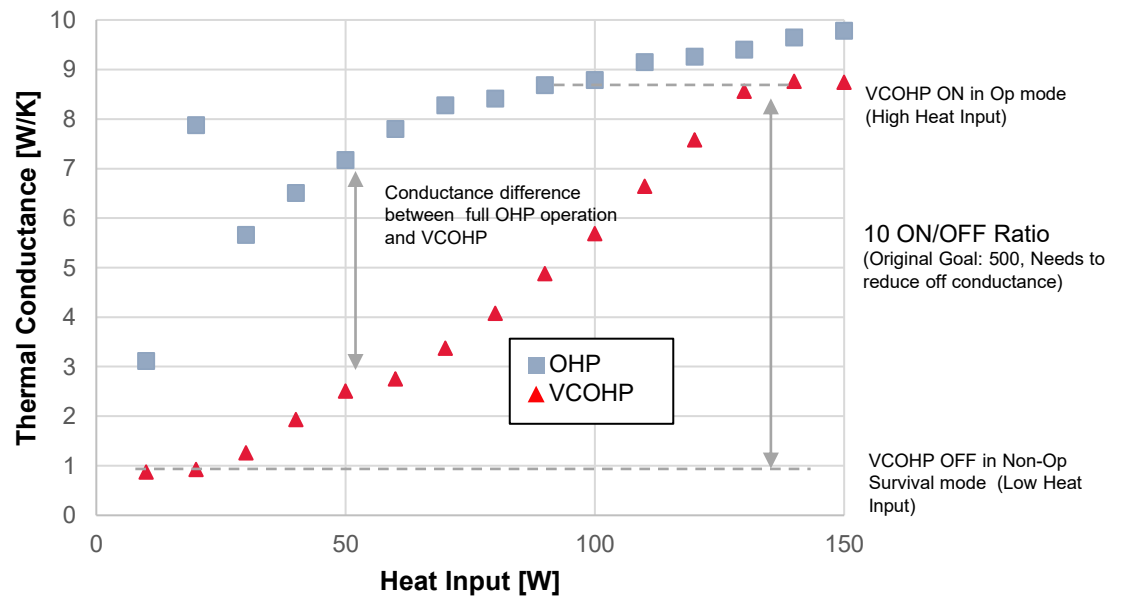
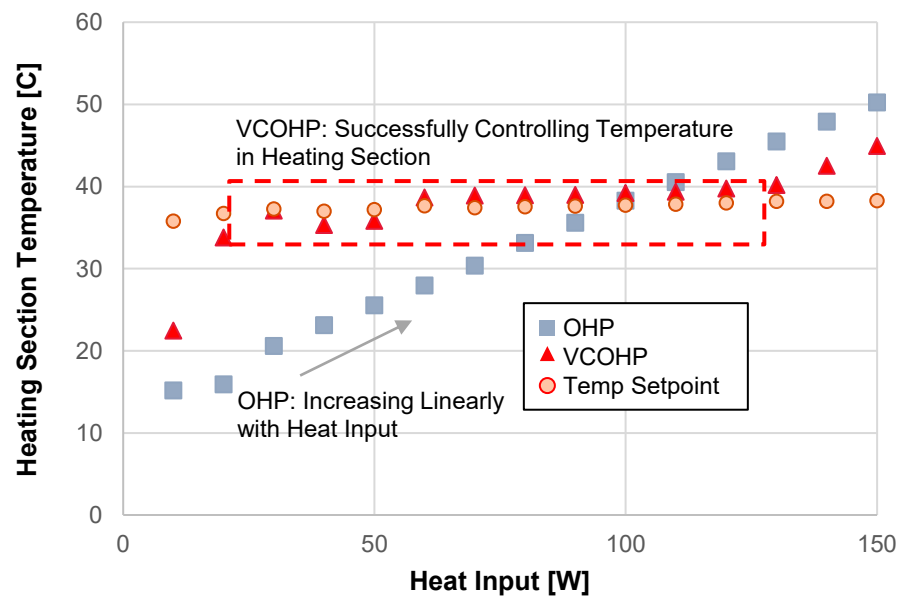
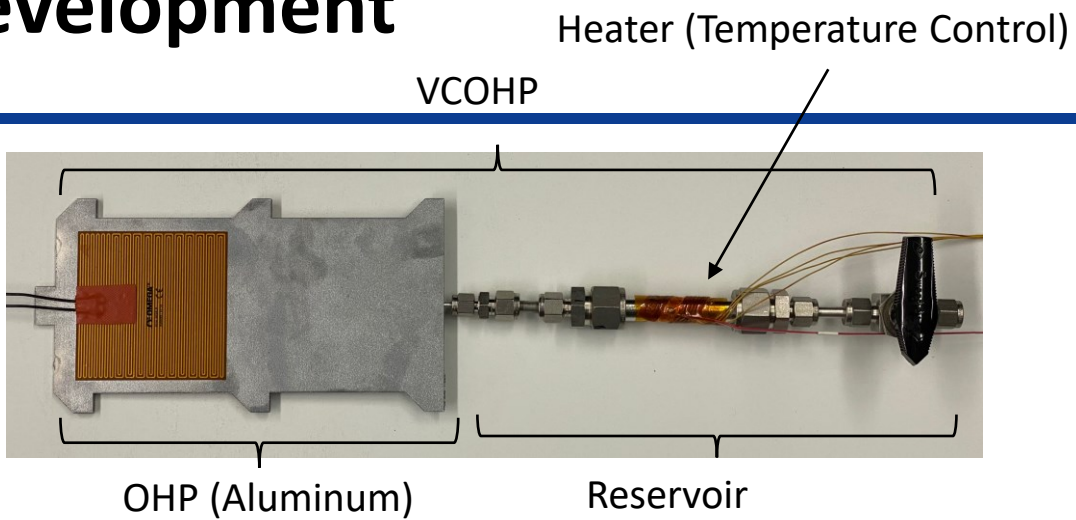
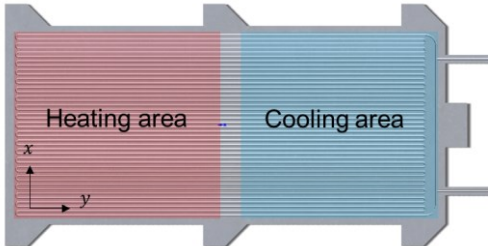
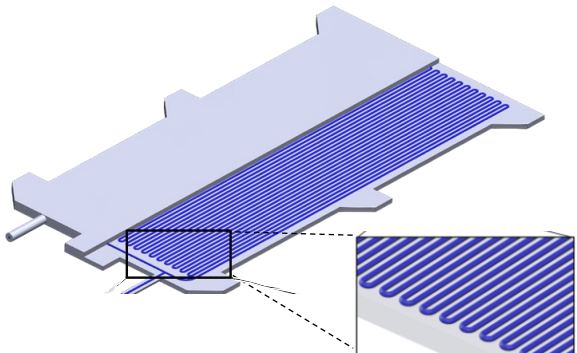
- Working Fluid: R134a
- 1mm Diameter Channel
- Operation Range -10 ~ 60C
- Material k: **6.7 W/m/K**

## Hiperco 60

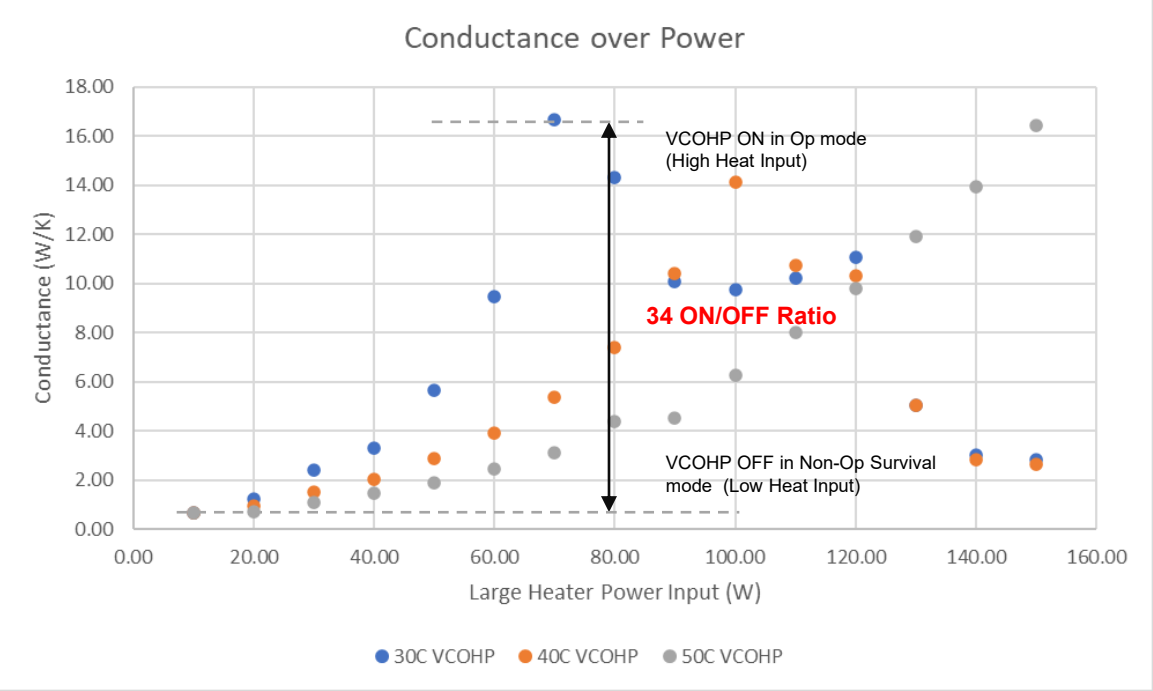
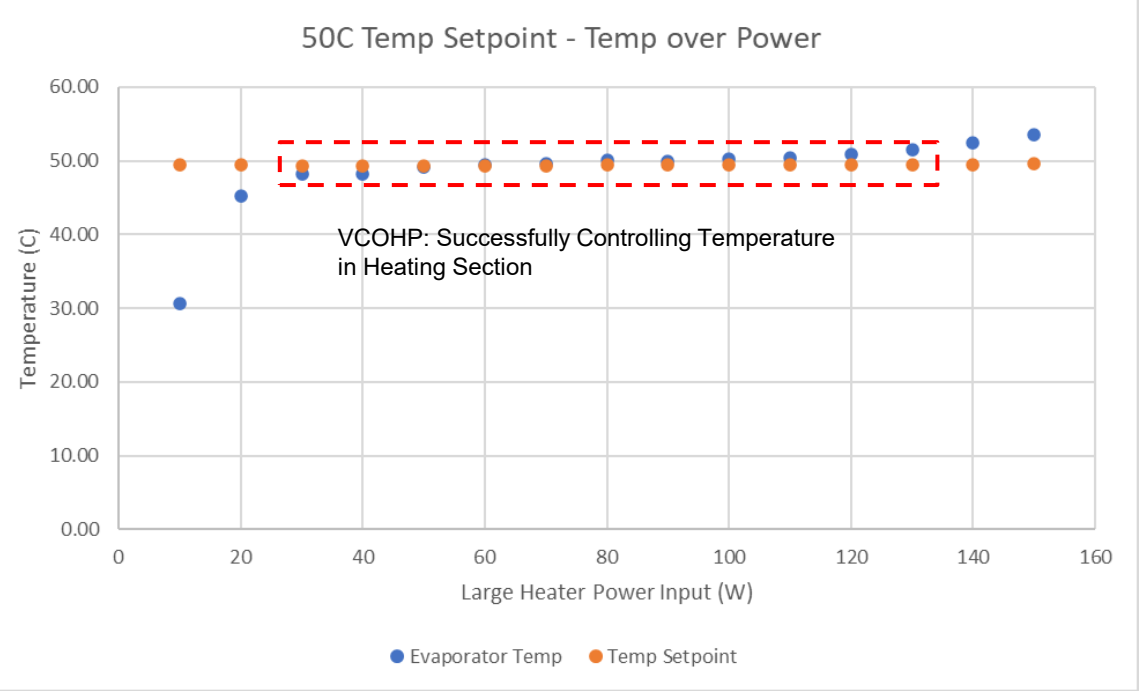
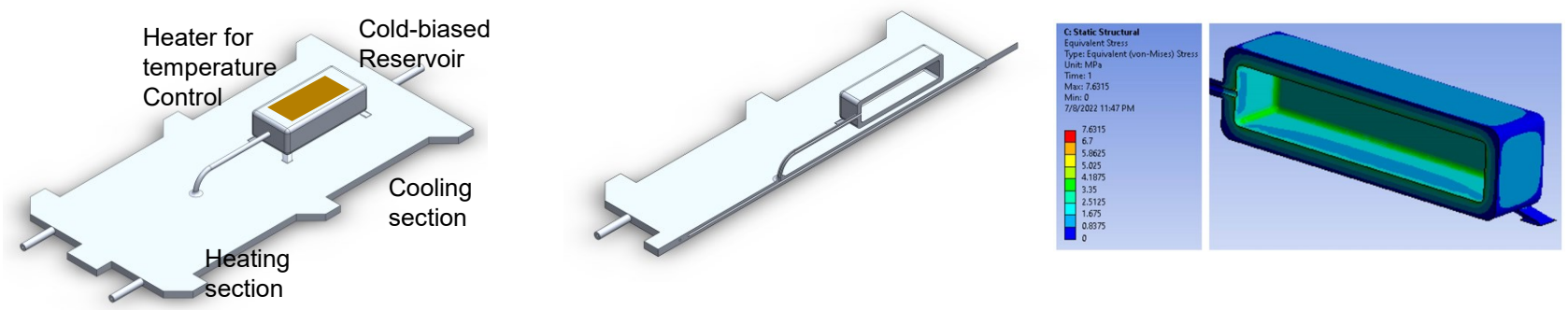


- Working Fluid: Water
- 2mm Diameter Channel
- Operation Range 25 ~ 300C
- Material k: **30 W/m/K**

# Temperature Controllable OHP Development



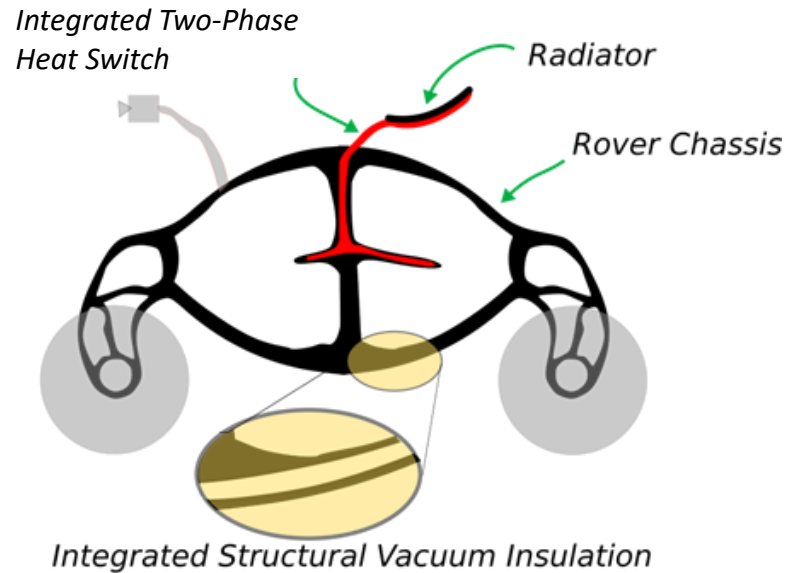
# AM-VCOHP Development





# Additive Manufacturing Rover Chassis

## with Topology Optimization and Integrated Two-Phase Thermal Control



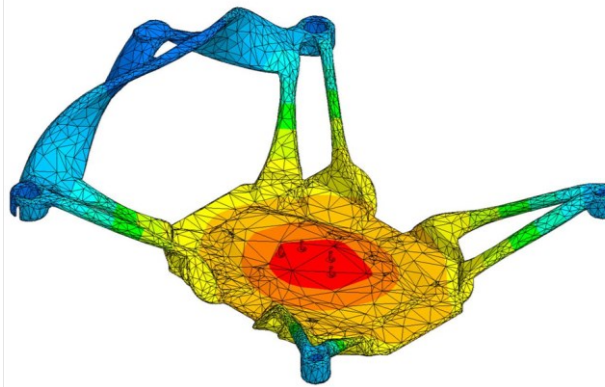
### Concept of AM Rover Chassis

#### Benefits compared to SOP/SOA:

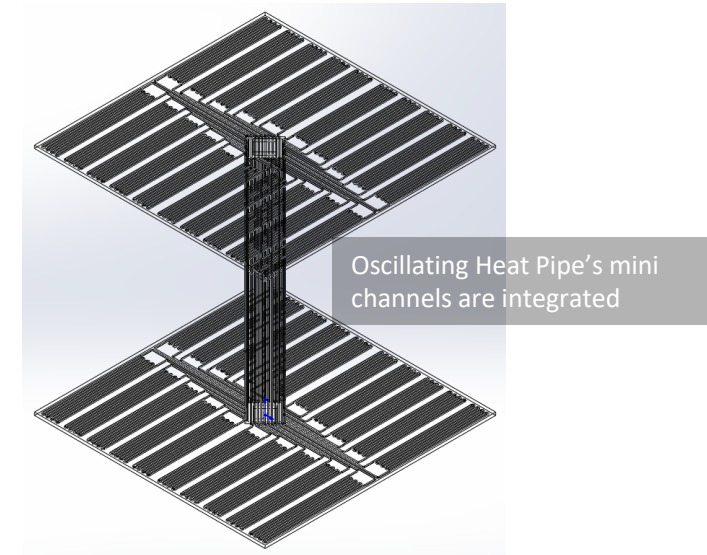
- For extremely cold environment missions
  - ✓ Lunar Surface
  - ✓ Mars Polar Region
  - ✓ Ocean-Worlds
- AM two-phase heat switch (THS) enables high heat rejection
- AM structural vacuum insulation (SVI) reduces heat loss
- Integrated AM thermal/structural design reduces mass and heat loss



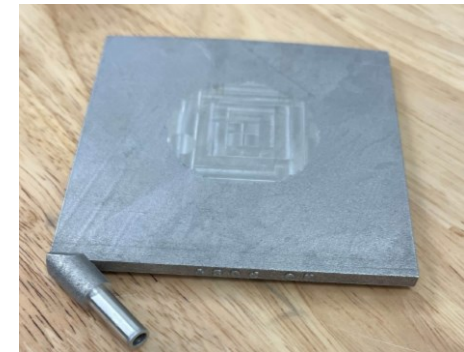
### Mechanical and Thermal Topology Optimization



### Thermal Simulation

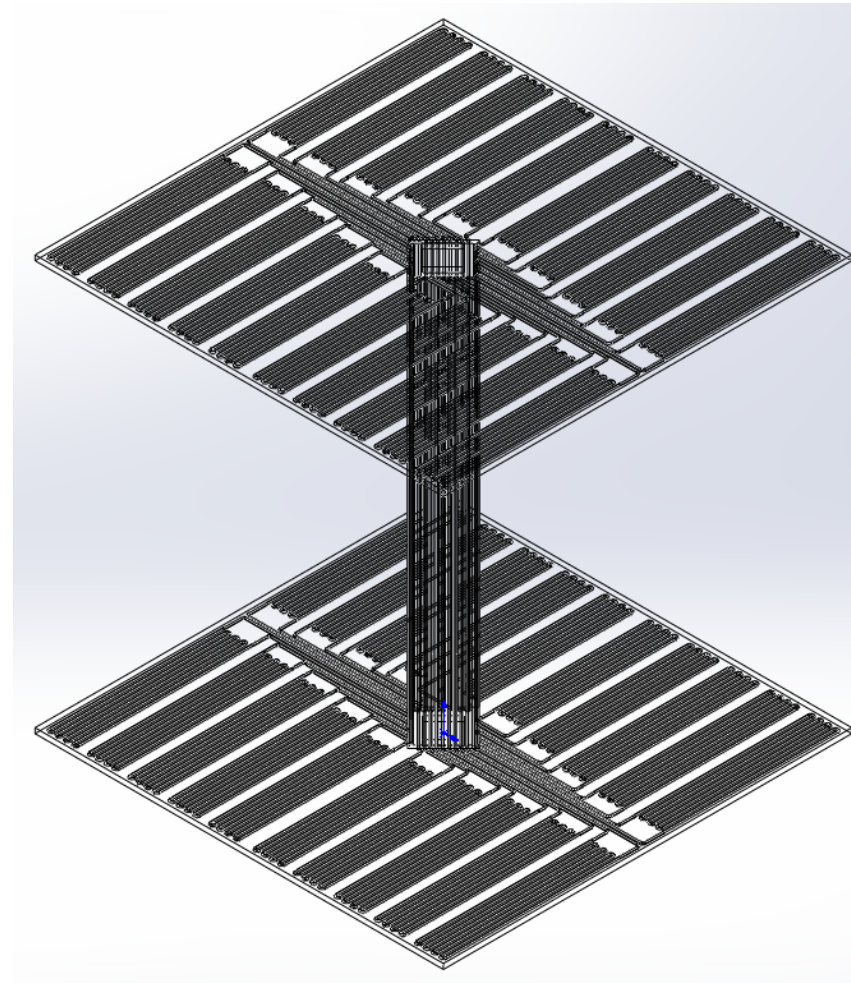
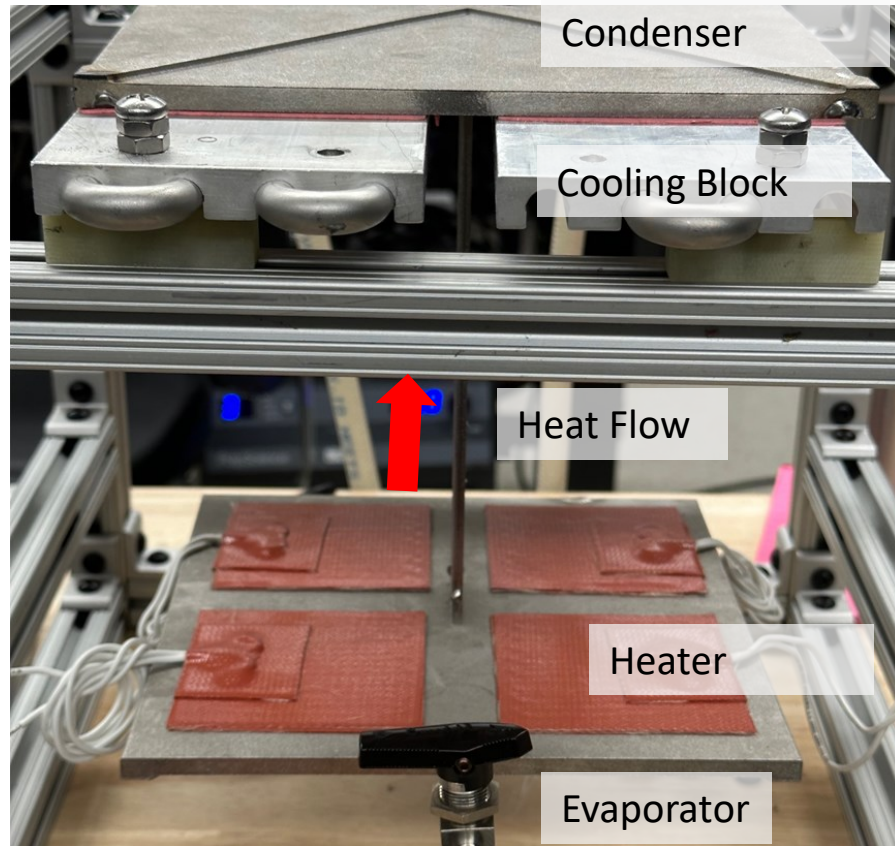


### OHP Heat Switch Design



### Vacuum Insulation Coupon

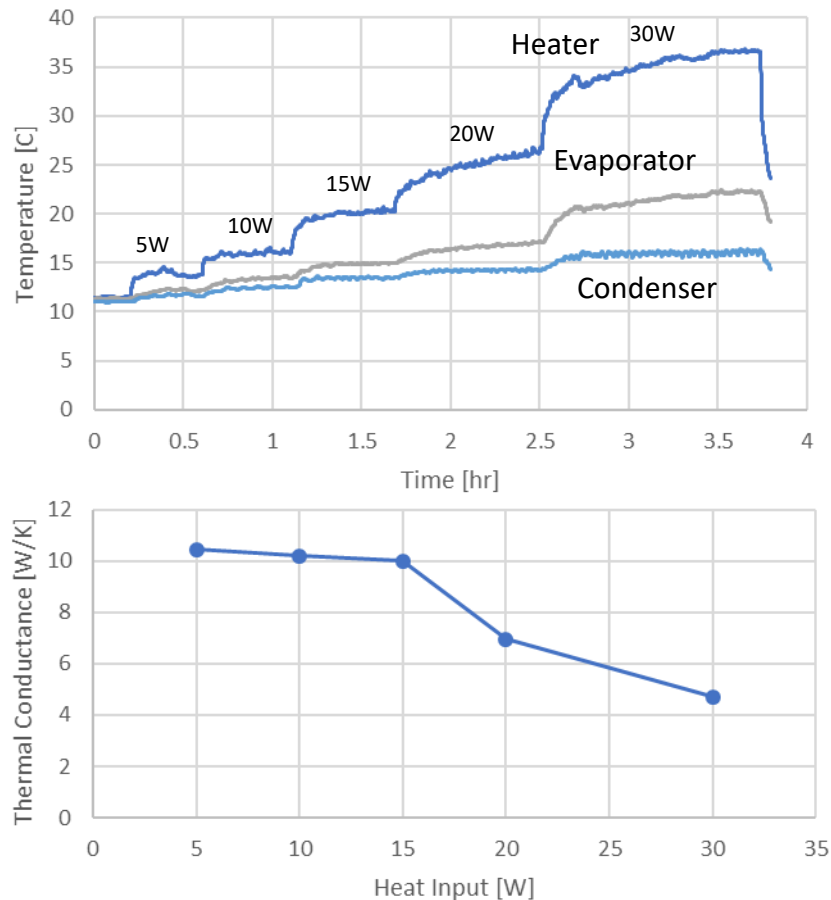
# OHP Heat Switch



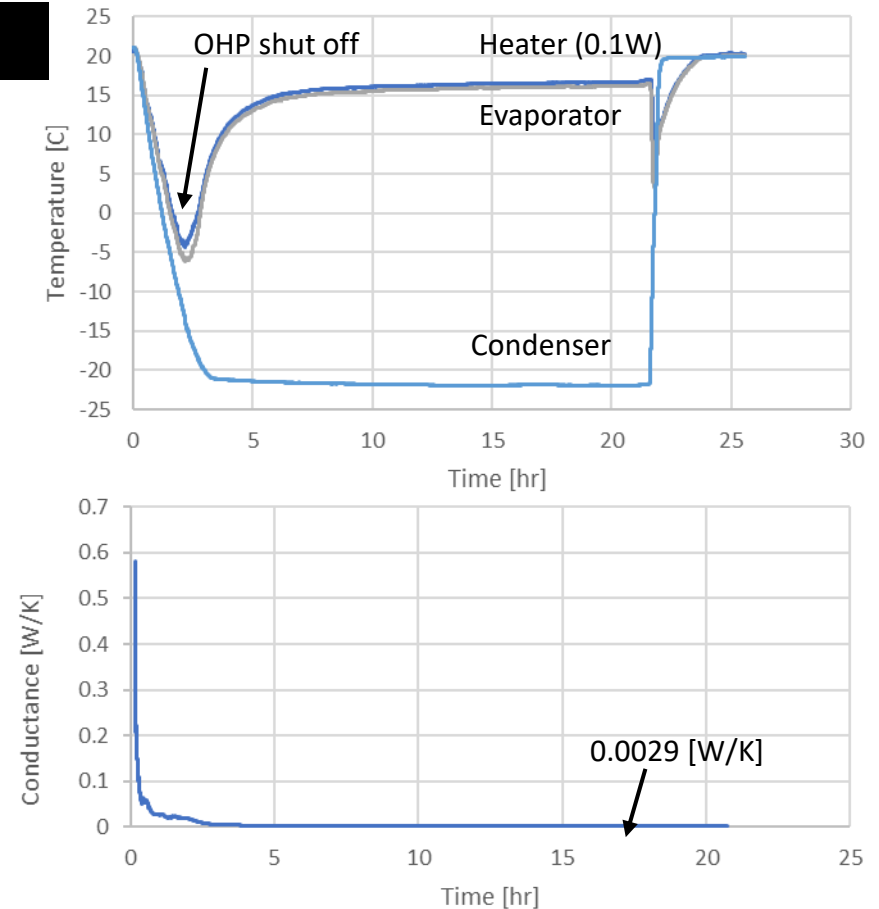
**Internal Channel Design**

# OHP Heat Switch

ON



OFF



## Successful Demonstration

- Stable operation over 3.5 hrs and 20hrs
- **ON Conductance: 10.4 W/K** (Goal 10W/K), Ti-OHP Thermal Conductivity is ~30472 W/m/K (vs 16 W/m/K, Titanium)
- **OFF Conductance: 0.0029 W/K** (Goal 0.001 W/K)



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**Thank you!**